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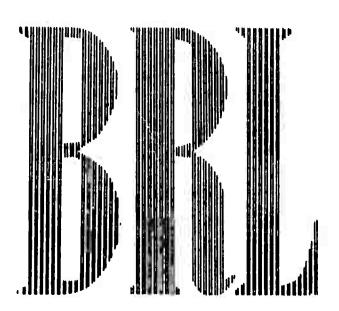
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MEMORANDUM REPORT NO. 1330 MARCH 1961

DOPLOC OBSERVATIONS OF REFLECTION CROSS SECTIONS OF SATELLITES

ARPA Satellite Fence Series

Harold T. Lootens

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NOX

Report No. 22 in the Series



Department of the Army Project No. 503-06-011
Ordnance Management Structure Code No. 5210.11.143
BALLISTIC RESEARCH LABORATORIES



ABERDEEN PROVING GROUND, MARYLAND

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MEMORANDUM REPORT NO. 1330

HTLootens/bt
Aberdeen Proving Ground, Maryland
March 1961

DOPLOC OBSERVATIONS OF REFLECTION CROSS SECTIONS OF SATELLITES

ABSTRACT

This report presents reflection cross sections observed for eight satellites during the period 1 January 1959 to 1 July 1960, using the DOPLOC "dark satellite" detection system developed by the Ballistic Research Laboratories. Several related areas are discussed; i.e., satellite "signature", spin and tumble, scintillation and ionized trails. A brief description of the DOPLOC receiving system and antenna configuration is included. The method used for calculation of cross sections is given in Appendix I.

TABLE OF CONTENTS

		Page
I.	INTRODUCTION	11
II.	DOPLOC SYSTEM DESCRIPTION	12
	A. Equipment	12
	B. Antenna Dimensions and Orientation	13
III.	FIELD DATA	14
	A. Satellite Records	14
	B. Unidentified Flying Objects	14
	C. Doppler Recording	22
	D. Signal Strongth	23
	E. Multiple Antenna Records	24
	F. Meteor's	
	G. Satellite Trails	24
	H. Predictions	25
IV.		25
	REFLECTION CROSS SECTIONS AND POWER RATIOS	27
٧.	CROSS SECTION SIGNATURE OBSERVATIONS	35
VI.	CROSS SECTION MODULATION DUE TO ATTITUDE CHANGE	39
VII.	SCINTILLATION	42
VIII.	REFERENCES	45
	Appendix I - Calculation of Power Ratio and Cross Section	47
	Appendix II - Bibliography of Reports in the BRL-DOPLOC	53
	Series	"

LIST OF FIGURES

- 1. Basic Interim DOPLOC System
- 2. Block Diagram of R. F. Section in DOPLOC Station Passive Tracking at 108 Mc/s
- 5. Block Diagram of DOPLOC Timing System
- 4. Block Diagram of Doppler Data Recording System for a Single Channel
- 5. Block Diagram of Automatic Lock-On System
- 6. ARPA-BRL DOPLOS Satellite Fence
- 7. Typical DOPLOS Data Output
- 8. Typical Single Pass Doppler Orbit Solution
- 9. High Gain Antennas at ARPA-BRL DOPLOC Transmitter Site Located at Fort Sill, Oklahoma
- 10. Initial Antenna Orientation
- 11. Antenna Orientation Following Deactivation of White Sands Station
- 12. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 6844
- 13. ARPA-BRL DOPIOS Doppler Record of 58 Delta, Rev. 7049
- 14. ARPA-ERL DOPLOS Doppler Record of 58 Delta, Rev. 7358
- 15. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 8386
- 16. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 8643
- 17. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 8683
- 18. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 8719
- 19. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 8734
- 20. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 8795
- 21. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9009
- 22. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9172
- 23. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9255
- 24. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9286
- 25. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9466
- 26. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9472
- 27. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9503
- 28. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9581

```
ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9612
29.
```

- ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9716 .0.
- ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9722 31.
- ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9826 32.
- ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9832 33.
- 34. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9842
- ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9848
- 36. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9905
- ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9927 37.
- 38. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9937
- 39. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9943
- 40. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 9959
- 1:1. ARPA-BRI. DOPLOC Doppler Record of 58 Delta, Rev. 9975
- 42. ARPA-BRL DOPICC Doppler Record of 58 Delta, Rev. 9991
- 43. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 10001
- 14. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 10007
- 45. ARPA-BRL DOPLOC Doppler Record of 58 Delta, Rev. 10023
- 46. ARPA-BRL DOPLOC Doppler Record of 59 Epsilon, Rev. 53
- 47. ARPA-BRL DOPLOC Doppler Record of 59 Epsilon, Rev. 60
- 48. ARPA-ERL DOPLOC Doppler Record of 59 Epsilon, Rev. 121
- 49. ARPA-BRL DOPLOC Doppler Record of 59 Epsilon, Rev. 314
- 50. ARPA-BRL DOPLOC Doppler Record of 59 Epsilon, Rev. 414
- 51. ARPA-BRL DOPLOC Doppler Record of 59 Epsilon, Rev. 438
- 52. ARPA-BRL DOPLOC Doppler Record of 59 Epsilon, Rev. 532
- 53. ARPA-BRL DOPLOC Doppler Record of 59 Zeta, Rev. 14
- 54. ARPA-BRL DOPLOC Doppler Record of 59 Zeta, Rev. 150
- 55. ARPA-BRL DOPLOC Doppler Record of 59 Zeta, Rev. 241
- 56. ARPA-BRL DOPLOC Doppler Record of 59 Zeta, Rev. 556
- 57. ARPA-BRL DOPLOC Doppler Record of 59 Zeta, Rev. 855
- 58. ARPA-BRL DOPLOC Doppler Record of 59 Zeta, Rev. 871
- 59. ARPA-BRL DOPLOC Doppler Record of 59 Zeta, Rev. 887
- 60. ARPA-BRL DOPLOC Doppler Record of 59 Kappa, Rev. 183
- 61. ARPA-BRL DOPLOC Doppler Record of 59 Lambda, Rev. 96

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62. ARPA-BRL DOPLOC Doppler Record of 59 Lambda, Rev. 138
```

- 63. ARPA-BRL DOPLOC Doppler Record of 59 Lambda, Rev. 278
- 64. ARPA-BRL DOPLOC Doppler Record of 59 Lambda, Rev. 685
- 65. ARPA-BRL DOPLOC Doppler Record of 59 Lambda, Rev. 1285
- 66. ARPA-BRL DOPLOC Doppler Record of 59 Lambda, Rev. 1516
- 67. ARPA-BRL DOPLOG Doppler Record of 69 Gramma 1, Rev. 318
- 68. ARPA-BEL DOPLOC Doppler Record of 60 Gamma 1, Rev. 403
- 69. ARPA-BRL DOPLOS Doppler Record of 60 Greens 1, Rev. 418
- 70. ARPA-ERL DOPLOC Doppler Record of 60 Gumma 1, Rev. 836
- 71. ARPW-ERL DOPLOS Doppler Record of 60 Gamma 1, Rev. 960
- 72. ARPA-BRI. DOPLOS Doppler Record of 60 Gamma 2, Rev. 1042
- 73. ARPA-ERL DOPLOS Doppler Record of 60 Delta, Rev. 30
- 74. ARPA-BRL DOPLOC Doppler Record of 60 Delta, Rev. 61
- 75. ARPA-BRL DOPLOC Doppler Record of 60 Delta, Rev. 117
- 76. ARPA-BRL DOPLOS Doppler Record or 65 Delta, Rev. 124
- 77. ARPA-ERL DOPLOS Doppler Record of Co Delta, Rev. 140
- 78. ARPA-BRL DOPLOS Doppler Record of 60 Delta, Rev. 156
- 79. ARPA-BRL DOPLOC Doppler Record of 60 Delta, Rev. 165
- 80. APPA-BRL DOPLOS Doppler Record of 60 Delta, Rev. 172
- 81. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 1, Rev. 53
- 82. ARPA-ERL DOPLOC Doppler Record of 60 Epsilon 1, Rev. 99
- 83. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 1, Rev. 150
- 84. ARPA-ERL DOPIOC Doppler Record of 60 Epsilon 1, Rev. 165
- 85. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 1, Rev. 386
- 86. APPA-BRL DOPLOC Doppler Record of 60 Epsilon 1, Rev. 522
- 87. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 106
- 88. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 137
- 89. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 147
- 90. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 153
- 91. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 194
- 92. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 303
- 93. ARPA-BRI DOPLOC Doppler Record of 60 Epsilon 2, Rev. 309
- 94. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 356

- 95. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 2, Rev. 617
- 96. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 3, Rev. 280
- 97. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 4, Rev. 280
- 98. ARPA-BRL DOPLOS Doppler Record of 60 Epsilon 5, Rev. 569
- 99. ARPA-BRL DOPLOC Doppler Record of 60 Epsilon 6, Rev. 265
- 100. ARPA-BRI. DOPLOC Doppler Record of 60 Epsilon 6, Rev. 301
- 101. ARPA-BRL DOPLOC Doppler Record of Unidentified Object
- 102. ARPA-BRL DOPLOS Doppler Record of Unidentified Object
- 103. ARPA-BRL DOPLOC Doppler Record of Unidentified Object
- 104. ARPA-BRL DOPLOS Doppler Record of Unidentified Object
- 105. ARPA-BRL DOPLOC Doppler Record of Unidentified Object
- 106. ARPA-BRL DOPLOC Doppler Record of Unidentified Object
- 107. ARPA-BRL DOPLOS Doppler Record of Unidentified Object
- 108. ARPA-BRL DOPLOG Doppler Record of Unidentified Object
- 109. ARPA-BRL DOPLOG Doppler Record of Unidentified Object
- 110. ARPA-BEL DOPLOC Doppler Record of Unidentified Object
- 111. ARPA-BRL DOPLOC Doppler Record of Unidentified Object
- 112. ARPA-BRL DOPLOC Doppler Record of Unidentified Object
- 113. ARPA-BRL DOPLOC Doppler Record of Unidentified Object
- 114. ARPA-ERL DOPLOC Doppler Record of Unidentified Object
- 115. DOPLOC Power Ratios, Center Antenna
- 116. DOPLOC Power Ratios, North Antenna
- 117. DOPLOC Power Ratios, South Antenna
- 118. DOPLOC Power Ratios, Center, North and South Antennas
- 119. Doppler Record of Active Track of 58 Delta, Rev. 9958
- 120. Doppler Record of Active Track of 60 Epsilon 1, Rev. 26
- 121. Doppler Record of Active Track of 59 Epsilon, Rev. 7
- 122. Doppler Record of Active Track of 58 Delta, Rev. 10007
- 123. DOPLOC Frequency and Rate of Change of Frequency as a Function of Position in the YZ-Plane
- 124. Center Antenna Geometry
- 125. Altitude and Ground Range Geometry in South Antenna
- 126. Satellite Inclination with Respect to Ease Line

- 127. Ground Range in North and South Antennas
- 128. North and South Antenna Geometry
- 129. Power Factor vs Angular Position for an 8 x 76 Degree Antenna

I. INTRODUCTION

During the period 1 January 1959 to 1 July 1960, the Ballistic Research Laboratories, under funding from the Advanced Research Projects Agency (ARPA Order 8-58), operated a three-station, reflection Doppler satellite tracking system, extending across the south-central United States from Tennessee to New Mexico. This system, known as DOPLOC (Doppler Phase LOCk), provided a means of detecting and tracking radiosilent, or "dark" satellites.

A transmitting station was located at Fort Sill, Oklahoma and receiving stations were located at White Sands Missile Range, New Mexico and at Forrest City, Arkansas. The stations were initially manned on a twenty-four hour, seven-day-per-week basis, as part of the nation-wide satellite surveillance net. Following permission from ARPA to discontinue routine twenty-four hour operation, the White Sands station was deactivated and a basic eight-hour work day was adopted at the Fort Sill and Forrest City stations on or about 1 October 1959. The actual hours of operation were chosen to adapt the work schedule to the times of most frequent satellite passes.

The flexible schedule by which the field stations operated has provided considerable data from known satellites, Unidentified Flying Objects (UFO's) and meteors. Many satellites and UFO's have been successfully detected and tracked by the DOPLOC technique and their time of crossing, altitude, east-west position and effective reflection cross section determined from single pass data from a single receiving station. Crossing time and position data was forwarded to Space Track Control Center for inclusion in their orbital prediction program.

II. DOPLOC SYSTEM DESCRIPTION

A. Equipment

The DOPLOC system consisted of a 50-kw continuous wave, 108 Mc/s transmitter located at Fort Sill, Oklahoma, which fed one of three narrow-beam, high-gain antennas. These high-gain antennas emitted narrow, fan-shaped beams, one directed 20 degrees above the northern horizon, one directed vertically and one directed 20 degrees above the southern horizon (see Figure 1).

The signal reflected from a satellite passing through the transmitter beam was received at one or both of the receiving stations. Each receiving station had three high-gain antennas oriented to "see" the space volume illuminated by the transmitter. The reflected Doppler signal was fed through a receiver to a bank of fixed audio frequency filters known as the Automatic Lock-On (ALO) and, subsequently, to a narrow band, phase-locked tracking filter. The appearance of a Doppler signal in one of the fixed filters activated a control circuit which pulled the tracking filter frequency over to the signal frequency and caused a phase lock between the two. The tracking filter then tracked the Doppler signal frequency continuously as the satellite passed through the antenna beam. Block diagrams of the DOPLOC receiving system and ALO are shown in Figures 2, 3, 4 and 5.

A satellite which crossed the base line joining the transmitter and receiver traversed each of the three overlapping, fan-shaped antenna beams (see Figure 6). This resulted in three separate Doppler records, one for each of the three antennas, separated in time by 30 - 60 seconds. The length of the records varied, averaging about 7 seconds in the center antenna and 15 - 25 seconds in the north and south antennas. The digital Doppler data as a function of time were printed on paper tape and also converted to binary form and punched into standard five-hole teletype tape for transmission via commercial teletype to the BRL computer center, where they were fed to the ORDVAC computer to obtain satellite orbital

parameters. Figure 7 shows typical DOPLOC data output and Figure 8 shows a typical orbital solution calculated using this type of data. In addition to recording the digital data, recordings on paper charts were made of the Doppler analog frequency and signal strength with respect to time. Reproductions of these records are presented in this report.

B. Antenna Dimensions and Orientation

The special high-gain antennas were 60 feet long and 10 feet wide with beam dimensions of 8 x 76 degrees and a gain of 16 db over isotropic. Three of these antennas were installed at each of the receiving stations and at the transmitting station. The antenna installation at the transmitting station is shown in Figure 9.

When the DOPLOC satellite detection system assumed twenty-four hour operational status in January 1959, the transmitter at Fort Sill served as the illuminator for both receiving stations. At that time, the high-gain antennas were oriented in azimuth and elevation as shown in Figure 10. In the fall of 1959, the receiving station at White Sands was deactivated and it was decided to tilt the antennas at Fort Sill and Forrest City to produce more favorable coverage. The reorientation of the antennas to the configuration shown in Figure 11 was made in November 1959.

A complete and detailed description of the DOPLOC satellite detection and tracking system may be found in BRL Report No. 1123, "The DOPLOC Instrumentation System for Satellite Tracking" (February 1961).

III. FIELD DATA

A. Satellite Records

During the 18-month operation of the DOPLOC system, lll refrections were received, resulting from observations of 89 individual satellite passes. More than half of these reflections were received by the center antenna alone, while the rest were observed by the north or south antennas or various combinations of the three antennas.

Reproductions of DOPLOC Doppler reflection data are presented as follows:

Sputnik III (58 Delta), Figures 12 to 45; Discoverer V (59 Epsilon), Figures 46 to 52; Discoverer VI (59 Zeta), Figures 53 to 59; Discoverer VII (59 Kappa), Figure 60; Discoverer VIII (59 Lambda), Figures 61 to 66; Transit 1B rocket (60 Gamma 1), Figures 67 to 71; Transit 1B (60 Gamma 2), Figure 72; Discoverer XI (60 Delta), Figures 73 to 80; Sputnik IV (60 Epsilon 1), Figures 81 to 86; Sputnik IV rocket (60 Epsilon 2), Figures 87 to 95; Sputnik IV fragments (60 Epsilon 3, 4, 5 and 6), Figures 96 to 100. A summary of these reflections arranged by satellite and antenna is given in Table 1, and a detailed explanation of each pass may be found in Table 2.

B. Unidentified Flying Objects

A number of reflections were received and recorded which could not be correlated with the predicted position of any known satellite. These were termed Unidentified Flying Objects and reproductions of 14 of these reflections are shown in Figures 101 to 114, with a detailed listing in Table 3.

TABLE I

DOPLOC SATELLITE REFLECTION SURMARY
1 January 1959 - 1 July 1960

					Antenna				
Satellite					North &	South &.	North &	Three	
Nounc	Designation	Center	North	South	Center	Center	South	Antenna	Total
Sputnik III	58 Delta	19	හ	Q	rt	3	,	-	F
Discoverer V	59 Epsilon	72	•	a	•		,	i 1	, ,
Discoverer VI	59 Zeta	9	,	ч	ı	,	•		- 1
Discoverer VII	59 Карра	ч	•	ı	,	,	•		- ,
Discoverer VIII	59 Lambda	17	ı	ı	Н	r-1	•		ч Ч
Transit 1B (rocket)	60 German 1	₹	ı	ı	,	1 1	,) (ט ע
Transit 1B	60 Gamma 2	н	•	•	1	•		1	Λ,
Discoverer VI	60 Delta	ı	ı	н	•		, ۱	י ע	⊣ 0
Sputnik IV	60 Epsilon 1	8	ч	Q	•	ı	1 (o	0 (
Sputnik IV (rocket)	60 Epsilon 2	7	Q	Q	-	ı ı	1	•	، م
Sputnik IV (fragment)	60 Epsilon 3	-	ı	1 1	ı	ı	ı	ı	O,
Sputnik TV (fragment)	1	، ا)	•	ı	•	•	ı	ч
	on sparron 4	٦.	•		ı	1	•	ı	Н
Sputnik IV (fragment)	60 Epsilon 5	н	•	•	•	•	,	(
Sputnik IV (fragment	60 Epsilon 6	Q	,	ı	1			1	- 1
	•	ļ		ı	1	ı	ı	•	a
			1	1		ļ	İ	l	
٠	Totals	53	ជ	70	К	4	н	7	89
Unidentified Objects		16	ı	1	1	ı	ŧ	•	16

TABLE 2 DOPLOC GATHLITE REFLECTIONS, 1 JANUARY 1959 - 1 JULY 1950

Satellite Name	Rev	Date 1959 - 60	Time G.M.T.	Pacs	Altitude	GP(1) Miles	Stenal	2/2da	Penk Signal	57. The
			!				ಚಾದರಾವ್ಯ			
58 Delta	6844	11 Sep 59	0140:12	د: ::	1.01;	283 3	G.	(to a year of the second seco	- 172.0	
58 Delta	70149	Sep	2201:56	S-1	575	23 to 25	30	#** \$ 1	- 172.0	
58 Delta	7358	8	1646:25	S-:	696	:=	(4	3	- 176.5	
58 Delta	8386	Dec	2156:23	ιο 12	1772	176 E	(~	<u>(i)</u>	- 166.5	
58 Delta	8643		1647:25	S-N	158	1,50 €	\$ 700	8	- 170.0	
58 Delta	8683	11 Jan 60	0651:30		413	Orenhead	•7	4)	- 175.0	
58 Delta	8719		1530:05	ιη ::	186	₹ 04	•	₫.	171.0	
58 Delta	8734	14 Jan 60	1454:02	0-s	186	326 H	g,	āŠ	- 170.0	
58 Delta	8795	18 Jan 60	1558:01	ະກ •••	146	158 %	Q.	٤	- 172.e	
58 Delta	6006	1 Feb 60	1011:12	S-2	1,54	# 9#	• 1	0	- 172.0	
58 Delta	9172	11 Feb 60	2121:06	S	315	£30 £	c.	Ož U	- 178.0	
58 Delta	9255	17 Feb 60	0523:57	S-12	128	350 8	1	51.	- 161.0	
58 Delta	9286	19 Feb 60	0458:45	N-S	125	:6	g\	106	- 366.5	
58 Delta	9946	1 Mar 60	1552:58	S-50	275	28% E	इ	ထ္ဆ	- 167.5	
58 Delta	2246	2 Mar 60	0127:31	8-15	130	30 %	Po 1	400	- 172.0	
58 Delta	9503	4 Mar 60	004.1:30	S	150	3 9	1	06	- 170.0	
58 Delta	9581	8 Mar 60	2317:41	S-::	र्भेट	3 22	Q)	101	- 168.0	
58 Delta	9612	10 Mar 60	2218:33	S-::	116	3 661	¢,	5	- 373.0	North Antenna
58 Delta	9716	17 Mar 60	1114:54	S-:	275	135 E	-7	0.	- 175.0	
58 Delta	9722	17 Mar 60	2044:31	S-::	113	:::	24	3	- 178.0	South Antenna
58 Delta	9886	24 Mar 60	0847:25	S-3	189	105 E	មា	6)	- 175.5	
58 Delta (2)	9832	24 Mar 60	1812:26	S-::	100	280 €	ø	ता	- 170.5	
58 Delta (2)	9832	24 Mar 60	1813:19	S-1:	100	E 20%	-7 ():	9	- 375.0	South Antenna
	9842	25 Mar. 60	0848:40	S-K	182	591 8	ሆ \ የተ	16	- 177.0	North Antenna
58 Delta (2)	8486	25 Mar 60	1811:46	S-::	100	155 8	6 7)	105	- 171.5	
58 Delta (2)	9848	25 Mar 60	1812:38	N-S	100	3 998	16	9,	- 173.0	South Antenna

TWBE 2 (continued)

Satellite Name	Rev Number	Date 1959 - 60	Time G.M.T.	Pass	Altftude Miles	(2) (2) (3) (3) (4)	Stepal Daration Seconds	office of	Fear Signi	
58 Delta	6 00€	29 Mar 60	0559:58	:: 	171	0,,	2.7	7.	178.0	South Asterney
58 Delta	7566	30 Mar 60	1613:25	ن. : :	105	3 82.	.*, C,		- 156.0	
58 Delta (3)	7566	31 Mar 60	0641:14	S-::	180	3 (01	<u>ئ</u>	C.*	- 172.0	South Antenna
	7566	31 Mar 60	0642:57	:: 50	180	in clas	ţ-	g) (g)	152.0	
58 Delta (5)	7566	31 Mar 60	0644:19	;; - ₩	180	141 141 141 141	C)	et et	0.927	March Artes
58 Delta (4)	5466	31 Mar 60	1601:55	S-1	10%	10 m	\#\ 18	67	0.550	
58 Delta (4)	€466	31 Mar 60	1603:20	53	10.		* ~	P V	166.0	
58 Delta	6666	1 Apr	1547:58	63	203	[1] C	\$ man \$6.5	60	- 256.0	THE STREET
58 Delta	5166	2 Apr 60	1530:24	<i>S</i>	101	30°	10	es r.	E	
58 Delta	1666	3 Apr	1509:18	S-1:	ů,	(4) (-)	6 1		0.0	
58 Delta (2)	10001	4 Apr 60	0529:14	:: ::	:2:	3 90 5	Ş	10 6.	- 177. E	
58 Delta (2)	10001	4 Apr 60	0530:49	:-S	124	(i)	\$	5	6. 5	
58 Delta	10007	L Apr S	14:44:57	:: ::	ě.	12	4.4	9.	4.000	100 mm
58 Delta	10023	5 Apr 60	1415.08	13 14	5	4	·	C+,	0.627	THE PARTY OF THE PARTY OF
59 Epsilon	5	17 Aug 59	0539:54	::- ::- ::-	24 25 25	6. 5. 10.	•	W %	- 177.0	
59 Epsilon	3	17 Aug 59	1701:10	S-11	240	4 91	ar)	S	- 177.0	
59 Epsilon	121	21 Aug 59	1620:02	8-8	0.1	3 007	000	0	0.471.	Series Attended
59 Epsilon	416	3 Sep 59	0306:50	::-s	135	3 667	ţ-	(7), (1)	0.00	
59 Epsilon	१८५	6 Sep 59	1322:15	\$9 + 15	11:	: -	4.0	(c) (i)	0.000	
59 Epsilon	458	11 Sep 59	0141:16	::	2.50	3 6 C	9.	500	\$\frac{1}{2}\$	で から で で で で で で で で で で で で で で で で で で
59 Epsilon	532	17 Sep 59	0105:49	S-::	22	:	0,	(t) eri	101.5	While - waster things
59 Zeta	17	20 Aug 59	1741:51	S -:	3.56	ម ភ កា	าอ	83	167.0	
59 Zeta	150	29 Aug 59	1650:08	٠ <u>٠</u>	171	:A	6.	gri et	15° 00° 00° 00° 00° 00° 00° 00° 00° 00° 0	では はない ない な
59 Zeta	241	4 Sep 59	1559:34	2-N	213	3 60	6-	23	- 263.5	
59 Zeta	925	25 Sep 59	91:4410	:- S	139	3 692	27	326	- 178.0	
59 Zeta	855	13 Oct 59	2347:38	22-05 05	138	3 611	t-	C.	- 169.0	
59 Zeta	67.1	14 Oct 59	2342:09	٠. بر ۲	175	3 80T	G,	5	- 165.0	
59 Zeta	887	15 Oct 59	2532:58	x-2	170	123 E	1.	C)	- 164.5	

Tank ? (continue!)

Satellite Name	Rev Number	Date 1959 - 60	Time G.M.T.	Para	Altitude Miles	GE(1) Miles	Stenal Fraction Seconds	21012	ran Signal	and the second of
59 Kappa	183	19 Nov 59	1704:59	27 ×25	11.5	a 355			**	
59 Lambda	96	27 30" 50	16.7.00		1		ń.	-7 -1	- 161.0	
59 Tambda	מאל ב		(N.) #0.7	ж- к	757	iii Q	-1	*** ***	0.496	
	170	SO NON DO	1644:51	ن: ت:	158	Overhead	ps."	5		
	278	10 Dec 59	1503:37	57	218	u,	. (: 1	0	
	278	10 Dec 59	1505:00	ن ا	218	៩ ដូ	0.		- 270.5	THE TOTAL STREET
59 Lambda (2)	685	8 Jan 60	0016:01	52	65.6		7 (io i		
59 Lambda (2)	, 685	8 Jan 60	0017:05	: :: ::) (C	5 (0	9	- 215.5	South Antenna
59 Lambda	1285	17 Feb 60	0514:55	: 0	0 0		-7	# 1 # 7 # 1	- 172.5	
59 Lambda	1516	3 Mar 60	0056.10	3 0	900		-1	9	1 CO	
1			01:50	νη •	5	# () ?	:1	Co.	156.0	
	318	4 May 60	12:57:23	N-S	1,40	30	4	14		
SO Games	403	10 May 60	0517:28		27.7			ρ	- 174.0	
8 64	418	11 %ay 60	0505:25	i i	ė č		<i>:</i> .	CIT CIT	C- 17	
60 Greens 1	836	7 Jun 60	1632-12	: 3	- C:		• •	• 1	- 271.0	
60 CHIMM 1	960	15 Jun 60	2018-10	: (346		(I) e-t	7	- 167.0	
, to them.	3		•	0	0:	:::	• •	99	- 165.0	
A CHIEF E	1042	21 Jun 60	1902:09	ن. :	250	255 €	* (
60 Delta (3)	30	17 Apr. 50	1848.12	:	4			· ·	0.474	
60 Delta (3)	30		1830.10	3 (d	77	u: ar	64	9.	- 174.0	
60 Delta (3)	0,0	17 Apr 60	1810.05	2 (1 5)	H	141 C2	·බ		- 368.0	
60 Delta (5)	. 59	10 Am 60	200:00	19 18	777	110	:-	35	- 176.0	11
	1 6		1001:55	در. 	111	2.0 H	4 618	3	174-0	Touth total
	711		2005:50	ر. 	111	\$20 m	o,	C.	169.0	
60 Telta (3)	101	30 144 62	1::1	×-%	20%	125 %	<u> </u>	e-1	1750	
	+ 77	<2 Apr 60	1727:08	رن - برد در - برد	120	350	6.7			
	124	23 Apr 60	1728:18	ري د -	120			87 €	0.72	Horth Antenna
60 Delta (5)	124	23 Apr 60	1729:25				s	118	- 167.0	
60 Delta (5)	140		1727.00	2 :	750	360 ∺	•7	23	- 178.0	South Antenna
60 Delta (3)	140		1728.07	; ;	å :		:	S.	- 161.0	North Anterna
60 Delta (3)			1728.57	2 :	ne :	32 OC	ý	112	- 171.0	Ī
			10.01	13 13 13 13 13 13 13 13 13 13 13 13 13 1	911	₹9 E	7.	50		South Antenna

Tille 2 (continued)

estator.	North Antenna		South Antenna	South Antenna		SOUTH ANTON	North Astenna		South Artema	South Asterna	Sortis Anti-	South Antenna				South Anthony			North Antenna	South Antenna	North Antenna			North Antenna		
Peak Signal dbu	- 363.0	- 175.0	- 175.0	- 172.0	- 365.0	- 165.0	- 170.0	160.0	- 168.0	- 168.0	- 172.0	- 177.0	161	0.691 -	- 165.0	- 176.0	200	To the last	- 172.5	178.5	- 174.5	- 176.5	- 165.0	- 172.0	- 177.0	- 166.0
\$10pe c ps/s	59	51	Q.	:3 :-1	5	2	4	168	Car	(₁)	- E1	05	ထု	76	11.	a]] er((D)	හි	16	36	25	ಕ	-	(S)	හි	74
Stgm1 Duration Seconds	- t	C.	٨	₽ \$	ത	ma N \	Ŭ\ erf		C.	000	J:	20		Ç	c)	974 974	\$T	(1)	₽; •7	er4 er4	17	7	;-	15	7	13
GR ⁽¹⁾ Mics	307	143 %	181 %	e C	8 8	185	3 55 c	3 044	3.85 m	5.7 C:	50°	<u>න</u> වර්දි	14 150 14	:A3	0.	1500	100 C.	66	1,52 3	3 211	39.9	1,20 5	305 8	30 8	3 11	3.858
Altítude Míles	112	211	122	142	11.2	1122	-	1-	i-	194	200	287	238	230	535	208	209	240	240	509	195	215	205	201	203	232
Fass Direction	េះ	S	S-11	# - 0	2-0	:: '0	S-11	S-22	\$		2°−0	S-2	5	::5	::- :>	46 25	2-2	S-3:		N-5	×-0	8-S	S-18	S	N-S	≅ .
G.M.T.	1717:39	1718:40	1719:31	0509:55	0611:13	0612:06	1653:08	1653:56	1654:29	0833:05	0801:32	1653:44	1624:59	0300:41	0028:41	1733:24	1625:46	0720:57	0722:37	1655:14	06:12:00	0354:58	1325:57	1240:39	2218:14	62:1440
Date 1959 - 60	25 Apr 60	25 Apr 60	25 Apr 60	26 Apr 60	26 Apr 60	26 Apr 60	26 Apr 60	26 Apr 60,	26 Apr 60	18 May 60	21 Nay 60	24 May 60	25 May 60	9 Jun 60	18 Jun 60	21 May 60	25 Nay 60	24 May 60	24 May 60	24 May 60	27 May 60	3 Jun 60	3 Jun 60	6 Jun 60	22 Jun 60	2 Jun 60
Rev Number	156	156	156	165	165	165	172	172	172	53	66	150	165	386	522	106	137	147	147	153	194	303	309	356	219	280
Satellite Name	60 Delta (3)	60 Delta (3)	60 Delta (5)	60 Delta (5)	60 Delta (5)	60 Delta (3)	60 Delta (3)(6)	60 Delta (5)(6)	60 Delta (3)(6)	60 Epsilon 1	60 Epsilon 1	60 Epsilon 1	60 Epsilon 1	60 Epsilon 1	60 Epsilon l	60 Epsilon 2	60 Epsilon 2	60 Epsilon 2 (4)	60 Epsilon 2 (4)	60 Epsilon 2	60 Epsilon 2	60 Epsilon 2	60 Epsilon 2	60 Epsilon 2	60 Epsilon 2	60 Epsilon 3

TiME ? (continued)

Satellite Name	0									
	Number	uate 1959 - 60	G.W.T.	Pass	Altitude Mles	GB(1) Wiles	Stern?	81026 2018	THE STATE	and of the state o
	0						200000			
# BOILSGE OO	280	2 Jun 60	04.45;21	17. 17.	0.50	111	ore ext		克 · 克 · · · · · · · · · · · · · · · · ·	
60 Epsilon 5	36	20 101						h n	7 - 10 1 1	
		00 000	5.555 HP	:= '5	20%	231 8	(4)	(i)	346	
60 Epsilon 6	265	1 Jun oo	50.00		4)	0.00	
60 Faction 6			2	-0	218	210 #	-17 PM	67	0 0	
	70	oo ame oo	1359:22	හ <u>-</u> ::	4°.	190	•	f		

General Notes:

Transmitting antenna tilted towned receiving untenna 26 October 1959.

2. ALO scan changed to 4 kc 27 November 1959. 5. All passes acquired by center antenna unless otherwise noted.

Under remarks, "RERUN" indicates original signal not acquired by A10. Pass was recovered from playback of magnetic tape.

5. Slope denotes rate of change of Doppler frequency.

References:

(1) Ground range East, West or Overhead Fort Sill trunsmitter.

(2) Acquired by South and Center antennus.

(3) Acquired by North, Center and South antennus.

 (μ) Acquired by North and Center antenns. (5) Acquired by North and South antennas.

(6) Terminal revolution.

TABLE >

DOPLOC UNIDENTIFIED OBJECT REFLECTIONS

1 January 1959 - 1 July 1960

	(3)				(3)			(3)						
REMARKS) s mottsgg 65				59 Epsilon 2 (59 Epsilon 2 (
PEAK SIGNAL	-166.0	-171-5	-159.0	-16:0	-184.	-164.5	-161.0	-170.0	-179.0	-177.0	-159.0	-167.5	-159.0	-174.0
SLOPE ⁽²⁾	110	50	50	100	2113	142	83	94	20	56	70	16	85	61
SIGNAL DURATION SECONDS	7.0	S.0	5.0	5.0	O	5.0	3.0	0.9	8.0	3.5	5.0	0.1	3.5	2.5
G. R. (1) MILES	Overhead	318 E	3 Otl	117 E	115 E	20 W	165 E	140 E	Overhead	135 E	国の内	130 E	117 Е	M 06
ALTITUDE MILES	125	14.32	398	190	178	100	198	485	006	190	265	190	19 5	300
TIME G.M.T.	1703:30	1514:01	1643:04	0235:22	1641:31	2258:01	1843:18	2056:00	2119:13	21,46:10	0101:22	0057:04	0120:43	0906: 33
DATE 1959-60	17 Aug 59	14 Sep 59	22 Sep 59	1 Oct 59	15 Oct 59	11 Dec 59	16 Dec 59	14 Jan 60	11 Feb 60	12 Feb 60	18 Feb 60	1 Mar 60	1 Mar 60	29 Mar 60

All reflections received on center antenna.

⁽¹⁾ Ground range East, West or Overhead Fort Sill transmitter. NOTES:

⁽²⁾ Rate of change of Doppler frequency.

⁽³⁾ Tentatively identified as the recovery capsule from Discoverer V (59 Epsilon 1).

C. Doppler Recording

The typical form in which DOPLOC data are recorded for satellite detection is shown in Figure 60, a record of Discoverer VII (59 Kappa). An explanation of this record and the ALO device by which it was obtained follows. The upper portion of the chart is an analog record of tracking filter output frequency. The short, evenly spaced marks indicate the successive frequencies at which the tracking filter is set while the system is in the search mode. The tracking filter is stepped in 1 kc/s intervals to maintain a frequency midway in the 1 kc/s spectrum to which the comb filters are set. This minimizes the time required to pull the tracking filter frequency to a signal frequency detected in one of the fixed filters. The comb filter bank consists of ten filters, each with a 20 c/s bandwidth, spaced 100 c/s apart. The filter bank "looks" at a 1 kc/s frequency band for 0.1 second; then it is switched up 1 kc/s by a heterodyne method and this process continues, until either the desired frequency band has been covered or a signal is detected. Figure 60 shows the tracking filter output when a 12 kc/s scan is used. The ALO can also be adjusted to scan a 4 kc/s range (see Figure 19) or a 2 kc/s range (see Figure 24), or it may be manually positioned to a desired frequency value. Manual operation is useful when attempting to lock the filter on a signal being played back from magnetic tape, where the initial Doppler frequency is known within a few cycles per second. Figures 13 and 22 are examples of manual filter positioning.

The initiation of phase-locked tracking is accomplished quickly when a signal frequency is detected in one of the fixed filters. The control circuit pulls the tracking filter frequency over to the received Doppler signal frequency in about 10 milliseconds and within 80 to 90 milliseconds all transients have subsided and phase-locked tracking begins. Figure 60 shows this transition from step scanning to continuous tracking at 1714:58 Z time. Concurrently, the digital counter and printer is started and the time period of 1000 cycles of the Doppler is printed at one second intervals on paper tape. Simultaneously, the period count is converted to binary form

and punched into five-hole teletype tape. The Doppler period count for Revolution 183 of 59 Kappa, which corresponds to the Doppler frequency analog record, is shown at the top left of Figure 60. The right five digits constitute the count, while the left six digits represent Universal Time in hours, minutes and seconds. The Doppler frequency in cycles per second is 10 times the reciprocal of the count. A Doppler frequency record readable to 0.1 c/s is obtained in this manner. If desired, the Doppler frequency may also be printed directly in digital form (see Figure 7).

D. Signal Strength

The lower part of the chart in Figure 60 is a record of the AGC voltage from the tracking filter. While in the search mode, the AGC is shorted, giving the clean, straight line at 2 mm deflection. When a signal is detected, the AGC voltage first decreases due to an initial threshold voltage of opposite polarity existing on the AGC line, which causes a deflection toward zero on the chart. Then, as the signal amplitude increases, the AGC voltage increases as shown by the scale calibration. The chart is calibrated in power input level (in dbw) to the receiver input terminals and also in relative signal in terms of the signal-to-noise ratio at the receiver output, i.e., in db below 1:1 S/N at the receiver output. With a 10 c/s bandwidth, the tracking filter can track signals that are 26 db down in the noise from the 16 kc/s bandwidth receiver.

The signal strength record of 59 Kappa in Figure 60 shows a maximum signal of -161 dbw which is 2 db down in the noise at the receiver output. The rather narrow, peaked signal response curve with the slight dip on the leading portion is "signature" information indicating considerable attitude change during the six second passage time through the antenna beam. The peak cross section for this pass of 59 Kappa was calculated to be 226 square feet from this record. The rather detailed treatment of this one pass of 59 Kappa has been given to illustrate the detailed nature, quantity, and quality of the data that are provided by the DOPLOC satellite tracking system from a single pass recorded by a single receiving station.

E. Multiple Antenna Records

The previous discussion of experimental results has been largely devoted to data received by the vertically directed center antenna. Figure 63 shows a similar Doppler record of a satellite signal received by the north antenna and later by the center antenna. The Doppler frequency has a low value and is nearly constant during transit through the north antenna beam which is directed 20 degrees above the horizon. During this interval the Doppler corresponds to the flat portion of the "S" curve. The region between the satellite signal in the north and the center antenna is of interest in this record since it represents a period of unusually high spurious signal activity. The short, steep slope lines are typical of meteor head echoes, and are easily distinguished from the satellite record either by their steep slope or their very short duration AGC record (one second or less). Two of the slopes are of opposite sign to those of the satellite record due to the extremely high velocity of the meteor, which places the Doppler frequency on the opposite side of the heterodyne frequency.

Optimum performance of the DOPLOC system is shown in Figure 80, which is a record of a satellite passing through the three antenna beams successively. This record depicts the step-scan frequency search, the lock-on, and the continuous track sequence as the satellite passed through the north antenna beam, the center beam and the south beam. It can be seen that the 4 kc/s scan range is switched up as soon as the satellite signal has ended in each antenna. This operation is performed manually by the operator who is visually monitoring the ALO output. This record is of particular interest since it is the last revolution of 1960 Delta over the Northern Hemisphere. During the latter part of this revolution, this satellite re-entered the earth's atmosphere over the Southern Hemisphere.

F. Meteor's

In addition to the satellite Doppler frequency record in Figure 60, other short lines of about one second duration are evident in the upper portion of the chart. These are spurious responses due to strong noise

pulses or meteor head echoes. It is significant to note that a spurious frequency signal occurred just a few tenths of a second prior to the satellite Doppler signal reception, yet the ALO was able to respond with full sensitivity to the desired signal. Spurious signals from meteors are identified by their short time duration and steep slopes. Signal reflections from meteor trails, which are large ionized columns moving at very low velocities, are recorded as nearly constant frequency, called "flats", which are close to or equal to the bias frequency.

G. Satellite Trails

There is some indication that the passage of a satellite through the ionsphere produces a cloud of ionized particles in its wake, 1,2,3, causing a constant frequency reflection similar to the "flat" reflections produced by meteor trails. The existence of such an ionized cloud is further supported by data as shown in Figures 16 and 18, where a "flat" is seen immediately following the Doppler reflection from 58 Delta. Other constant frequency reflections appearing after a satellite pass may be seen in Figures 26, 27, 30, 66, 89, 92 and 100.

Figure 15 shows an interesting example of a satellite pass occurring simultaneously with a "flat". Revolution 8386 of 58 Delta was detected three seconds after the constant frequency reflection was observed. This example demonstrates the ability of the DOPLOC system to detect and track a satellite in the presence of a large, interfering signal.

H. Predictions

Satellite predictions computed and distributed by Space Track Control Center were used to determine base line crossing times for known satellites. Two chart speeds were used for the analog recordings; 2.5 mm/second during specifically selected search periods when a satellite was predicted to cross the DOPLOC base line and 1 mm/second at all other times during routine surveillance.

Not every satellite known to have passed between the transmitter and receiver was detected, apparently because of insufficient reflected signal

due to satellite attitude at the time it traversed the antenna beam. For the same reason, many passes were detected by one or two antennas but not by all three antennas. These one or two-antenna reflections prove extremely useful, however, when examined in conjunction with the three-antenna data, in analysis and comparison of cross sectional areas, "signature", sat lite attitude changes (spin and tumble) and seintillation.

IV. REFLECTION CROSS SECTIONS AND POWER RATIOS

As previously stated, the Doppler frequency vs time data are digitally recorded at the receiving station as the satellite passes through the antenna beam. From these data, we may calculate the Doppler slope (rate of change of Doppler frequency) and, subsequently, the altitude and east-west position of the satellite as it crossed the base line joining the transmitter and receiver. Using these values and the method described in Appendix I, we calculate the power ratio (ratio of calculated received power to measured received power) and the apparent cross section observed for each satellite pass through each antenna beam. Table 4 presents these values.

Before calculating the cross section and power ratio for a specific satellite it is necessary to estimate the dimensions of the satellite and ealculate the power that would be radiated from an object of this size, assuming it were located at a point in space corresponding to the satellite position (altitude and east-west location). Subsequently, when the true measured power is determined using the actual received signal amplitude reflected from the satellite, the ratio of calculated power to measured power gives the power ratio. Since all the reflected power readings from one satellite are compared to the calculated value for that satellite alone, we are permitted to examine the individual power ratios as a composite group, regardless of the satellite from which they were determined. In other words, a power ratio of 1 indicates that the measured power equals the calculated power, regardless of the physical size of the satellite involved. Figures 115, 116, and 117 present the power ratios measured in the center, north and south antennas, respectively, and Figure 118 shows all the ratios, regardless of antenna.

^{*} A power ratio value of 1 indicates that the measured power equals the calculated power; a value of 10 indicates the measured power equals 1/10 of the calculated power.

TABLE 4

CRUSS SECTION AND POWER RATIO SUMMARY

South Antenne	Power	Ratio	•	•	•	•		ı	•		,	•			ı	ı	ı	•	•		•	13.1	•	13.0		13.5	2.6		1.6	
South	Cross Section	Sq. Ft.	•	•	•		•				•	•	•			•		•	•	•	•	24.9	•	25.1		24.2	0.N	•	198.6	•
North Antenna	Power	Diam'r.	•	•	ı					•						•			• (٥.٥			•	I	·:		' (* ·	ο, α	•
forth	Section So. Pt		• (• 1	•		•	•	•	•	•	•	ı	•	•	•		- 6		1 1	1 4	1	75 7	-	•	27.7	יייייייייייייייייייייייייייייייייייייי	66.6)
Center Antenna	Power Ratio*		6	9	4.4	10.3	7	1	0 -	0 00	7	- '9) ()	, 0	יא ו יא ו	\ _) (10.7		20.2	•	25.8	14.2		14.7			1.8	7. 7	
Cross	Section Sq. Ft.	Rerun**	86.0	7.0	68.5	25.6	87.0	27.1	28.2	15.1	40.5	49.0	178.2	248.0	87.1	. 292 . 0	22.9	38.0	•	16.2	•	75.51	7.1	•	22.3	•	•	180.0	74.5	
¢	Number	4489	640/	0001	926	Oppo Oppo	8683	8719	877 877	8795	6006	9172	9255	9286	9466	27.72	9503	9581	2106	9710	2,1 cc	9000	2000 0.100	7 a g	5.00	3903	9921	- 1000	5466	
מ+;[[מ+מצ	Name	58 Delta																												

TABLE 14 (Continued)

Antenna	Power		. ' 양, ' ' '	· · · · · ·	•
South Antenna	Cross Section Sq. Ft.		53.5 - - Rerun**	97.0	1
Antenna	Power	2.5 2.7 2.5 4.5 5.7			ı
North	Cross Section Sq. Ft.	140.0 2.3 31.8 10.2 43.4 7.5 - 96.7 3.4 14.0 23.3	11111		ı
ntenna	Power Ratio	11.1	26.6 25.9 18.8 38.0 14.7 46.3 70.2 9.8	5.2 24.7 11.6 15.4	5.1
Center A	Cross Section Sq. Ft.	29.6	26.6 18.8 14.7 70.2 Rerun**	132.0 Rerun** 27.7 59.3 51.2 151.2	225.6
	Rev. Number	9959 9975 9991 10001 10007	53 121 414 478 578	14 241 241 556 87 887	183
	Satellite Name	58 Delta	59 Epsilon	59 Zeta	59 Карра

TABLE ! (Continued)

South Antenna	cer Section Power			20.3 16.1 178.2 17.0 19.2 20.2 16.2 16.2 69.8 4.7
in tennu	Power Satto		1 1 1	35 44448 34 40448
iorth.	Revuise			20.3 302.3 302.3 303.3 308.2 37.2
Antenna	Power Ratto 6.1 11.4 7.9 9.5	2 NO 2 NO	2.1 3.11	19.2 12.7 52.7 52.5 5.3
Cross	Section Power Sq. Ft. Ratio 102.2 6.1 6.1 11.1 11.9.8 1.6 Rerun*** 72.9 9.1	400 400 700 400 400	. 118.0 97.1 32.4	35.9 - 53.1 20.9 10.6 71.5
	Rev. Number 96 138 278 685 1285	318 403 418		50 61 117 124 140 156 165
	Satellite Name 59 Lambda	60 Gamma 1	60 Gamma 2	60 Delta

TABLE 14 (Continued)

South Antenna	Power Retio	ল ল 	0, 14, 11111		1 1
South	Cross Section Sq. Ft.	296.1	5.47 5.47	1 1 1	, 1 1
cotenna	Power Ratio	17.0		, , ,	
North /	Section Power Sq. Ft. Ratio	တ္ (၁)	303.6 121.2 - 314.3	ı ı ı	
Antenna	Section Power Sq. Ft. Ratio	1.4.4.	23.5 23.5 23.5 23.5 5.5	2.9	1.4 23.1
Center	Cross Section Sq. Ft.	303.7 81.0 206.1	23.7 14.0 14.0 96.2 6.2	112.0 23.7 82.5	227.0 14.2
	Rev. Number	53 150 286 522 522	106 137 153 194 305 305 617	. 280 280 569	265 301
	Satellite Name	60 Epsilon 1	60 Epsilon 2	60 Epsilon 4 60 Epsilon 4 60 Epsilon 5	60 Epsilon 6

*Calculated power divided by measured power.

**Not calculated since original signal not acquired by ALO. Pass was recovered from magnetic tape playback and original signal strength calibrations are not valid.

It is noted that 45% of the ratios in Figure 118 occur between 1 and 5, and that slightly more than 80% are between 1 and 15. This means that almost half of the reflections possess a power value which is 1/5 of the calculated value or larger, while 8 out of 10 reflections are 1/15 the calculated value or larger.

An interesting situation exists in connection with the cross sections measured for 60 Delta (see Table 4). Of the eight passes received, six passes were three-antenna reflections, offering an excellent opportunity for comparison of cross sections as measured by the different antennas. It is also noted that, of the six three-antenna passes, five passes exhibited the largest cross section in the north untenna, and four passes were almost equal to the calculated value, as indicated by the power ratio approaching unity (Revolutions 124, 140, 156, and 165). No clear cut explumation can be presented for this preponderance of large cross sections in the north antenna. All the antennas were identical in configuration, dimensions and operating specifications, and all were oriented with reference to a first-order geodetic survey. Subsequent to installation, a signal generator was mounted in an airplane and a series of flights were made over the antenna field at each station. In this manner, the radiation patterns and antenna alignments were measured and determined to be optimum. Thus, it would appear that each of the antennas should "see" a satellite in the same way, and any variance in apparent size from one antenna to another would be purely random, dependent solely on such variables as satellite altitude, east-west location, and attitude. The observed cross sections for 60 Delta do not appear random, however.

Further examination of the data, specifically Table 5 which presents average values for all cross sections and power ratios, indicates that in three of the four instances where a comparison can be made between the three antennas for one satellite (58 Delta, 60 Delta, 60 Epsilon 1 and 60 Epsilon 2), the largest average cross section is that measured by the north antenna. Since it appears that the north antennas consistently produced larger power and cross section values, we might conclude that the north antennas were perfectly aligned, or possessed greater gain than either the center or south antennas.

TABLE 5

AVERAGE VALUES OF CROSS SEXTIONS AND POSTE BATICS

				Center Antenna	Pover	North Antenna Cross Power	Power	South Antenna	done d	All Anternas Cross Fowe	POWET
Satellite Name	Number	of Obse	Number of Observations Center North South	Section Sq. Ft.	Antio	8 6 6 6 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	Sec. to	Sec. 192	01:4:	Section Sq. Ft.	Fatto
58 Delta	23	10	\0	73.2	6.8	0.17	5.3	62.6	6.0	72.5	8.1.
59 Epsilon	- ‡	O	٦	3.5	30.0	•	•	93.5	12.9	%.e	36.6
59 Zeta	5	0	٦	84.3	11.9	•	•	0.72	### 	86.h	11.1
59 Карра	1	ဂ	0	225.6	5.5	•	•	•		525.6	7.7
59 Lambda	S	0	~	112.0	7.3	•	•	53.6	20.5	9. Q.	4.0
60 Gums 1	2	0	0	63.9	ù	•	•	•	•	63.9	9.4
60 Glemit 2	ત	0	0	32.4	10.1	•	•	•	•	3.5%	10.3
60 Delta	9	7	හ	67.0	23.4	185.9	79	65.5	6.3	115.0	ដ
60 Epsilon l	М	٦	CI	196.9	2.2	18.8	17.0	296.6	e .	200.5	a
60 Epsilon 2	'n	8	٥	8.05	23.5	246.4	3.6	92.8	2.7	107.9	12.9
60 Epsilon 3	ч	O	0	0.511	5.9	•	•	•	•	112.0	5.9
60 Epsilon 4	Т	0	0	23.7	13.8	a	•	•	•	23.7	13.8
60 Epsilon 5	ч	0	0	82.5	3.9	ı	•	•	t	82.5	3.9
60 Epsilon 6	α	0	0	120.6	12.3	ı	•	•	ı	120.6	12.3

As soon as we make this tentative assumption, however, we are faced with contradictory data as observed for 60 Epsilon 1. Table 5 shows six passes of this satellite, three measured by the center antenna, two by the south and one by the north. Excluding the north antenna page for the moment, the five remaining passes combined have an average cross section of 256.8 square feet and m. average power ratio of 1.8. These values represent the largest average cross section and best average power ratio of any satellite observed, yet the one pass of this satellite received by the north antenna has a cross section that is smaller than this average by a factor of 12. The fact that we observed this one small cross section in the north antenna is not significant, since a single observation cannot be considered statistically meaningful. However, it is significant that the average cross sections for the five passes received by the center and south antennas are large, nearly equal to the calculated value. Why these two antennas operated so excellently in this she satellite is a matter of conjecture.

Referring again to Table 5, we note that the south antenna produced average cross sections that were larger in 5 out of 7 instances than those observed by the center antenna for the same satellite. Here again, it must be pointed out that 2 of the 5 are based on only one pass each (59 Epsilon and 59 Zeta), and no valid conclusions can be drawn from such a small amount of data.

V. CROSS SECTION "SIGNATURE" OBSERVATIONS

When the signal strength reflected from a number of passes of one satellite is recorded in analog form, the shape of the observed cross section envelopes may appear very much alike, provided that the satellite attitude remains reasonably constant from pass to pass and that propagation conditions are similar. Under ideal conditions, a particular satellite may consistently produce a unique "signature", thus permitting identification on this basis alone. Others may produce a wide variety of shapes, which appear to possess nothing in common. A satellite with dimensions small compared with the wavelength will generate an approximately semicircular or semi-elliptical signal strength pattern that is independent of physical configuration. Signal strength changes due to scintillation and attitude change (spin and tumble) are usually present with large satellites which produce a very complex received signal envelope configuration and make "signature" identification difficult.

In general, the DOPLOC records obtained by the center antenna are characterized by a short duration, symmetrical envelope that rises and decays smoothly and possesses a rounded peak, with little or no oscillation visible at any time. The average duration and signal strength of the 67 center antenna measurements are 7.5 seconds and -169.4 dbw. The north and south antennas produced two general shapes; either a semi-circular configuration with a smoothly changing signal level or a semi-rectangular envelope with an abrupt rise and decay and a reasonably constant signal level. Considerable variations, some quite large in amplitude, are usually visible in both types. Signals from the north and south antennas were longer than those from the center antenna, since the beams were directed 20 degrees above the horizon as opposed to the vertical center beam. For the 22 observations in the north antenna the averages are 24.8 seconds and -170.4 dbw, while in the south antenna the averages are 15.7 seconds and -173.6 dbw. The overall averages for the 44 reflections are 20.3 seconds and -172.0 dbw. A discussion of several specific "signatures" follows.

58 Delta - An outstanding example of "signature" is evident in Figure: 37 and 40, where a sharp null is prominent on each record approximately 10 seconds after the appearance of the Doppler signal. In both cases the reflection is in the north antenna and the length of record permits easy observation of this characteristic null. A severe signal dropout is evident in Figure 25 and, to a lesser extent, in Figure 22. This may be an indication of "signature". Similarly, a small dip is seen in each of the south antenna portions of Figures 31, 35, 36 and 38, approximately 3 seconds prior to signal loss.

59 Epsilon - There is no clear Indication of "signature", despite the prevalence of periodic change in signal level, especially in Figures 48, 50 and 52. This change in signal amplitude is probably caused by satellite attitude change rather than by satellite configuration.

59 Zeta - A sharp spike is easily visible on the leading edge of each of the envelopes in Figures 53, 55, 58 and 59. The latter two records show a periodic change in signal strength, whereas none of the other 59 Zeta records possessed this characteristic. The variations in Figures 58 and 59 are due to attitude change, which apparently commenced (or at least increased in frequency) late in the orbital lifetime of 59 Zeta. Even as late as Revolution 855 (Figure 57), there is no indication of this periodic modulation. It is very evident, however, in Revolutions 871 and 877, the last two DOPLOC observations of this satellite before its re-entry into the earth's atmosphere (estimated by Space Track Control Center to have occurred between Revolutions 964 and 966).

59 Kappa - Only one pass was received for this satellite and it shows no significant features.

59 Lambda - The very abrupt signal loss in the middle of Revolution 96 of 59 Lambda (Figure 61) may be "signature" information, though two factors are present which make this assumption appear somewhat doubtful. First, there is no evidence of this unique signal dropout on any of the other records of 59 Lambda. Secondly, and of greater importance, is the extremely rapid signal decay. It is unlikely that any motion or physical configuration of the satellite could produce such a steep decay curve. The almost instantaneous decay strongly suggests that equipment failure or propagation phenomena, rather than satellite movement or shape, caused this sudden loss of signal.

60 Gamma 1 - Another excellent example of "signature" is found in Figures 69, 70, and 71, where the V-shaped peak is easily seen. Figure 67 also shows traces of this shape, but the high altitude of this pass (hho miles) severely attenuated the received signal and prevented this characteristic from being more prominent. This pass was the highest pass recorded by the DOPLOC system during its operation. A further identifying feature in these data for 60 Gamma 1 appears to be the signal losses shown in Figures 68, 69, and 70 which occurred in the middle of the passes where the signal is normally at a maximum. On two of these passes, the tracking filter lost lock completely, but the ALO was able to lock-on again when the signal level increased.

60 Gamma 2 - Only one pass was recorded for this satellite and it displays no significant envelope shape.

60 Delta - Here we see a severe null in both Figure 78 (center) and Figure 79 (north). Once again, the tracking filter lost lock during the former pass but the ALO regained the signal automatically when it became stronger. A hint of a similar null may be visible in both Figure 73 (south) and Figure 78 (south). It is worthwhile to point out that Revolution 61 (Figure 74) was received by the north and south antennas but not by the center antenna. Perhaps the satellite was oriented in its maximum null-producing attitude during the middle portion of this pass and, consequently, this orientation reduced the reflected signal to a level lower than the ALO threshold level and no lock was obtained. Repeated rerunning of the magnetic tape of this pass in an attempt to lock the ALO on the Doppler signal was without success. It appears that the center antenna portion of Revolution 156 (Figure 78) may have been approaching this complete signal fadeout condition also.

60 Epsilon 1 - There does not appear to be any clear indication of "signature" in these data. The record of revolution 150 (Figure 83) exhibits a nine second signal loss in the middle of the pass. The absence of signal dropout in the other records of 60 Epsilon 1 makes it difficult to correlate these data with "signature" information. It may be noted that data shown in Figure 84 was recorded at 1 mm/second which served to

condense the envelope. The 2.5 mm/second chart speed would have enlarged the envelope and produced a trace similar to Figure 85.

60 Epsilon 2 - The distinguishing feature here seems to be the oscillations observed in the center antenna portions of Figures 88, 92 93 and 95. Reflections from the majority of other satellites observed in the center untenna are smooth. Again it is noted that the chart speed in Figures 92 and 93 is 1 mm/second, while in Figures 88, 89 and 95 it is 2.5 mm/second. Had the former passes been recorded at 2.5 mm/second, the resulting envelope would be very similar to Figure 95.

of Epsilon 3. 4. 5 and 6 - It is not possible to evaluate individually these data in terms of "signature", since only one or two passes of each satellite are available and none seems to possess any distinguishing features. Intercomparison is not valid either, even though they are all fragments from 50 Epsilon 1 (Sputnik IV), because the physical configuration of the fragments is probably not the same and, therefore, any similarity in envelope shape would be purely random.

VI. CROSS SECTION MODULATION DUE TO ATTITUDE CHANGE

Evidence of periodic signal modulation, such as might be caused by spin or tumble, is wisible in the majority of signal strength records. This cross section modulation is visible both in the records of groundoriginated reflected signals and in the records of signals which originate from a satellite-borne transmitter. For this discussion, we shall term the former method of satellite detection and observation "pussive" tracking, and the latter method "active" tracking. Measurement and analysis of signal strength modulation observed with the two methods of tracking are difficult for several reasons. Propagation variances alone may introduce periodic changes in observed signal strength which can be mistaken for modulation caused by satellite attitude change. The relatively short duration of the passive records recorded by the center antenna make it impossible to determine a modulation periodicity in this antenna of more than a second or two. The passive records recorded by the north and south antennas, though fewer in number, are of greater value in modulation analysis since they are considerably longer in duration and permit measurement of several cycles of a modulation possessing a period in the order of 1 cycle per 5 seconds or longer. Also, when active tracking records are examined, especially those taken on Sputnik III and IV which transmitted on a nominal frequency of 20 Mc/s, the Faraday effect must be considered. This effect varies inversely with the reciprocal of the frequency squared. Thus, it is definitely a prime factor in producing periodic signal modulation at 20 Mc/s, considerably less a factor at 108 Mc/s (the DOPLOC reflection frequency), and virtually non-existent at still higher frequencies such as the transmitting beacons in the Discoverer satellites. Fluctuations in received signal, while not necessarily periodic, may also be caused by changes in the amplitude of the transmitted signal or the gain of the receiver. Within these limitations, an attempt has been made to analyze the DOPLOC records for evidences of cross section modulation caused by satellite attitude change.

Examination of passive and active records for 58 Delta (Sputnik III) and 60 Epsilon 1 (Sputnik IV) show signal strength nulls occurring in a 2:1 ratio, i.e., two nulls are observed on the passive records for every

null on the active records. Figures 29 (north), 33 (south) and 39 (north) are records of passive reflections from 56 Delta and show an average of 1 null per 3 seconds, while Figure 119, an active record for 58 Delta, displays 1 null per 6 seconds. Similarly, Figures 32 (north), 83 (south), and 85, passive observations of 60 Epsilon 1, show an average of 1 sull per 1.5 seconds, and an active record of the same satellite (Figure 120) shows i null per 3 seconds. It is evident from these data that the physical configuration of the satellite produces a four-lobed pattern when reflecting the ground-originated DOPLOC signal, whereas the transmitting antennas on the satellite radiate a two-lobed pattern for active tracking. As the satellite tumbles and spins, these radiation patterns produce the 2:1 ratio seen in the signal strength null frequency.

Evidence of similar periodic nulls is seen in the records of 59 Epsilon (Discoverer V). Here, however, the passive to active ratio is 5:1. Figures 48 (south), 50 and 52 (south) are passive reflection records of 59 Epsilon and display an average of 1 null per 3 seconds. Each of two active records for 59 Epsilon, one of which is shown in Figure 121, show 1 null per 24 seconds. Apparently, the cylindrical shape of the Discoverer satellite creates a multi-lobed reflection pattern, while the transmitting antenna radiates the two-lobed pattern. The validity of this assumption regarding the reflection pattern is enhanced by the fact that the length of the cylinder (19.2 feet) is approximately twice the wavelength of the DOPLOC frequency (9.1 feet). Where the length of the reflecting object is large compared to the wavelength, as in this case, the result is a multi-lobed radiation pattern.

It is worthwhile to note that a total of nine active records taken on 59 Zeta and 60 Delta (Discoverer VI and XI) show an average of 1 null per 24 seconds, with individual values ranging from 11 to 40 seconds. The agreement with the null rate observed for 59 Epsilon is striking.

Table 6 summarizes the data discussed above and lists null rates for several other satellites as well.

TABLE 6

SIGNAL STRENGTH NULL RATES

	Pas	Passive	Active	ve	
Satellite	Number of Observations	Avg. Time Between Nulls secs.	Number of Observations	Avg. Time Between Nulls secs.	Passive-Active Null Ratio
Sputniks					
58 Delta	8	Ю	г	\9	2:1
60 Epsilon 1	8	1.5	2	2	2:1
60 Epsilon 2	. 9	1.2	(no tra	transmitter)	ı
Discoverers					
59 Epsilon	м	ĸ	C)	201	8:1
59 Zeta	Q	Q	7	214	12:1
60 Delta	a	1.8	CV	24	13:1

VII. SCHMPILLATION

A number of the DOPLOC records of reflected signal strength show traces of scintillation and Table 7 presents a summary of the most prominent examples, grouped according to pass time. The estimated peak-to-peak scintillation in db is given for each pass, as well as the antenna in which the reflection was received. A notation is also made concerning the season of the year in which the pass was recorded. Certain qualitative features in the data are rather pronounced and deserve additional comment.

The existence of a diurnal cycle is apparent, with the scintillation amplitude being greater at night than during the day. The average peak-to-peak signal fade for 9 observations made at night is 2.4 db, while for 20 daytime observations the average value is only 1.4 db. Also, the diurnal effect is much more pronounced in summer than in winter, evidenced by the fact that 8 of the 9 night passes occurred in the months between March and October. A series of one-way radar transmission tests conducted by the Bell Telephone Laboratories, at wavelengths ranging from 30 cm to 17 m, revealed corresponding diurnal and seasonal variations. Similarly, tests carried out by Ross A. Hull, with the aid of other radio amateurs, the U. S. Weather Bureau and Harvard University in the 5 to 8 m wavelength region, yielded comparable results. 5, 6

The data in Table 7 show that scintillation occurs more frequently in the north and south antenna reflections than in the center antenna. Considering the total number of reflections received by each antenna, we find scintillation appearing in 45% of the north antenna passes (10 of 22), 41% of the south antenna passes (9 of 22) and only 15% of the center antenna passes (10 of 67). The indication here is that the reflecting and diffracting properties of the terrain and the irregularities of the atmosphere near the earth's surface combine to produce interference waves which, in turn, cause variation in signal strength. These phenomena would heavily influence reflections in the north and south antennas, since they are beamed only 20 degrees above the horizon. The

TABLE 7 SCINTILLATION VS PASS TIME

			Penk-to-Penk Scintillation	• • • •	a Time	
Satellite Name	Revolution Number	Antenna	in db	υ. τ.	C. S. T.	Schaon
60 Epsilon 2	505	¢	2.0	0355	2155	s
60 Delta	117	# MA 17E	2.0	6628	0028	\$
60 Epstlan 2	194	38	3.0	0642	QC42	\$
58 Dolth	9957	*	2.0	octs.	COV5	\$ 5
58 Delta	9951	G	2.0	CAAR	6042	S S S S
98 Delta	bins	c	2.0	(652	0052	y G
53 Delta	9905	43	3.0	0700	2100	s <u>ទ</u>
60 Epsilon 2	147	36	2.5	0/21	0121	S
58 Delta	955.7	×	3.0	0849	0249	S
	#this	.4	2.0	1526	0726	S
60 Epsilon 2	50) 0mas	đ C	2.0	1 558	0758	w
98 pelta	8795	C	1.5	1460	රජිරව	3
60 Epsilon 6	501	e e	:.0	1454	(85)4	¥
58 pelta	87%		1.0	1504	0904	¥
5) Lambda	278 640 6	8	2.0	1550	0930	W
98 Delta	8719	C X	1.0	1548	0948	s
98 Delta	9959		1.0	1605	1003	s
98 Delta	994.5	8	1.0	1647	1047	1
98 Delta	8045	c s	1.0	1650	1050	w S
5, Zeta	150	S	1.0	1735	1155	S S K K
60 EpsΩon 2	106		1.5	1803	1205	s &
60 Delta	61	×	1.5	1812	1212	s I
98 Delta	9848	S	2.0	1813	1213	s
58 Delta	9852	s 	2.0	1840	1240	s
60 Deltu	3 0	¥		2045	1445	s
58 Delta	9722	8	1.5	2136	1536	ų l
58 Delta	8386	С	1.0	2218	1618	s s
60 Epsilon 2	617	С	2.0		1619	S
58 Delta	9612	31	1.0	2219	1816	¥
59 Lambda	685	S	2.0	0016	1010	H

^{*}S - south, C - center, II - north
**W - winter (November - February)

^{**}S - summer (March - October)

greater length of the observations by the north and south antennas may also contribute to the more frequent observations of signal fade in these antennas.

From an examination of a number of active tracking records made at Aberdeen Proving Ground, Maryland, there appears to be more scintillation present in these active records than in the passive reflection records. Figure 122 shows a Doppler record of 58 Delta obtained at Aberdeen by tracking the 20 Mc/s satellite-borne transmitter. Peak-to-peak signal variations for this record are approximately 6 db, roughly twice as large as any variations noted in the passive records. This satellite was probably experiencing severe buffeting when this recording was made, since re-entry occurred only 30 revolutions later. These drastic oscillations may have increased the scintillation amplitude to some extent, since records taken earlier in the lifetime of 58 Delta show signal fade in the order of 3-4 db. Then too, the very slight Doppler frequency shift during this pass indicates that the satellite was at a great distance from the station and that the inclination of the line of sight to the horizontal was small. Thus, as mentioned previously with respect to scintillation in the north and south antennas, interference waves may have contributed to the increased amplitude of the signal variations.

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APPENDIX I

CALCULATION OF POWER RATIO AND CROSS SECTION

Altitude And Ground Range Determination

Center Antenna - As previously stated, the Doppler frequency va time data are digitally recorded at the receiving station us the satellite passes through each of the three antenna beams. From the digital Doppler data we calculate the Doppler slope (rate of charge of Doppler frequency). Ensuing the Doppler frequency and slone and assuring circular notion, it is possible to obtain the altitude and east-west location of the satellite as it crosses the base line joining the transmitter and receiver, with the aid of a chart of the type presented in Figure 125. Constant contour lines have been computed and drawn for the Doppler frequency and the time derivative of the frequency (the slope) in the vertical prane containing both the transmitter and receiver. These contours wary with satellite orbital inclination, so that several charts are required for orbits of various inclinations. Each quart is valit, however, only for data received in the center beam. Given a Dappler frequency of 300 cps and a slope of 90 cps at the midpoint of the record, we locate the curves representing these values on the chart and, at their intersection, read an ultitude of lyV miles and a sub-satellite point on the base line located 350 miles east of Fort Sill. Having determined these distances, a sketch similar to Figure 124 is prepared to assist in completing the power and cross section calculations.

North and South Antennas - The geometry and calculations necessary to locate the point of interaction of a satellite in the north or south antenna, and its corresponding ground range from Fort Sill, vary somewhat from the method described for the center antenna. In these antennas, the intersection point and its corresponding distance east or west of Fort Sill are functions of satellite altitude, and the inclination of the orbital plane and base line with respect to the equator. For purposes

of this discussion, we shall confine ourselves to the south antenna. Calculations in the north antenna are similar. Figure 125 gives a graphic presentation of the geometry involved.

The south entenes is elevated 20 degrees above the horizon and oriented in azimuth as shown in Figure 11. The altitude of the satellite intersection point in the center antenna is taken as the height of the point of intersection in the south antenna pattern. The curvature of the earth is ignored here since it is negligible over the relatively short distance involved. Referring to Figure 125, we calculate X, the perpendicular distance from the base line to the point of intersection of the satellite in the south antenna, using the trigonometric function

Subsequently, the perpendicular distance Y, from the 20 degree plane of the south antenna to the 90 degree plane of the center antenna is determined by

$$Y = \frac{Z}{\tan 20^{\circ}}$$

The next step is to calculate $\hat{\varphi}_i$, the angle at which the satellite crosses the base line. If we let i equal the orbital inclination at the equator and i' the orbital inclination at any North Latitude β , then

$$\cos t' = \frac{\cos t}{\cos \phi}$$

Solving for i' gives the angle at which the satellite crosses the particular North Latitude β . The Forrest City - Fort Sill base line is a segment of a great circle which is not parallel to the equator; hence as shown in Figure 126, the satellite crosses the base line at an angle different from i', the difference being equal to the inclination of the base line to the equator. The base line is inclined 3 degrees to the equator so, for a north-south pass 3 degrees must be added to i', while for a south-north pass 3 degrees is subtracted from i' to obtain θ_i .

Finally, knowing θ_i and Y, the distance d moved along the base line in Figure 125 can be determined by the trigonometric relation

$$\tan \theta_i = \frac{Y}{d}.$$

This value d must be added to or subtracted from r (the ground range from Fort Sill as determined for satellite intersection in the center antenna), depending on the pass direction and the antenna involved. As shown in Figure 127, for a north-south pass, ground range in the north antenna is equal to r - d, while for the same pass in the south antenna the ground range is r + d.

Having thus determined the distance from the base line to the satellite intersection point in the 20 degree plane and the corresponding ground range from Fort Sill, a sketch is prepared as shown in Figure 128 to aid in completing the calculations.

Power and Cross Section Calculations

Having determined satellite altitude (or the perpendicular distance from the base line to the point of intersection in the 20 degree plane), as well as the ground range east or west of Fort Sill, and knowing that the distance between the transmitter and receiver is 435 miles, the distances R_1 and R_2 and the angles α and β can be computed using standard trigonometric functions (refer to Figures 124 and 128). The azimuth angles of the transmitting and receiving antennas are known (see Figure 11); hence the angles γ and δ can be determined and, from the curve presented in Figure 129, the factors F_T and F_R , which are decimal representations for the angles γ and δ , are obtained. It is now possible to proceed with the computation of the calculated power P_R , the measured power received P_M , the power ratio P_R/P_M , and the apparent cross section of the reflecting object \sum .

The calculated power in watts that would be radiated from an object of known dimensions at a given distance from the transmitter and receiver is determined first. For example, a length of 20 feet and a radius of 2.5 feet is assumed for the Discoverer satellite. The maximum reflection

cross section that an object having these dimensions would present is calculated by using the formula 1

$$\sum = 5\pi \frac{y}{r \tau_5}$$

where A at 108 Me/s is 9.1 feet. Solving for E, a value of 690 square feet is obtained for the Discoverer satellite. Admittedly, this process is an approximation, since accurate determination of \(\sum_{is} \) dependent upon r and ℓ being large compared to λ . Similar estimates of physical dimensions were made for Sputnik III and IV and for Transit I, and the resulting values of Twere used in the calculations pertaining to those satellites. Using the basic radar equation,

$$P_{R} = \frac{P_{T} G_{T} G_{R} \lambda^{2} F_{T} F_{R} \sum}{(4\pi)^{3} (R_{1})^{2} (R_{2})^{2}}$$
(1)

the value of \(\sum_\) and the following DOPLOC system constants are inserted:

$$G_{\rm R} = G_{\rm T} = {\rm antenna} \ {\rm gain} \ {\rm in \ power} = 40$$

$$\lambda = 9.1 \text{ feet}$$

Thus, equation (1) becomes

$$P_{R} = \frac{1.86 \times 10^{8} F_{T} F_{R}}{(R_{1})^{2} (R_{2})^{2}}$$
 (2)

where
$$R_1$$
 and R_2 are expressed in feet or
$$P_R = \frac{2.41 \times 10^{-6} F_T F_R}{(R_1)^2 (R_2)^2}$$
(3)

where R₁ and R₂ are expressed in miles.

Next, equation (3) is solved for P_R to determine the power in watts that would be radiated from an object of this size if it were located at the point in space occupied by the satellite.

To determine the measured received power in watts, the peak of the received signal recorded in db below unity S/N at receiver output is converted to a voltage ratio X.

Ridenour, L., "Radar System Engineering", MIT Radiation Laboratory Series No. 1, p. 66. (1947).

$$X = \frac{\log^{-1} (\max db)}{20}$$

If E₁ is the received signal for a 1:1 S/N ratio at the receiver output (which for the DOPLOC system is 0.07 microvolts) and E₂ is the measured received signal voltage, the measured power in watts is

$$P_{M} = \frac{(E_2)^2}{R}$$

where R, the antenna resistance, is equal to 50 ohms. The power ratio, $P_{\rm R}/P_{\rm M}$ expresses the relation between calculated and measured power.

To calculate the cross section corresponding to the peak signal received, equation (1) is used to solve for \sum .

After substituting P_{M} for P_{R} , equation (1) may be rearranged to give

$$\sum = \frac{P_{M} (R_{1})^{2} (R_{2})^{2}}{3.49 \times 10^{-9} F_{T} F_{R}}$$

Solving for \sum gives the apparent cross section of the reflecting object in square feet.

APPENDIX II

BRL-DOPLOC REPORTS

No. 1

BRI. Memo Report No. 1055 - October 1958
"Doppler Signals and Antenna Orientation for a Doppler System"
by L. P. Bolgiano, Jr.

CONFIDENTIAL

No. 2

BRL Memo Report No. 1185 - January 1959 First Semi-Annual Technical Summary Report Period 1 July 1958 - 31 December 1958

by L. G. deBey, V. W. Richard, A. H. Hodge, R. B. Patton, C. L. Adams

(BML 39-60)

CONFIDENTIAL

No. 3

BRI. Tech Note No. 1265 - June 1959 "Orbital Data Handling and Presentation" by R. E. A. Putnam

UNCLASSIFIED

No. 4

BRL Tech Note No. 1266 - July 1959
"An Approach to the Doppler Dark Satellite Detection Problem" by L. G. deBey

CONFIDENTIAL

No. 5

BRL Memo Report No. 1220 - July 1959 Second Semi-Annual Technical Summary Report Period 1 January - 30 June 1959 by L. G. deBey, V. W. Richard and R. B. Patton

(BML 208-59)

CONFIDENTIAL

No. 6

BRL Tech Note No. 1278 - September 1959 "Synchronization of Tracking Antennas" by R. E. A. Putnam

UNCLASSIFIED

No. 7

BRL Memo Report No. 1237 - September 1959
"A Method of Solution for the Determination of Satellite Orbital Parameters from DOPLOC Measurements" by R. B. Patton, Jr.

UNCLASSIFIED

BRI - DOPLOC REPORTS (continued) No. 3 BRL Memo Report No. 1093 - Murch 1960 "The Dynamic Characteristics of Phase-Lock Receivers" by Dr. Keats Pullen UNCLASSIFIED No. 9 "Station Geometry Studies for the DOPLOC System" Stanford Research Institute UNCLASSIFIED No. 10 Final Report, Part B, Stanford Research Institute - July 1960 "DOPLOC System Studies" by W. E. Scharfman, H. Rothman, H. Guthart, T. Morita UNCLASSIFIED No. 11 Phileo Corporation - 4 May 1960 "Polystation Doppler System" UNCLASSIFIED No. 12 Space Science Laboratory, General Electric Co. - October 1960 "Orbit Determination of a Non-Transmitting Satellite Using Doppler Tracking Data" by Dr. Paul B. Richards UNCLASSIFIED No. 13 Final Technical Report - University of Delaware - June 15, 1960 "Quantum Mechanical Analysis of Radio Frequency Radiation" by L. P. Bolgiano, Jr. and W. M. Gottschalk UNCLASSIFIED No. 14

Final Report F/157, Columbia University - February 11, 1960 "Summary of the Preliminary Study of the Applicability of the Ordir System Techniques to the Tracking of Passive Satellites"

UNCLASSIFIED

BRL-DCPLOC REPORTS (continued)

No. 15

BRL Report No. 1110 - June 1960
"Precision Frequency Measurement of Noisy Doppler Signals"
by W. A. Dean

UNCLASSIFIED

No. 16

Third Technical Summary Report - Period July 1959 through June 30, 1960 BRL Memo Report No. 1267 by A. L. G. delley

UNCLASSIFIED

No. 17

Columbia University Tech. Report No. T-1/157 - August 1, 1959
"The Theory of Phase Synchronization of Oscillators with Application to the DOPLOC Tracking Filter"
by E. Kreindler

UNCLASSIFIED

No. 18

BRL Tech Note No. 1345 - August 1960 "DOPLOC Receiver for Use with Circulating Memory Filter" by K. Patterson

UNCLASSIFIED

No. 19

BRL Tech Note No. 1354 - October 1960 "Parametric Pre-Amplifier Results" by K. Patterson

UNCLASSIFIED

No. 20

BRL Tech Note No. 1367 - December 1960
"Data Generation and Handling for Scanning DOPLOC System"
by Ralph E. A. Putnam

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No. 21

BRL Report No. 1123 - January 1961
"The DOPLOC Instrumentation System for Satellite Tracking" by C. L. Adams

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ERL-DOPLOC REPORTS (continued)

No. 22

BRL Memo Report No. 1330 - March 1961 "DOPLOG Observations of Reflection Cross Section of Satellites" by H. T. Lootens

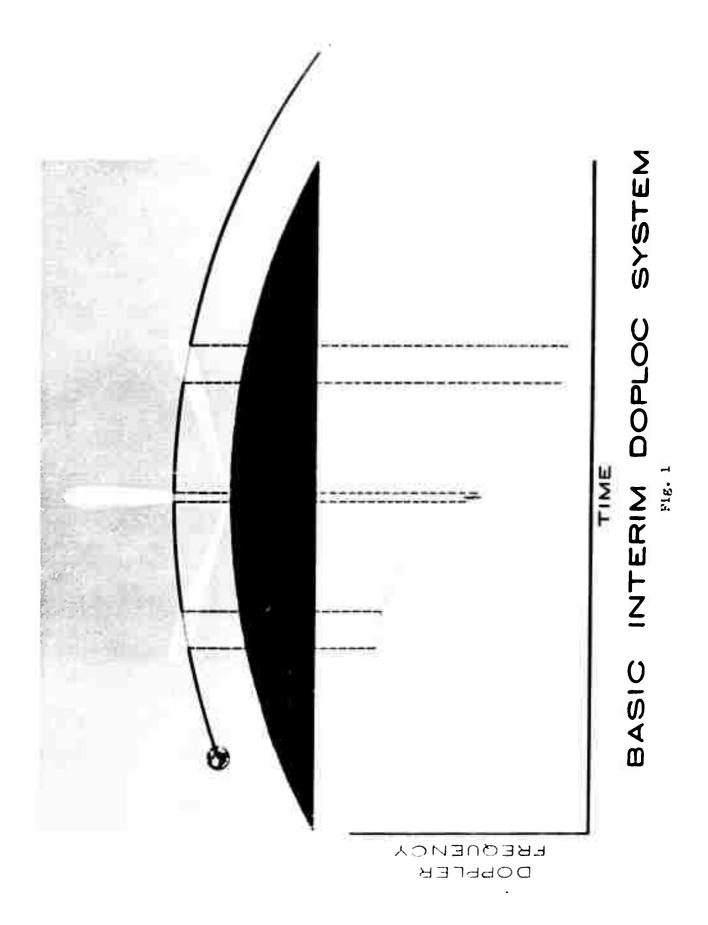
UNCLASSIFIED

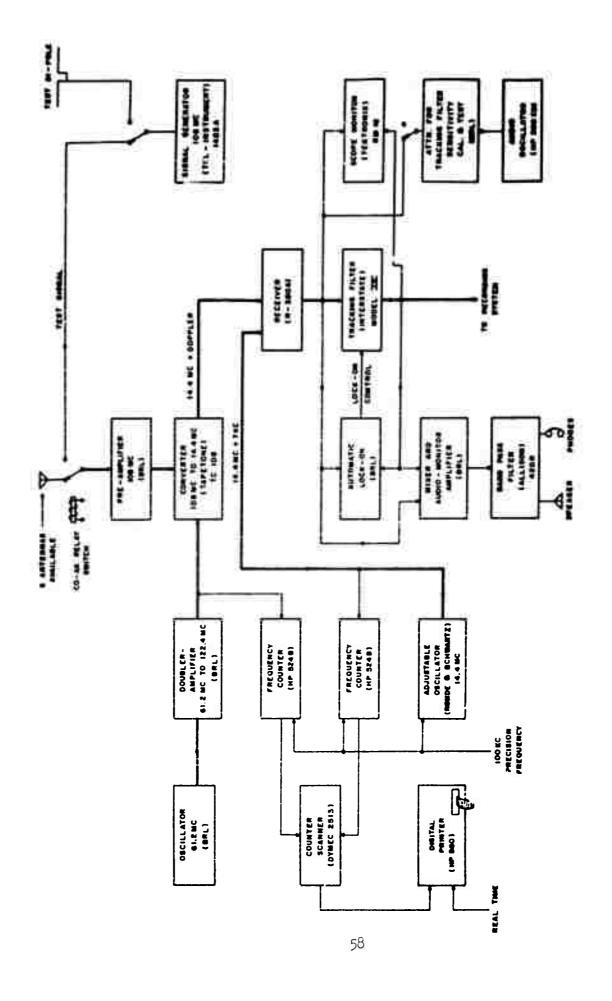
In Preparation

[&]quot;DOPLOC Comb Filter" by R. Vitek

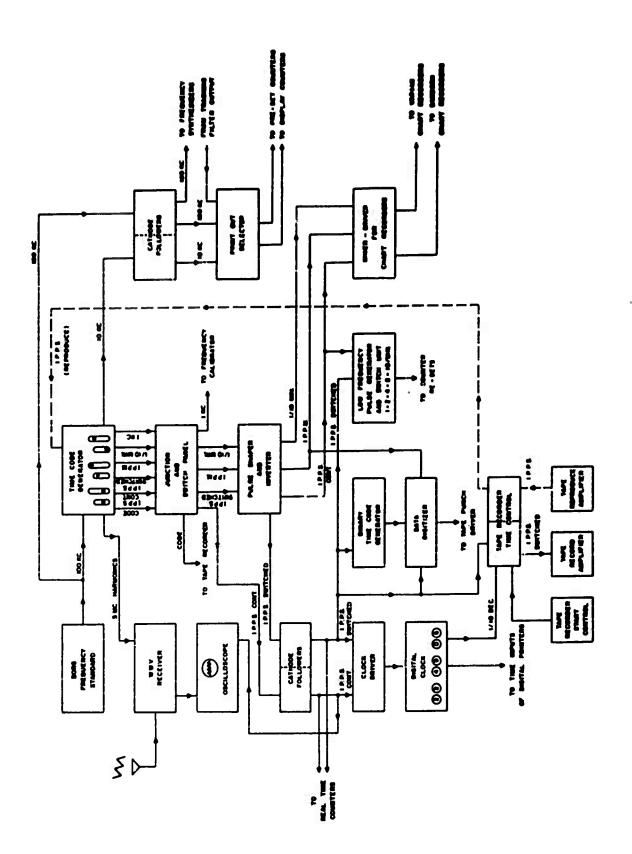
[&]quot;DOPLOC Orbit Determination Methods" by R. Patton, Jr.

[&]quot;Final Summary Report on the BRL-DOPLOC Project" by Dr. A. H. Hodge

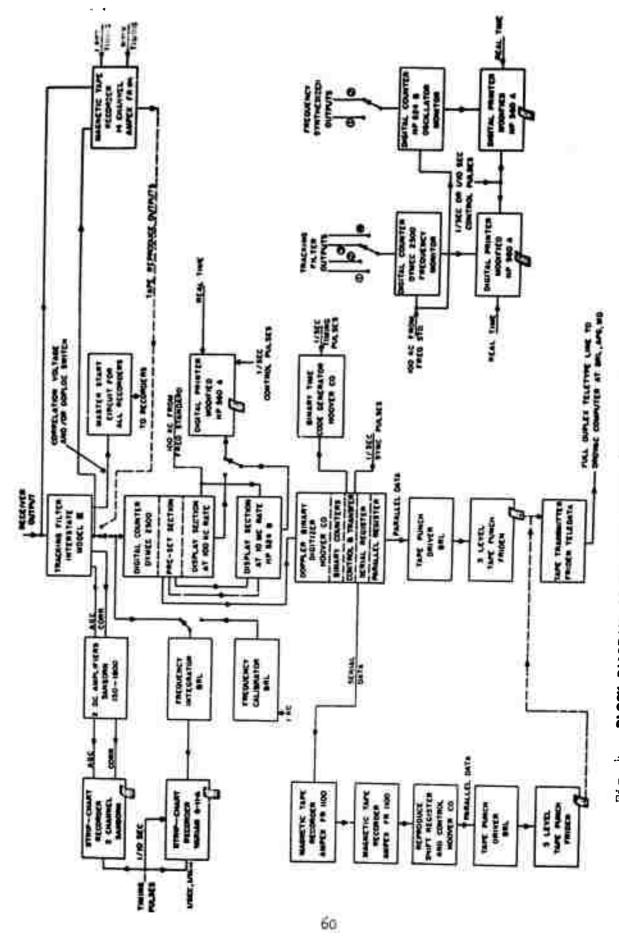




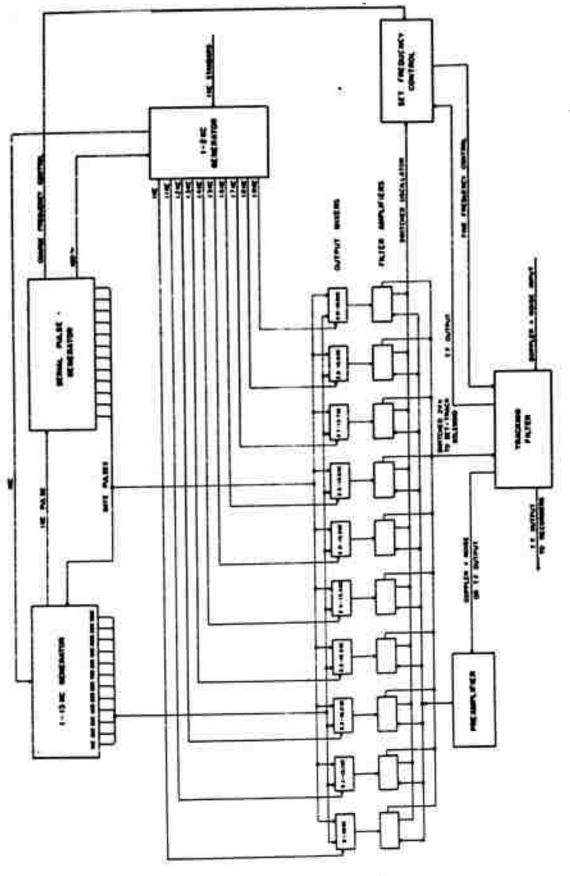
Pie. 2 BLOCK DIAGRAM OF R.F. SECTION IN DOPLOC STATION PASSIVE TRACKING AT 108 MC



59



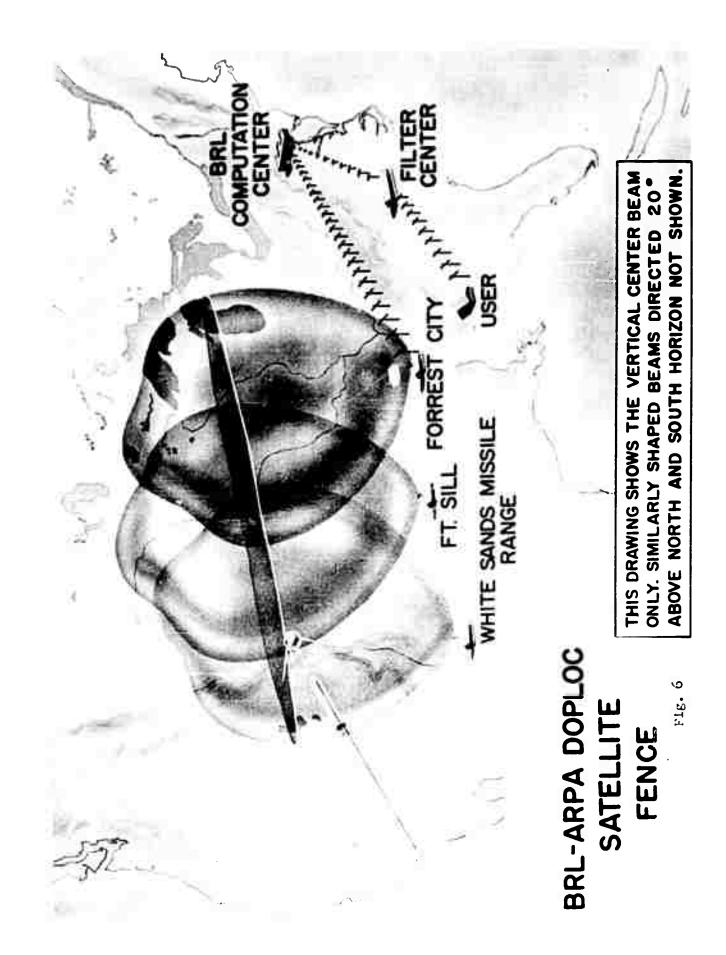
BLOCK DIAGRAM OF DOPPLER DATA RECORDING SYSTEM FOR A SINGLE CHANNEL Fig. 4



61

TYPICAL DOPLOC DATA OUTPUT

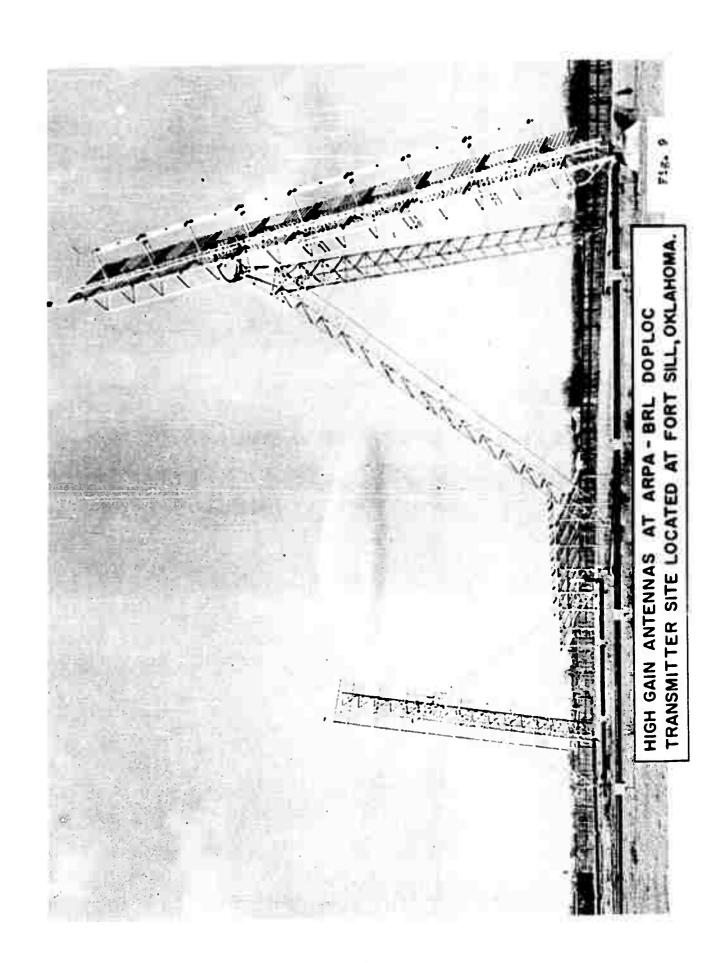
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<u> </u>		PLES

SOLUTION SINGLE PASS DOPPLER ORBIT (BY ARMY - DOPLOC SATELLITE TRACKING SYSTEM)
REVOLUTION NO. 30 OF 1960 DELTA (DISCOVERER XI) APOGEE . 254 DIFFERENCE 0000 -3 ₩ . 29 8 M200 189 M 200 486 M. 00295 80.15* M.W. TYPICAL ORBITAL PARAMETERS STH MAJOR AXIS ECCENTRICATY NCL STATE **80339**



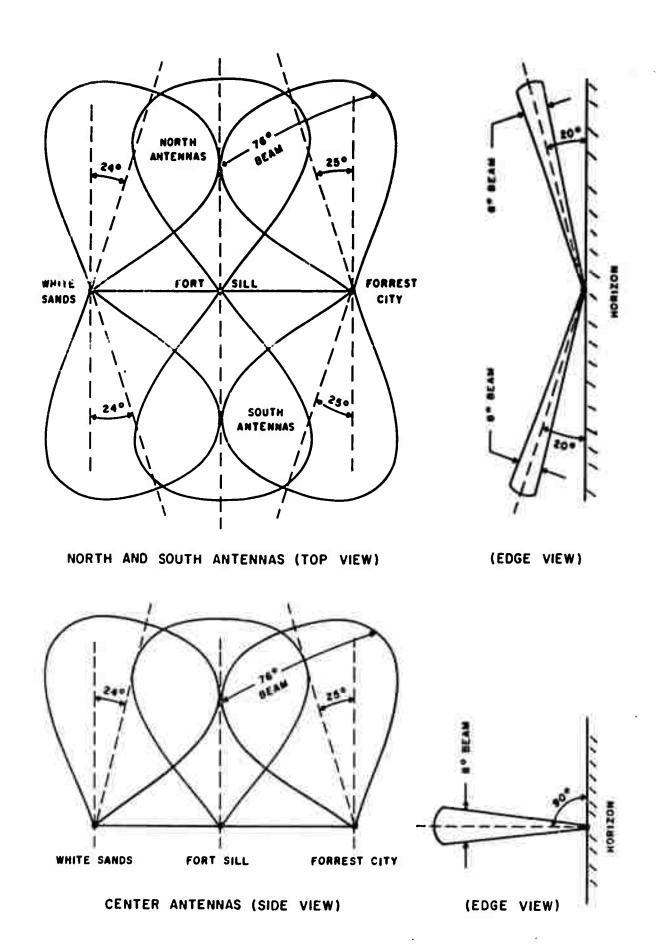


FIG. 10 - INITIAL ANTENNA ORIENTATION

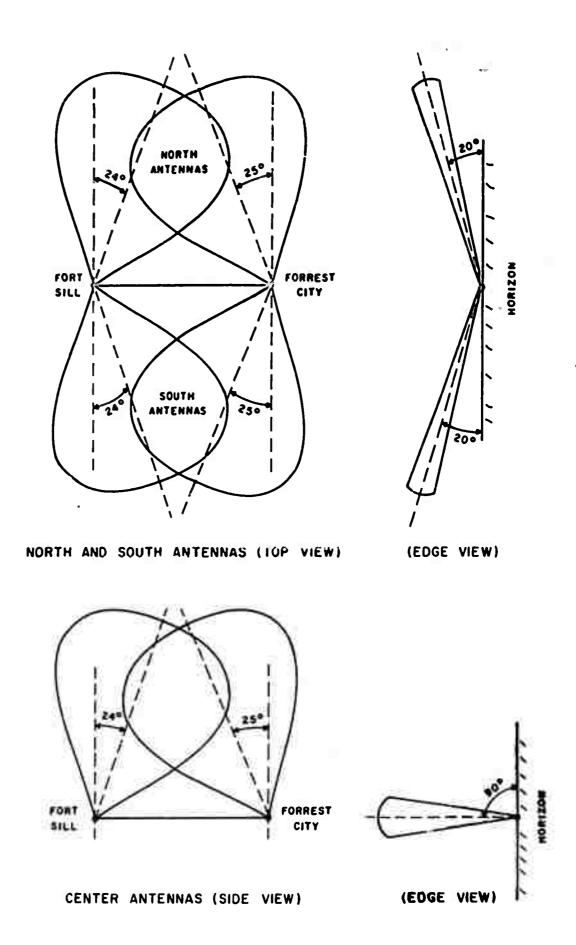


FIG. II - ANTENNA ORIENTATION FOLLOWING

DEACTIVATION OF WHITE SANDS STATION

ARPA-BRL DOPLOC DOPPLER RECORD OF 58 DELTA AND 59 EPSILON FORREST CITY, ARKANSAS

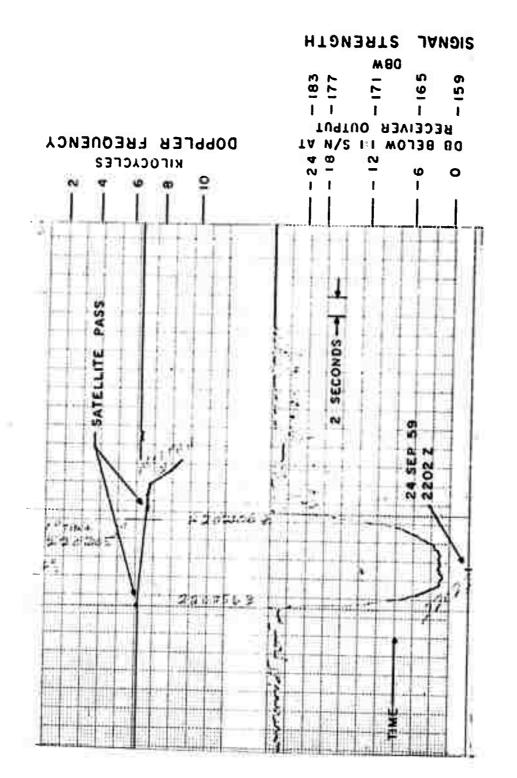
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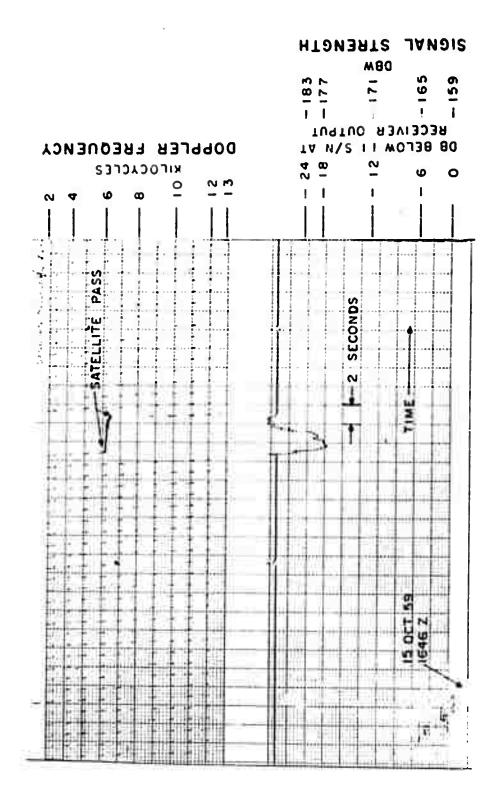
59 EPSILON REV 438	MEASURED DIALIGZ, PREDICTED DIAGZ	CENTER ANTENNA, SOUTH-NORTH PASS
58 DELTA REV. 6844	ED 0140:12 Z,	CENTER ANTENNA, NORTH - SOUTH PASS

FT. SILL

F1g. 12



ARPA-BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV. 7049, FORREST CITY, ARKANSAS
MEASURED 2201:562, PREDICTED 21562
ALTITUDE 375 MILES, 214 MILES EAST FT. SILL
CENTER ANTENNA, NORTH-SOUTH PASS



ARPA-BRL DOPLOC DOPPLER RECORD OF

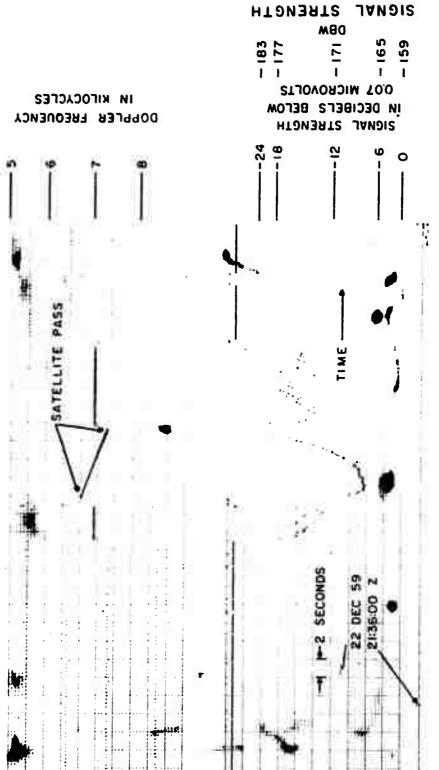
SO DELTA REV. 7358, FORREST'CITY, ARKANSAS

MEASURED 1641 Z

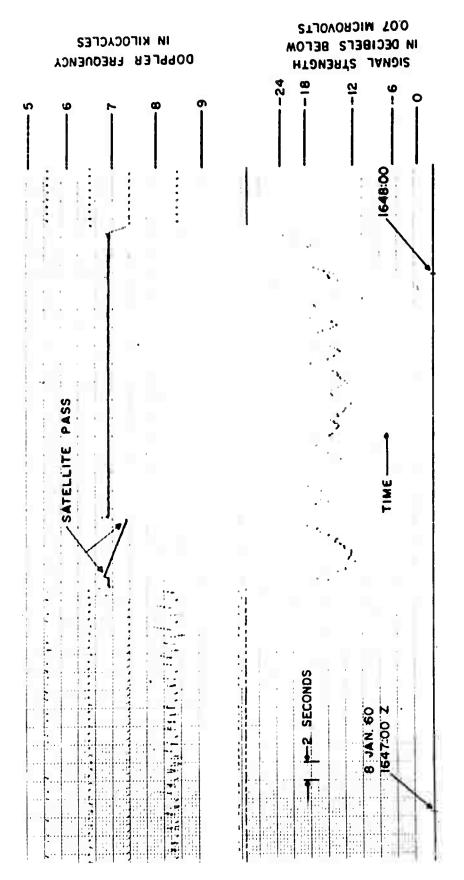
ALTITUDE 369 MILES, 74 MILES WEST FT. SILL

CENTER ANTENNA, NORTH-SOUTH PASS

Fig. 14

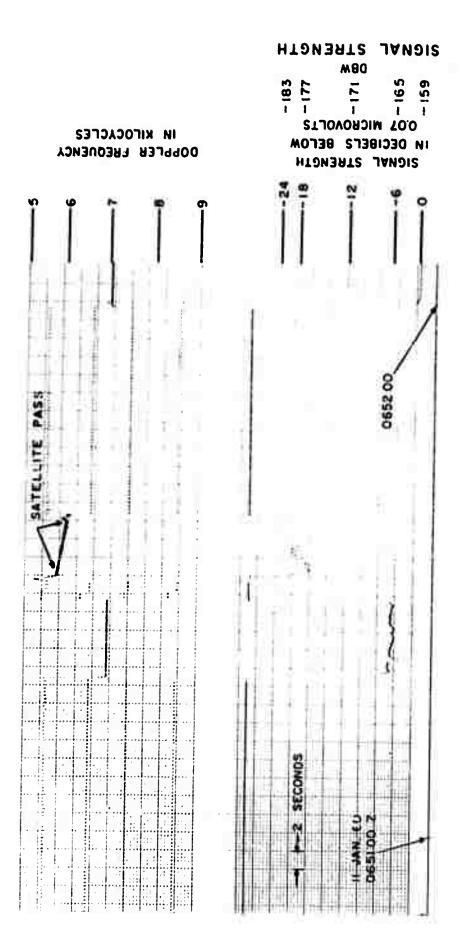


ARPA - BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV 8386 FORREST CITY, ARKANSAS MEASURED 21:36:23 Z, PREDICTED 21:32 Z ALTITUDE 172 MILES, 176 MILES EAST FT. SILL CENTER ANTENNA, N-S PASS

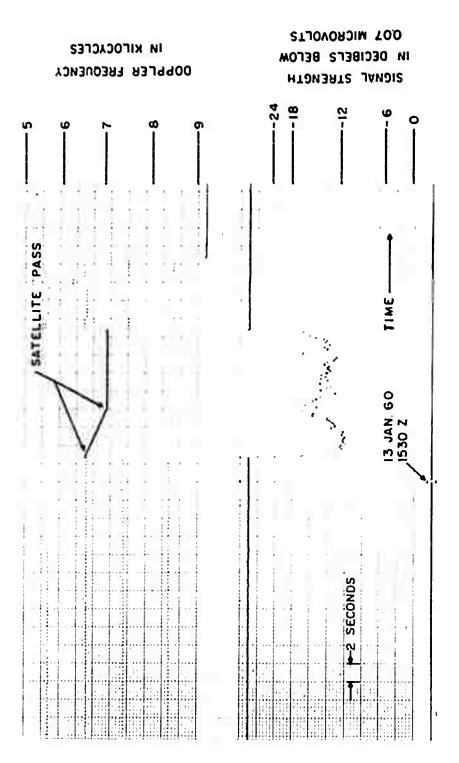


ARPA — BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 8643, FORREST CITY, ARKANSAS MEASURED 1647:25 Z, PREDICTED 1647 Z ALTITUDE 156 MILES, 139 MILES EAST FT. SILL CENTER ANTENNA, NORTH—SOUTH PASS

72

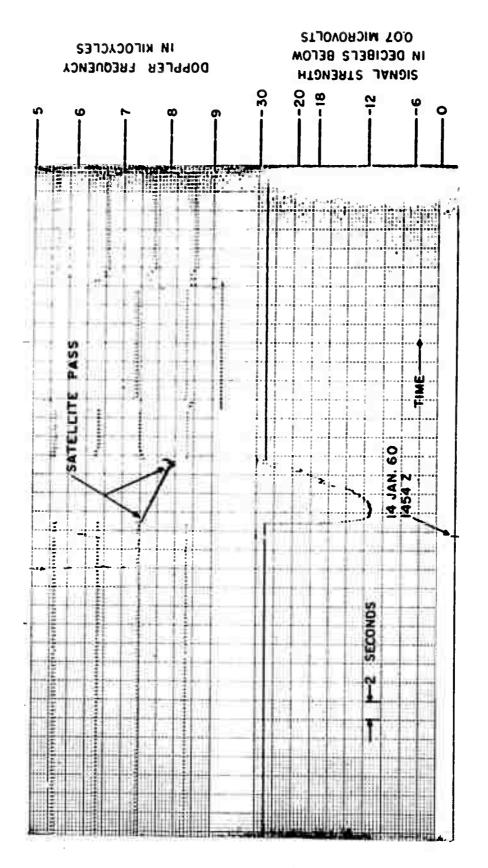


ARPA -- BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 8683, FORREST CITY, ARKANSAS MEASURED 0651:30 Z, PREDICTED 0651 Z ALTITUDE 413 MILES, OVERHEAD FT. SILL CENTER ANTENNA, SOUTH-NORTH PASS

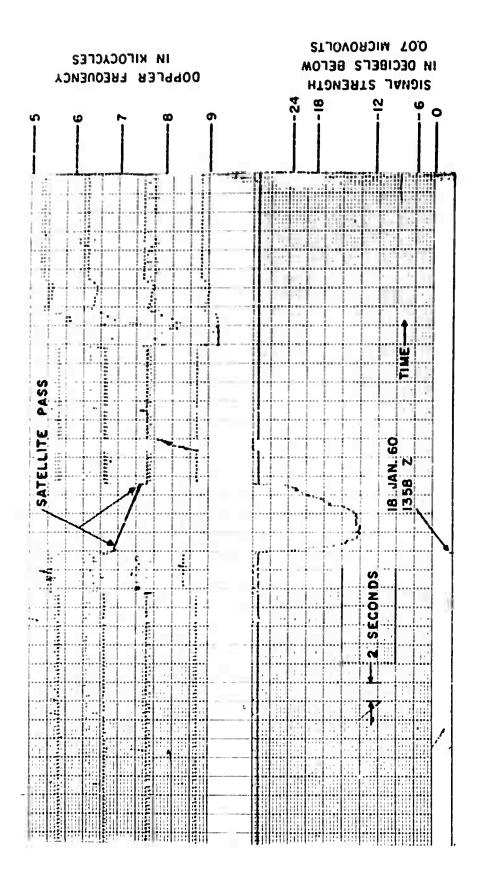


ARPA-BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 8719, FORREST CITY, ARKANSAS MEASURED 1530:03 Z, PREDICTED 1528 Z ALTITUDE 186 MILES, 50 MILES EAST FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS

F18. 18

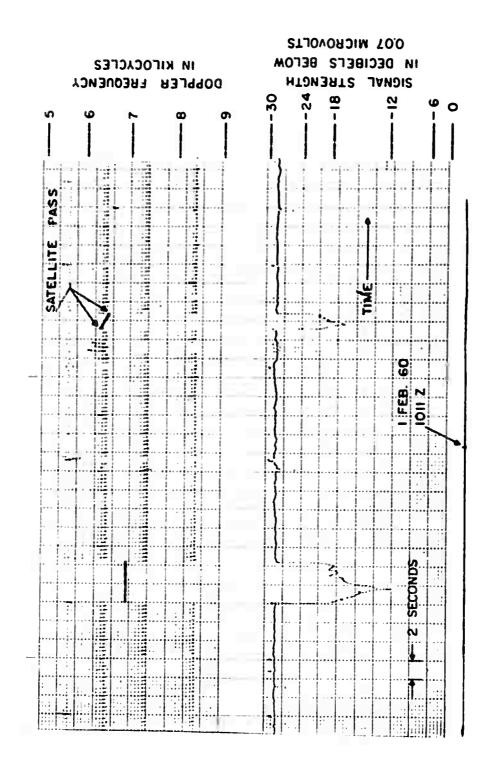


58 DELTA REV. 8734, FORREST CITY, ARKANSAS ARPA -- BRL DOPLOC DOPPLER RECORD OF ALTITUDE 186 MILES, 326 EAST FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS MEASURED 1454:02 Z, PREDICTED 1452 Z



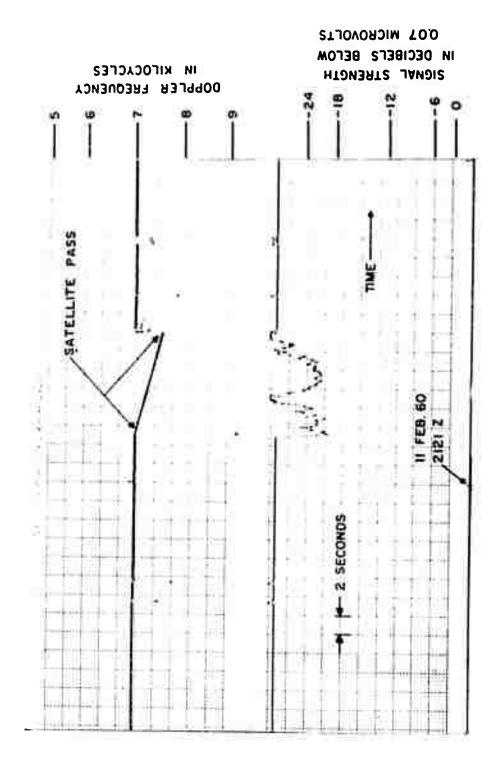
ARPA-BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 8795, FORREST CITY, ARKANSAS MEASURED 1358:01 Z, PREDICTED 1356 Z ALTITUDE 146 MILES, 158 MILES EAST FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS

18.30



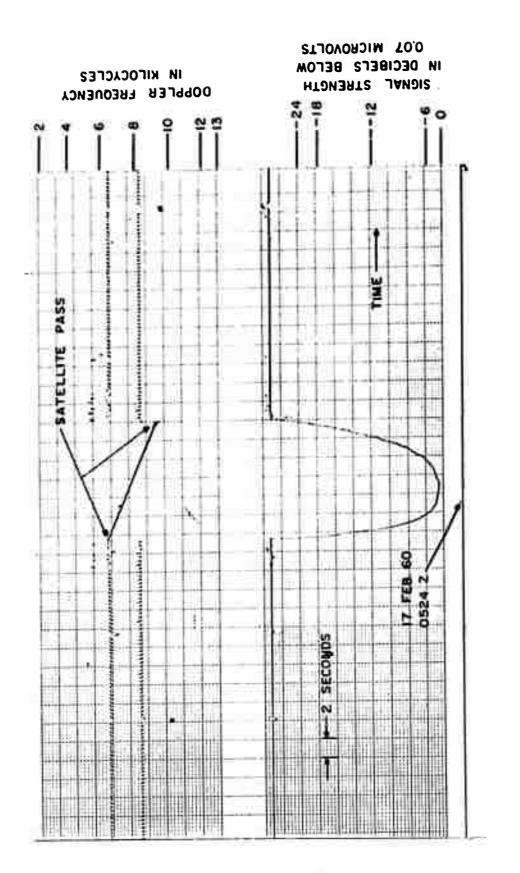
ARPA - BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 9009, FORREST CITY, ARKANSAS MEASURED IOII:12 2, FREDICTED 1009 2 ALTITUDE 134 MILES, 46 MILES WEST FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS

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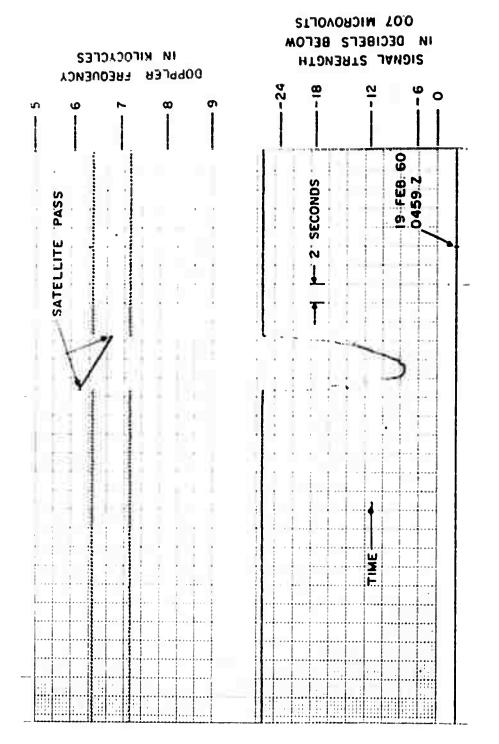
ARPA-BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 9172 FORREST CITY, ARKANSAS MEASURED 2121:06 Z, PREDICTED 2119 Z ALTITUDE 315 MILES, 300 MILES EAST FT. SILL CENTER ANTENNA, SOUTH-NORTH PASS

F16. 22



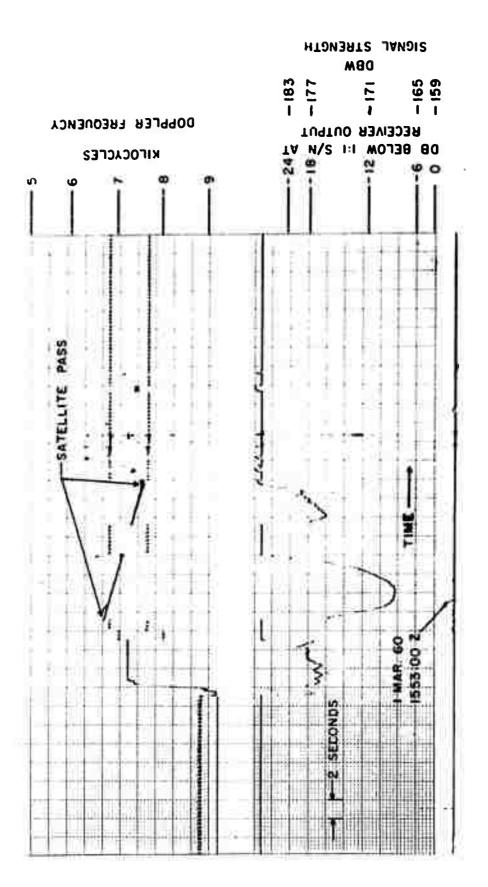
ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV. 9255 FORREST 'CITY, ARKANSAS
MEASURED 0523:57 Z, PREDICTED 0521 Z
ALTITUDE 128 MILES, 120 MILES EAST FT. SILL
CENTER ANTENNA, NORTH-SOUTH PASS

F16. 23

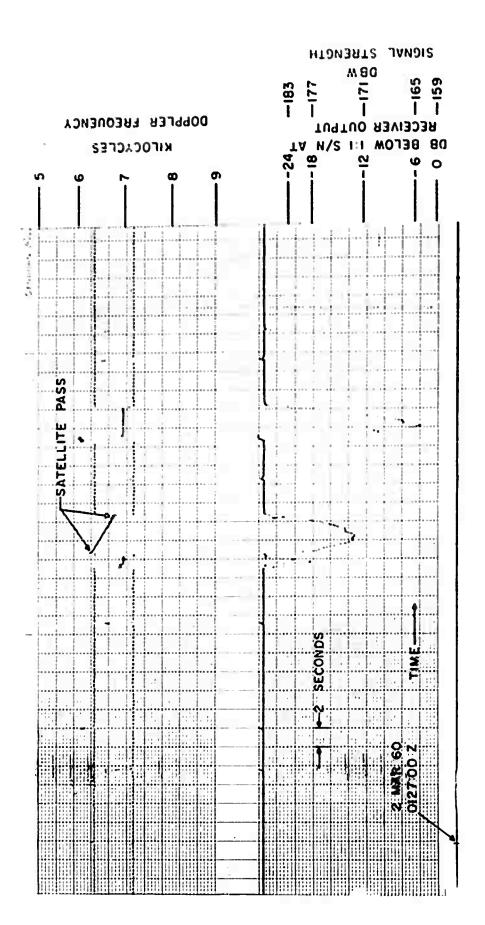


ARPA — BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 9286 FORREST CITY, ARKANSAS MEASURED 0458:45 2, PREDICTED 0457 2 ALTITUDE 123 MILES, 39 MILES WEST FT. SILL CENTER ANTENNA, NORTH—SOUTH PASS

Fig. 24



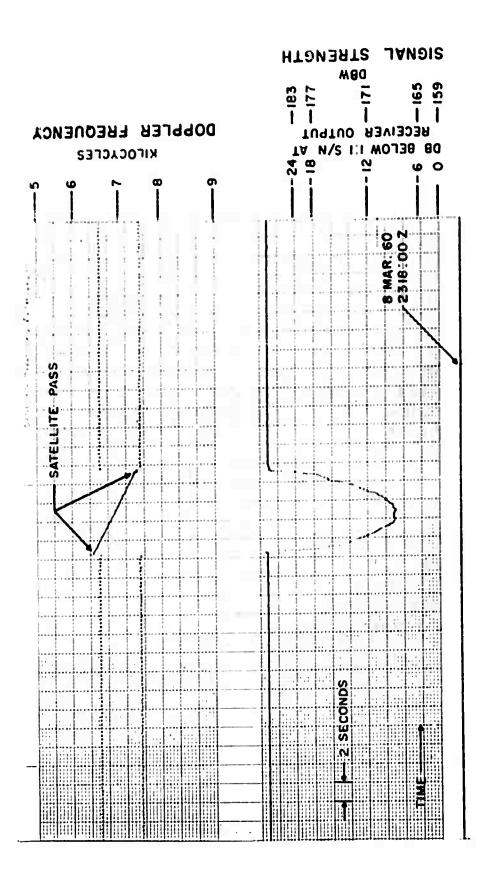
ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV. 9466, FORREST CITY, ARKANSAS
MEASURED 1552:58 Z, PREDICTED 1548 Z
ALTITUSE 275 MILES, 289 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH - NORTH PASS
F16. 25



ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV. 9472, FORREST CITY, ARKANSAS
MEASURED 0127:31 Z, PREDICTED 0128 Z
ALTITUDE 130 MILES, 80 MILES WEST FT. SILL
CENTER ANTENNA, NORTH - SOUTH PASS
F16. 26

ARPA-BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV 9503, FORREST CITY, ARKANSAS MEASURED OO!4:30 2, PREDICTED 0043 2 ALTITUDE 130 MILES, 56 MILES EAST FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS

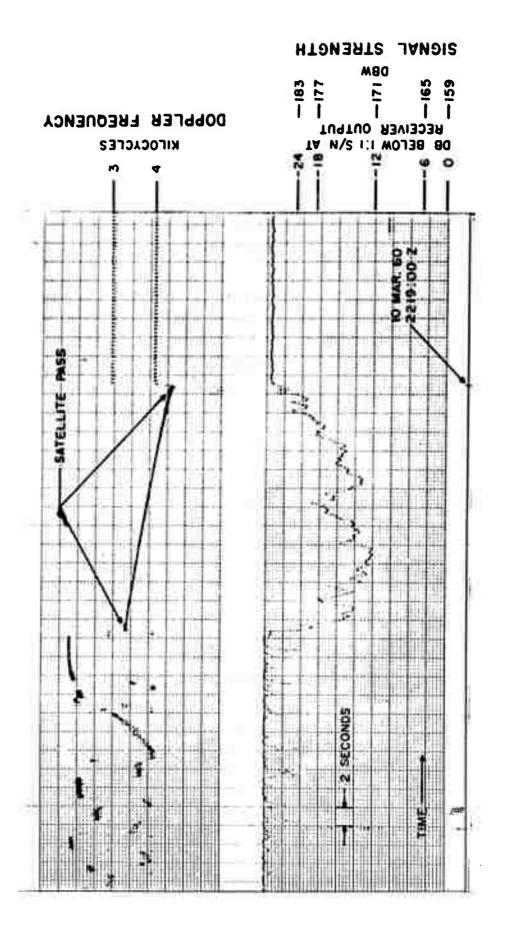
F18. 27



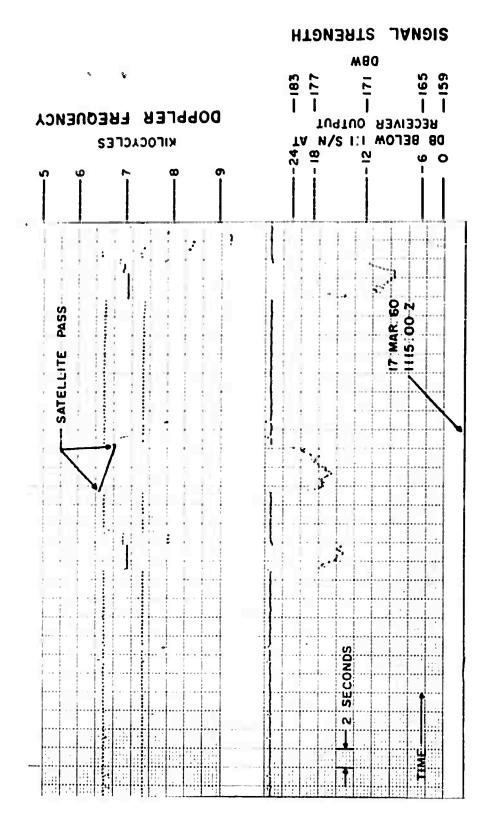
ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV 9581, FORREST CITY, ARKANSAS
MEASURED 2317:41 Z, PREDICTED 2317 Z
ALTITUDE 116 MILES, 33 MILES EAST FT SILL
CENTER ANTENNA, NORTH - SOUTH PASS

F18. 28

F1E . 29



FORREST CITY, ARKANSAS ARPA - BRL DOPLOC DOPPLER RECORD OF SILL PREDICTED 2220 Z MEASURED 2218:33 Z, 58 DELTA REV 9612,

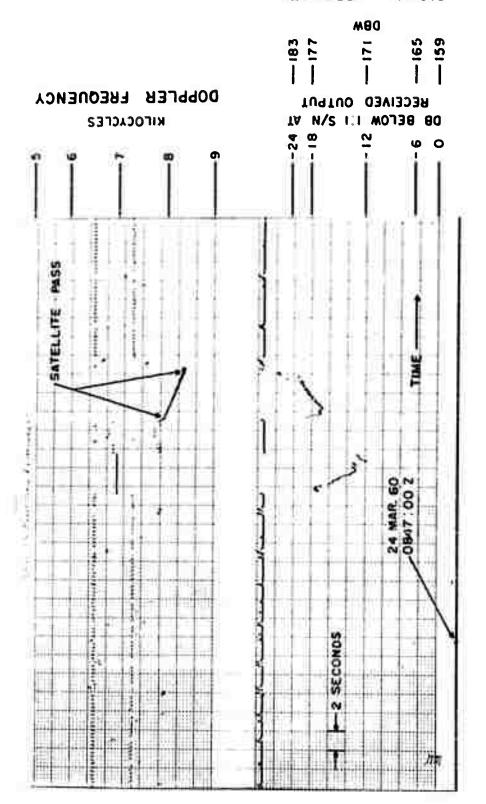


ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV 9716, FORREST CITY, ARKANSAS
MEASURED 1114:54 Z, PREDICTED 1116 Z
ALTITUDE 285 MILES, 135 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS

SIGNAL STRENGTH

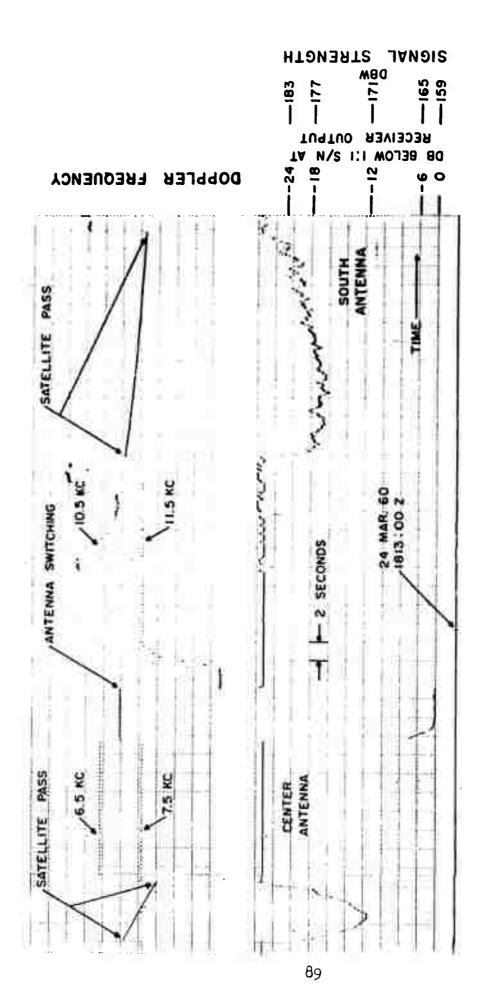
ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV 9722, FORREST CITY, ARKANSAS
MEASURED 2044:312, PREDICTED 20452
ALTITUDE 113 MILES, 122 MILES WEST FT. SILL
SOUTH ANTENNA, NORTH - SOUTH PASS

SIGNAL STRENGTH



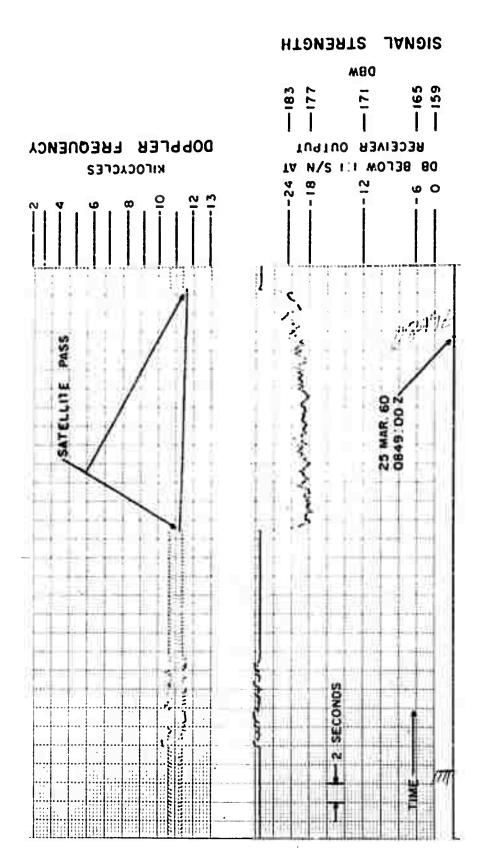
ARPA - BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV 9826, FORREST CITY, ARKANSAS MEASURED 0847:25 Z, PREDICTED 0848 Z ALTITUDE 189 MILES, 403 MILES EAST FT. SILL CENTER ANTENNA, SOUTH - NORTH PASS

35. 35



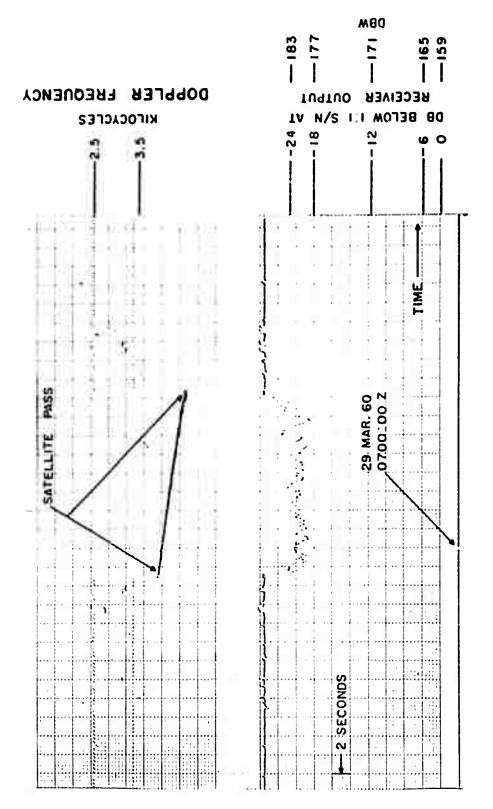
ARPA - BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV 9832, FORREST CITY, ARKANSAS MEASURED 1812:26 Z, PREDICTED 1813 Z ALTITUDE 110 MILES, 244 MILES EAST FT. SILL CENTER AND SOUTH ANTENNA, NORTH - SOUTH PASS

F1g. 33

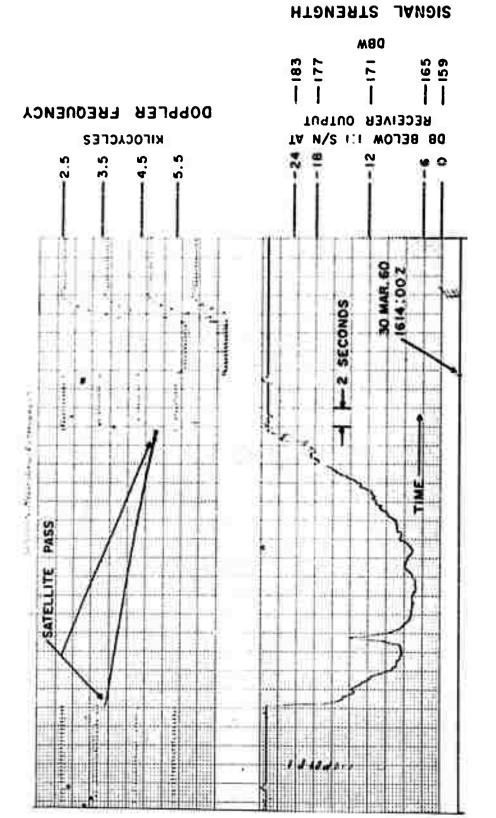


ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV 9642, FORREST CITY, ARKANSAS
MEASURED 0848:402, PREDICTED 08482
ALTITUDE 189 MILES, 137 MILES EAST FT. SILL
NORTH ANTENNA, SOUTH - NORTH PASS

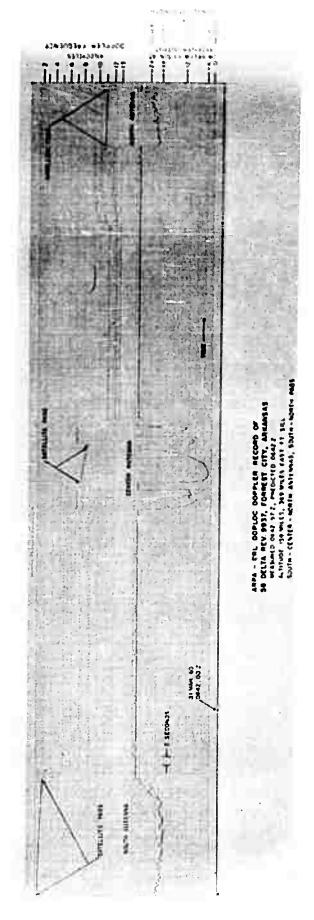
ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV. 9848, FORREST CITY, ARKANSAS
MEASURED 1811:46 Z, PREDICTED 1813 Z
ALTITUDE 109 MILES, OVERHEAD FT. SILL
CENTER AND SOUTH ANTENNA, NORTH-SOUTH PASS



ARPA - BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV 9905, FORREST CITY, ARKANSAS
MEASURED 0659:58 Z, PREDICTED 0701 Z
ALTITUDE 171 MILES, 600 MILES EAST FT. SILL
SOUTH ANTENNA, SOUTH - NORTH PASS

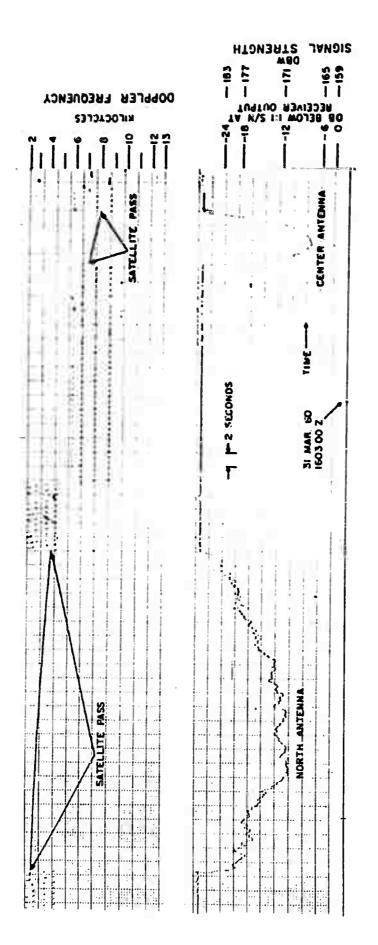


ARPA - BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV 9927, FORREST CITY, ARKANSAS MEASURED 1613:252, PREDICTED 16142 ALTITUDE 105 MILES, 375 MILES EAST FT. SILL NORTH ANTENNA, NORTH - SOUTH PASS

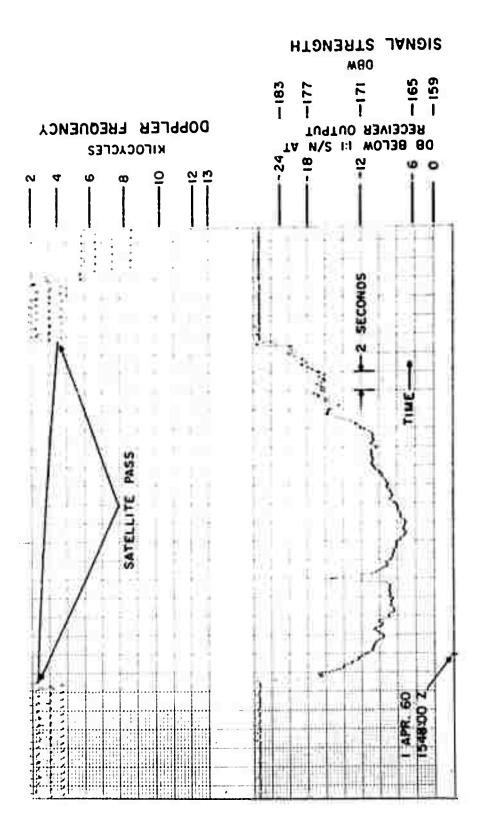


71g. 38



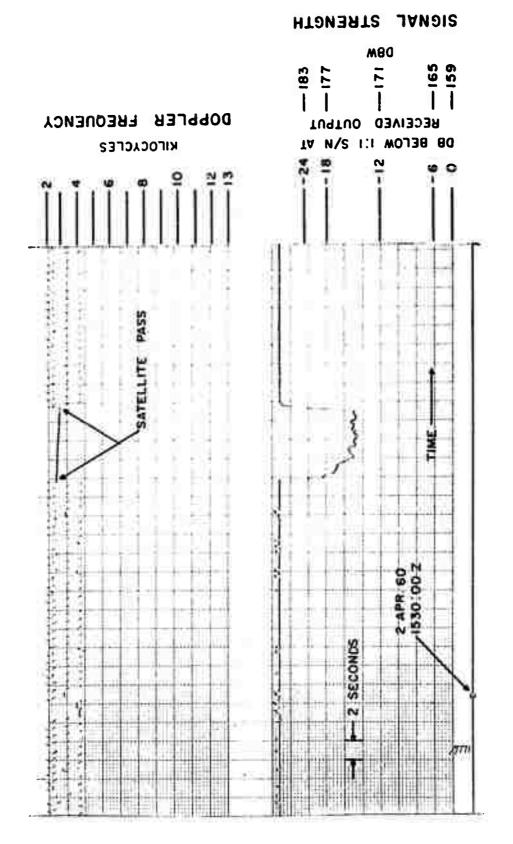


ARPA — BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 9943, FORREST CITY, ARKANSAS MEASURED 1601:53 Z, PREDICTED 1602 Z ALTITUDE 105 MILES, 310 MILES EAST OF FT. SILL NORTH AND CENTER ANTENNAS, NORTH-SOUTH PASS

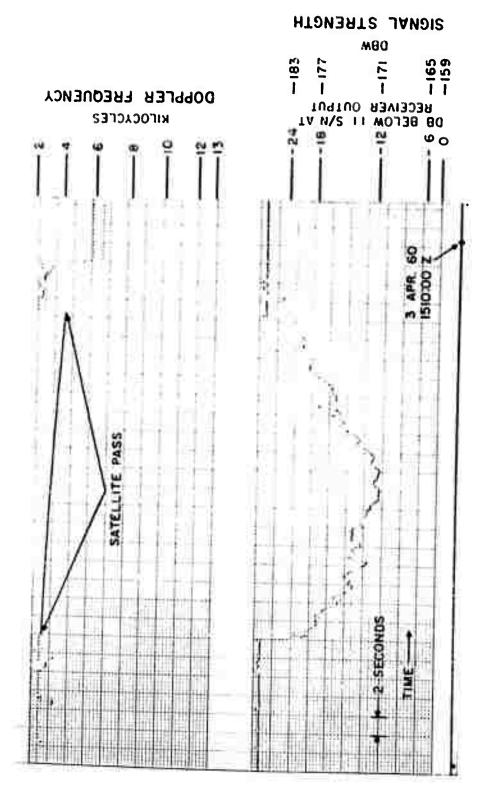


ARPA — BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV. 9959, FORREST CITY, ARKANSAS
MEASURED 1547:58 2, PREDICTED 1547 2
ALTITUDE 103 MILES, 266 MILES EAST OF FT. SILL
NORTH ANTENNA, NORTH - SOUTH PASS

Fig. 40

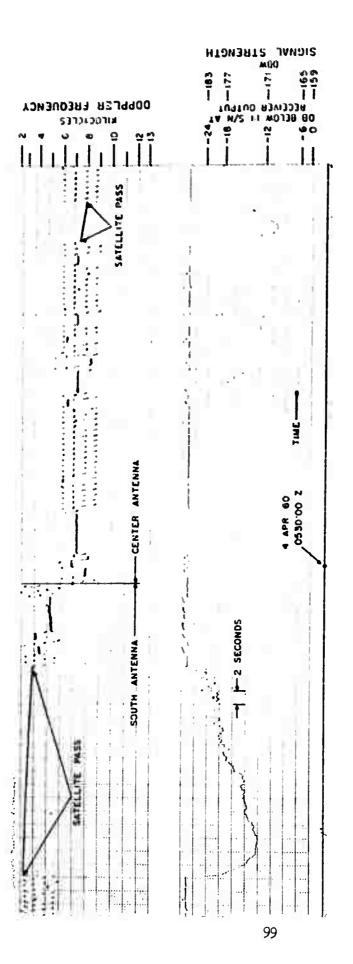


ARPA - BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV 9975, FORREST CITY, ARKANSAS MEASURED 1530:24 Z, PREDICTED 1531 Z ALTITUDE 103 MILES, 211 MILES EAST FT. SILL NORTH ANTENNA, NORTH - SOUTH PASS

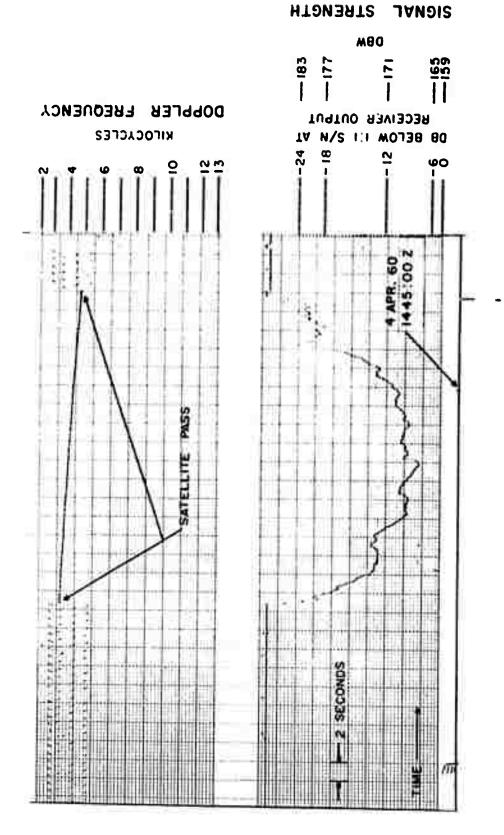


ARPA -- BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 9991, FORREST CITY, ARKANSAS MEASURED 1509:18 Z, PREDICTED 1515 Z ALTITUDE 95 MILES, 246 MILES EAST OF FT. SILL NORTH ANTENNA, NORTH -- SOUTH PASS



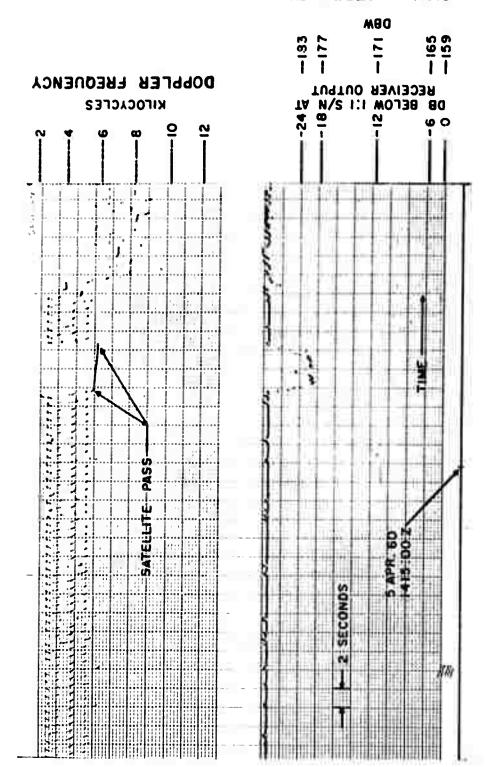


ARPA-BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV. 10001, FORREST CITY, ARKANSAS MEASURED 0530:49 Z. PREDICTED 0537 Z ALTITUDE 124 MILES, 285 MILES EAST OF FT. SILL SOUTH AND CENTER ANTENNAS, SOUTH-NORTH PASS

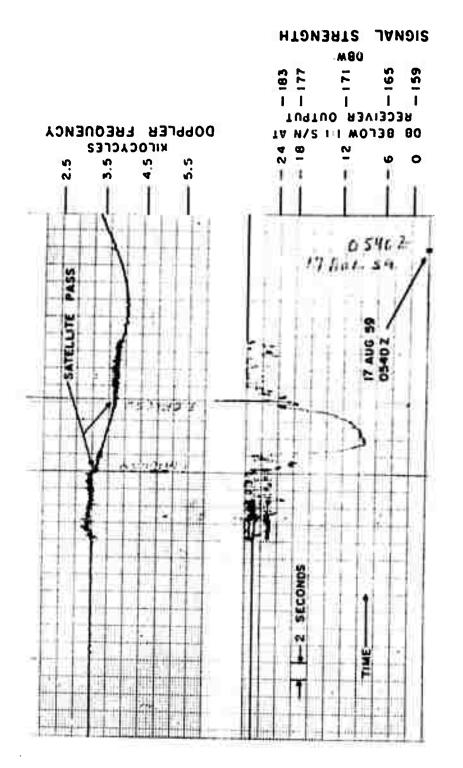


ARPA - BRL DOPLOC DOPPLER RECORD OF 58 DELTA REV 10007, FORREST CITY, ARKANSAS MEASURED 1444:37, PREDICTED 1454 Z ALTITUDE 95 MILES, 380 MILES EAST FT. SILL NORTH ANTENNA, NORTH - SOUTH PASS

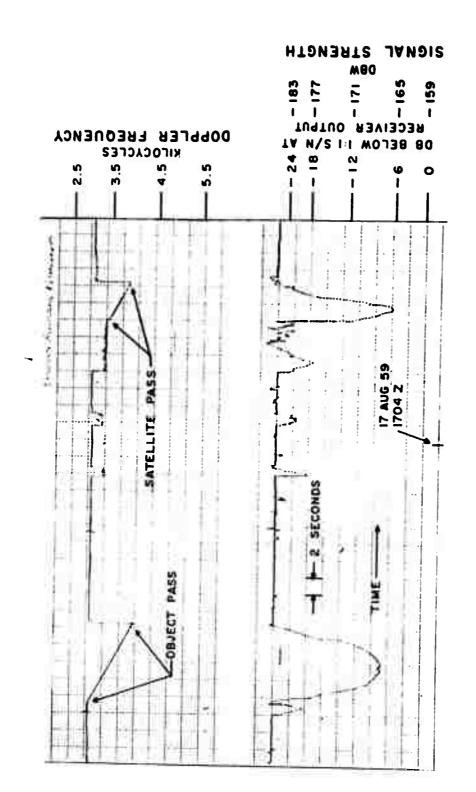
SIGNAL STRENGTH



ARPA-BRL DOPLOC DOPPLER RECORD OF
58 DELTA REV 10023, FORREST CITY, ARKANSAS
MEASURED 1415:082, PREDICTED 14082
ALTITUDE 95 MILES, 600 MILES EAST FT. SILL
NORTH ANTENNA, NORTH - SOUTH PASS



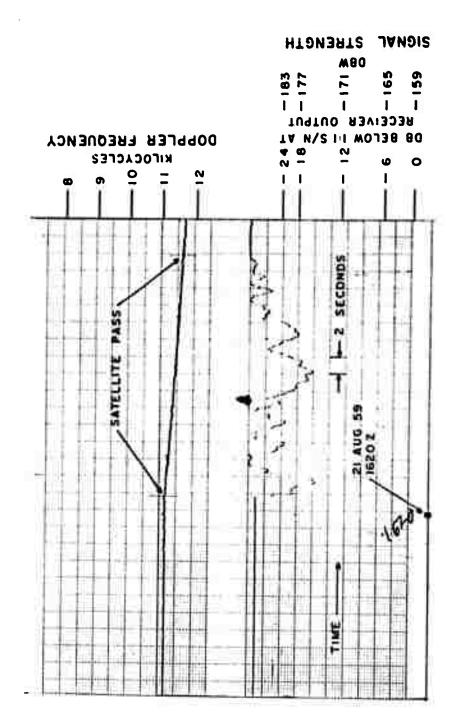
ARPA-BRL DOPLOC DOPPLER RECORD OF
59 EPSILON REV. 53, FORREST CITY, ARKANSAS
MEASURED 0539:34 Z, PREDICTED 0536 Z
ALTITUDE 341 MILES, 279 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS



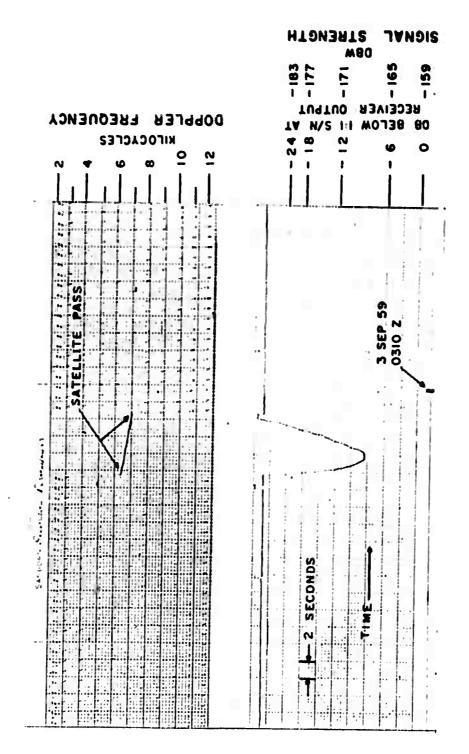
ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT AND 59 EPSILON FORREST CITY, ARKANSAS

UNIDENTIFIED OBJECT MEASURED 1703:302, CENTER ANTENNA ALTITUDE 125 MILES, OVERHEAD FT. SILL

59 EPSILON REV. 60 MEASURED 1704:14 Z, PREDICTED 1701 Z ALTITUDE 140 MILES, 76 MILES WEST FT. SILL CENTER ANTENNA, N-S PASS

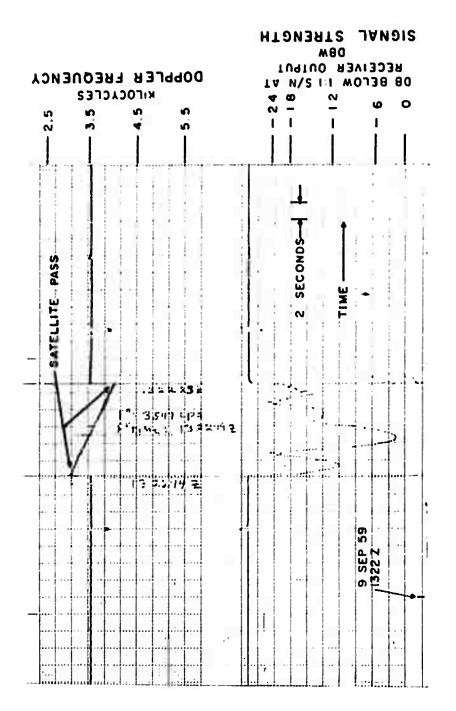


ARPA - BRL DOPLOC DOPPLER RECORD OF 59 EPSILON REV. 121, FORREST CITY, ARKANSAS MEASURED 1620:02 2, PREDICTED 1616 2 ALTITUDE 140 MILES, 100 MILES EAST FT. SILL SOUTH ANTENNA. NORTH - SOUTH PASS

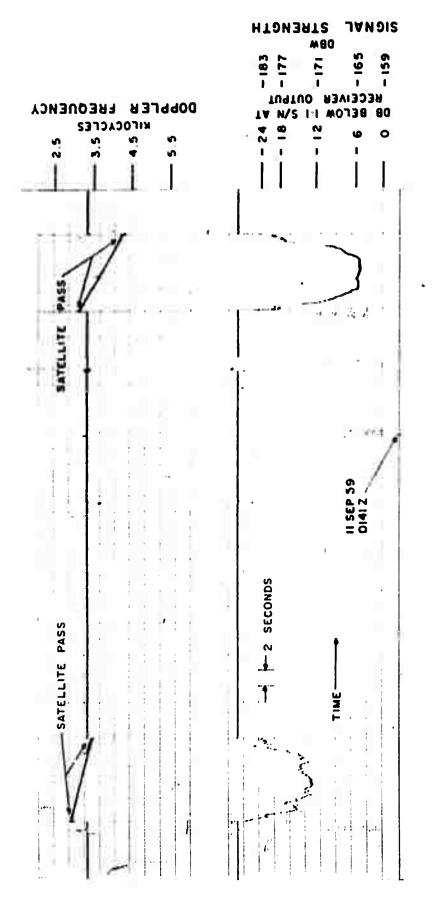


ARPA-BRL DOPLOC DOPPLER RECORD OF 59 EPSILON REV. 314, FORREST CITY, ARKANSAS MEASURED 0309:50 2, PREDICTED 03112 ALTITUDE 185 MILES, 191 MILES EAST FT. SILL CENTER ANTENNA, SOUTH - NORTH PASS





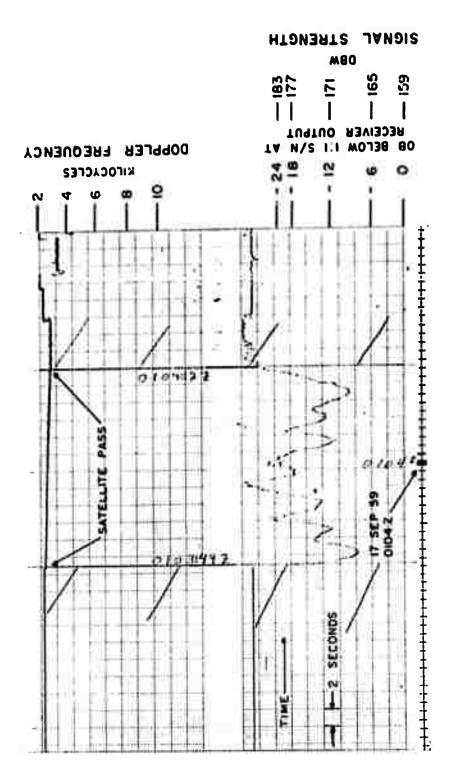
ARPA-BRL DOPLOC DOPPLER RECORD OF 59 EPSILON REV. 414, FORREST CITY, ARKANSAS MEASURED 1322:14 Z, PREDICTED 1324 Z ALTITUDE 211 MILES, 17 MILES WEST FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS



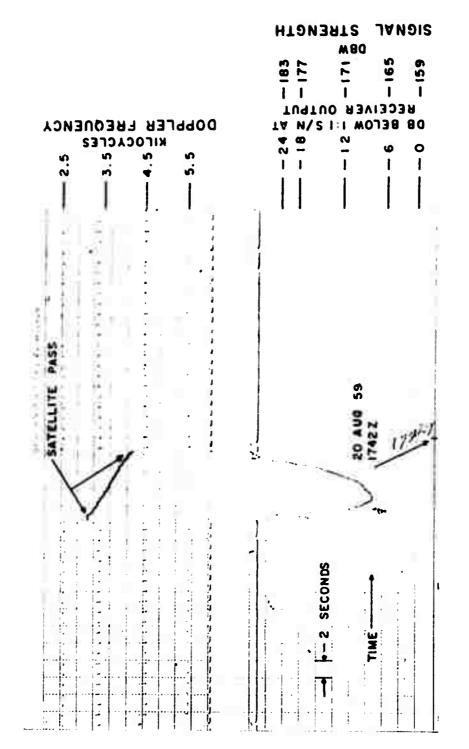
ARPA-BRL DOPLOC DOPPLER RECORD OF 58 DELTA AND 59 EPSILON FORREST CITY, ARKANSAS

58 DELTA REV. 6844
MEASURED 0140:122, PREDICTED 01382
ALTITUDE 407 MILES, 283 MILES EAST FT. SILL
CENTER ANTENNA, NORTH-SOUTH PASS

S9 EPSILON REV. 438
MEASURED 0141:16 Z, PREDICTED 0140 Z
ALTITUDE 150 MILES, 313 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS

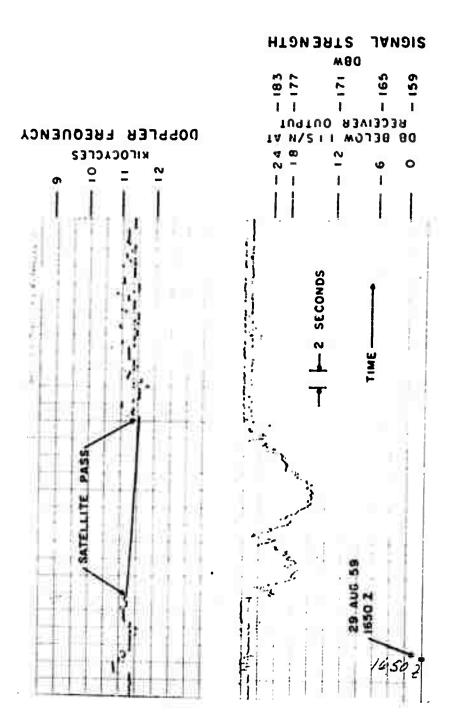


ARPA - BRL DOPLOC DOPPLER RECORD OF
59 EPSILON REV. 532, FORREST CITY, ARKANSAS
MEASURED 0103:492, PREDICTED 01062
ALTITUDE 122 MILES, 32 MILES WEST FT. SILL
SOUTH ANTENNA, SOUTH - NORTH PASS

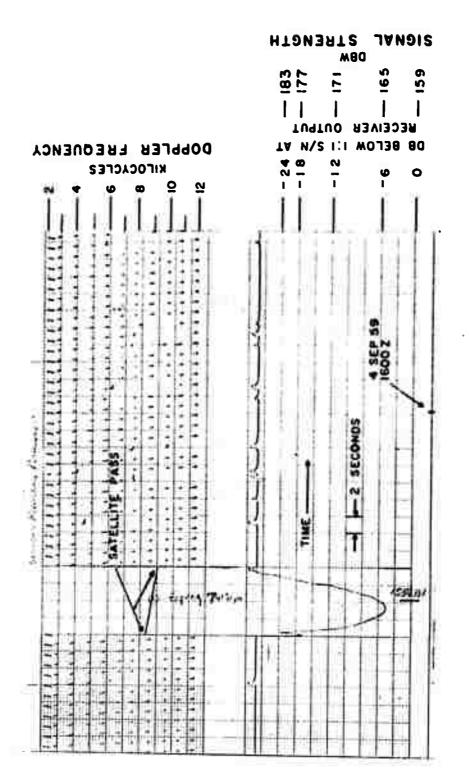


ARPA-BRL DOPLOC DOPPLER RECORD OF 59 ZETA REV. 14, FORREST CITY, ARKANSAS MEASURED 1741:512, PREDICTED 1745 Z ALTITUDE 136 MILES, 145 MILES EAST FT.SILL CENTER ANTENNA, NORTH-SOUTH PASS

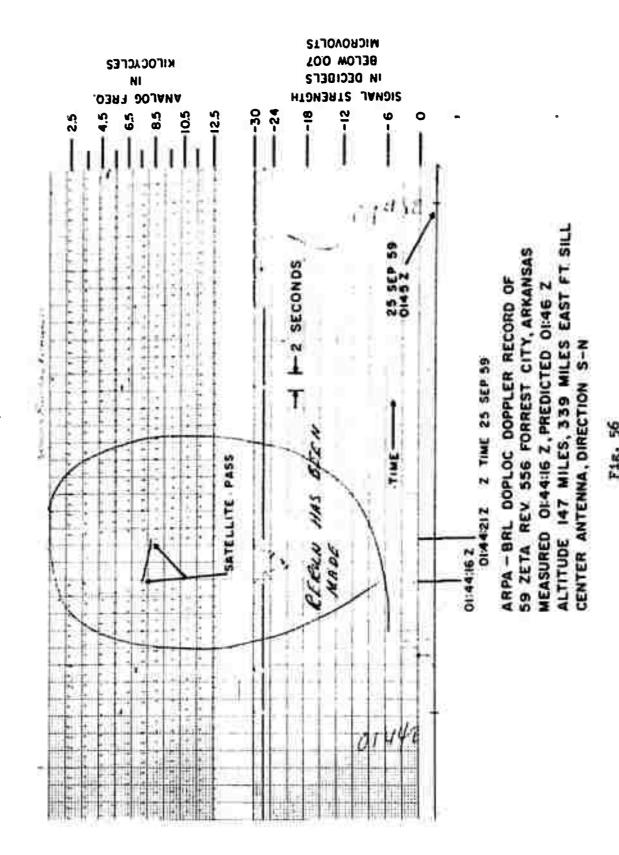


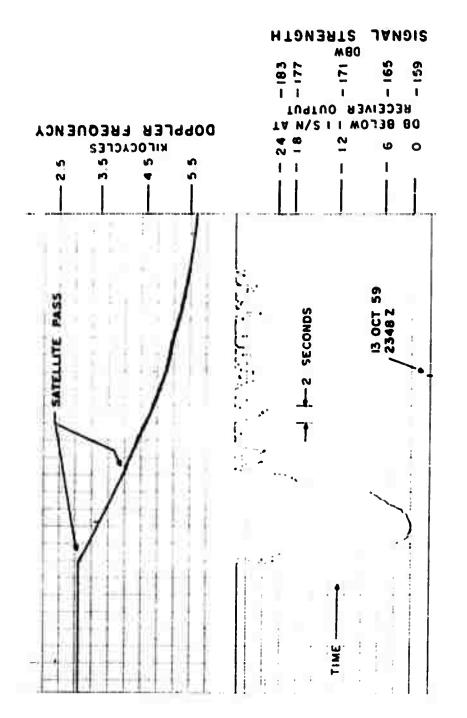


ARPA - BRL DOPLOC DOPPLER RECORD OF 59 ZETA REV. 150, FORREST CITY, ARKANSAS MEASURED 1650:07 Z, PREDICTED 1648 Z ALTITUDE 171 MILES, 15 MILES EAST FT. SILL SOUTH ANTENNA, NORTH-SOUTH PASS

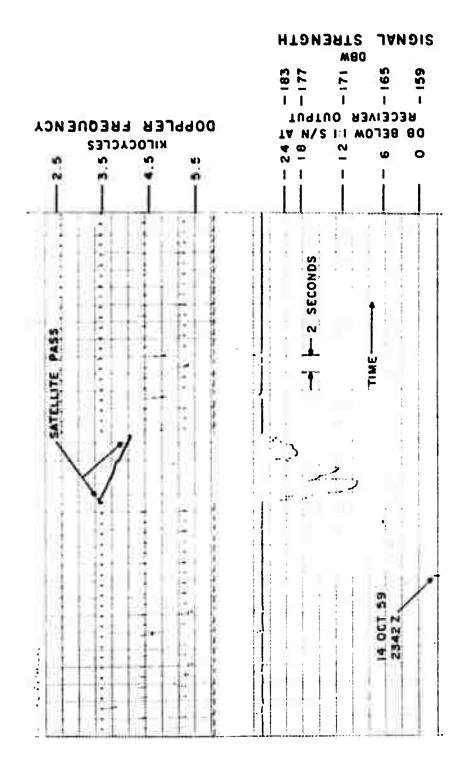


ARPA - BRL DOPLOC DOPPLER RECORD OF 59 ZETA REV. 241, FORREST CITY, ARKANSAS MEASURED 1559: 342, PREDICTED 16012 ALTITUDE 213 MILES, 123 MILES EAST FT. SILL CENTER ANTENNA, NORTH - SOUTH PASS

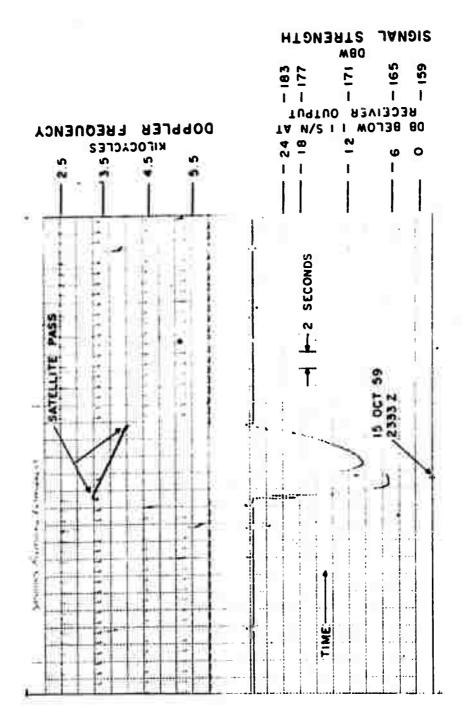




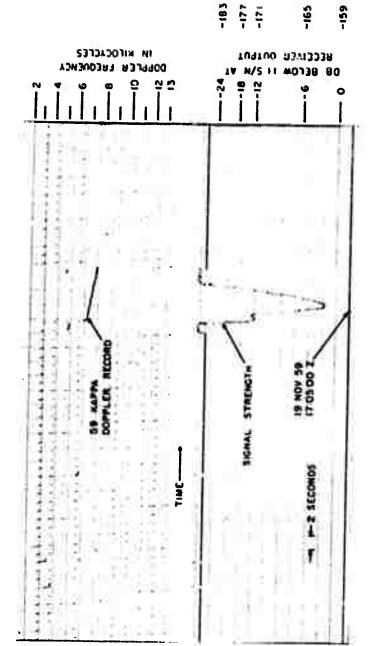
ARPA-BRL DOPLOC DOPPLER RECORD OF
59 ZETA REV. 855, FORREST CITY, ARKANSAS
MEASURED 2347:38 Z, PREDICTED 2346 Z
ALTITUDE 138 MILES, 179 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS



ARPA-BRL DOPLOC DOPPLER RECORD OF
59 ZETA REV. 871, FORREST CITY, ARKANSAS
MEASURED 2342:09 Z, PREDICTED 2345 Z
ALTITUDE 175 MILES, 108 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS



ARPA-BRL DOPLOC DOPPLER RECORD OF
59 ZETA REV. 887, FORREST CITY, ARKANSAS
MEASURED 2332:58 Z, PREDICTED 2338 Z
ALTITUDE 170 MILES, 123 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS



PERIOD

0 0 0

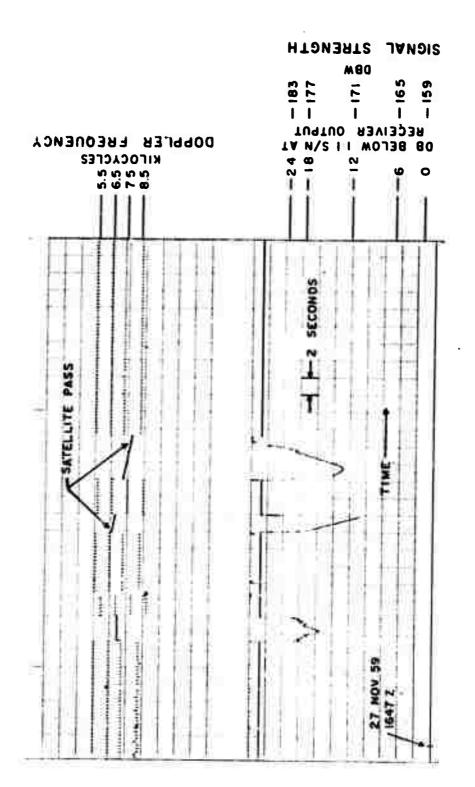
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ARPA -- BRL DOPLOC DOPPLER RECORD OF 59 KAPPA REV 183 FORREST CITY, ARKANSAS MEASURED 17.04:58 Z, PREDICTED 17.06 Z ALTITUDE 115 MILES, 314 MILES EAST FT SILL CENTER ANTENNA, NORTH -- SOUTH PASS

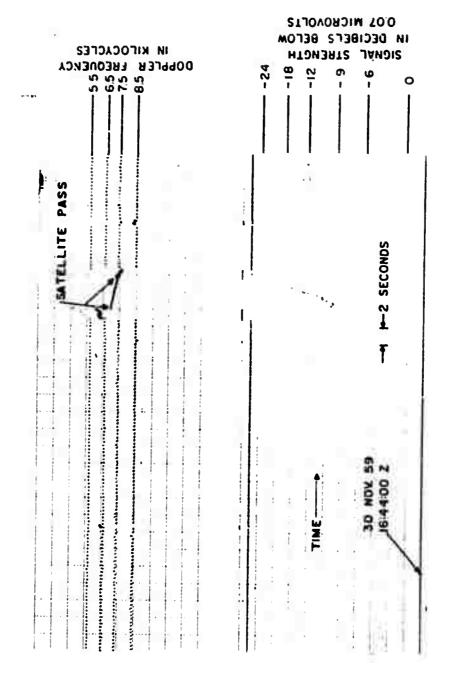
F18. 6

DIGITAL DOPPLER PERIOD PRINT RECORD

O 01



ARPA - BRL DOPLOC DOPPLER RECORD OF 59 LAMBDA REV. 96, FORREST CITY, ARKANSAS MEASURED 1647: 24 Z, PREDICTED 1639 Z ALTITUDE 124 MILES, 323 MILES EAST FT. SILL CENTER ANTENNA, NORTH - SOUTH PASS

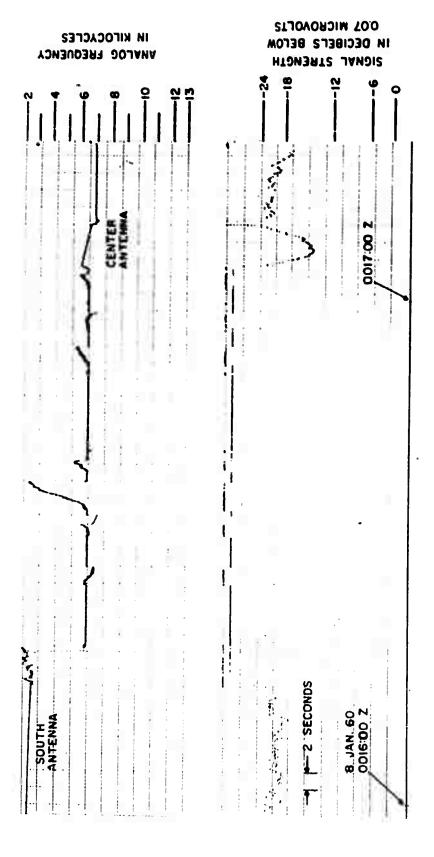


ARPA-BRL DOPLOC DOPPLER RECORD OF 59 LAMBDA REV 138 FORREST CITY, ARKANSAS MEASURED 16:44:31 Z, PREDICTED 16:40 Z ALTITUDE 138 MILES, OVERHEAD FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS

F18. 62

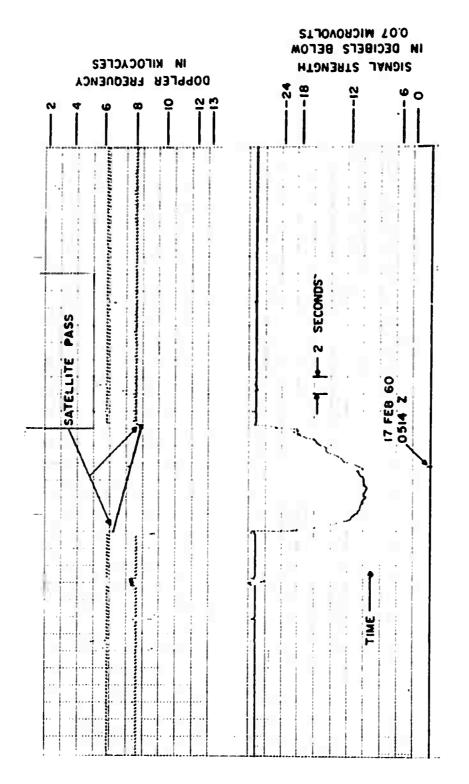
119

ARPA - BRL DOPLOC DOPPLER RECORD OF 59 LAMBDA REV 27B FORREST CITY, ARKANSAS MEASURED 15:03:36 Z, PREDICTED 15:17 Z ALTITUDE 218 MILES, 59 MILES EAST FORT SILL NORTH AND CENTER ANTENNAS, N-S PASS

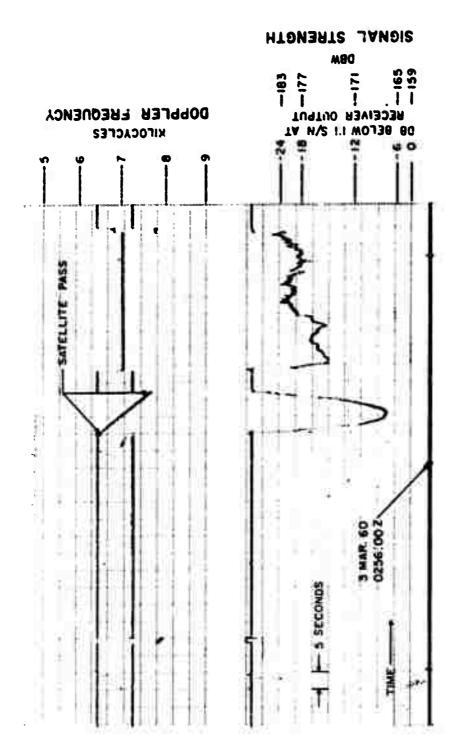


ARPA—BRL DOPLOC DOPPLER RECORD OF 59 LAMBDA REV. 685, FORREST CITY, ARKANSAS MEASUREL OOI6:01 Z, PREDICTED OO20 Z ALTITUDE 138 MILES, 26 MILES WEST FT. SILL SOUTH AND CENTER ANTENNAS SOUTH — NORTH PASS

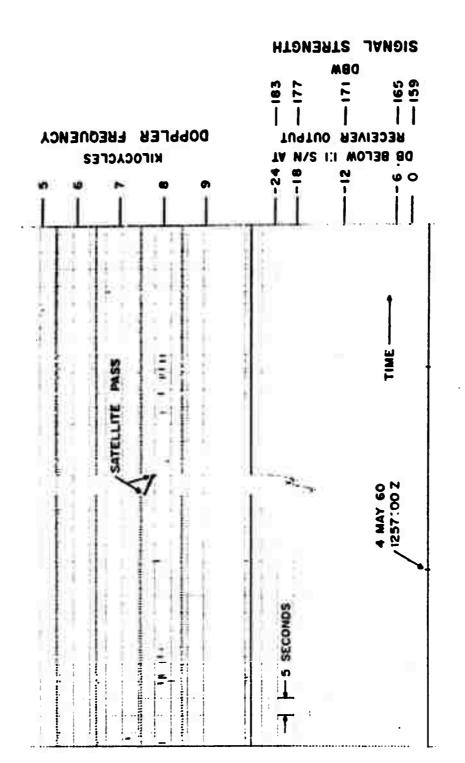
Fig. 64



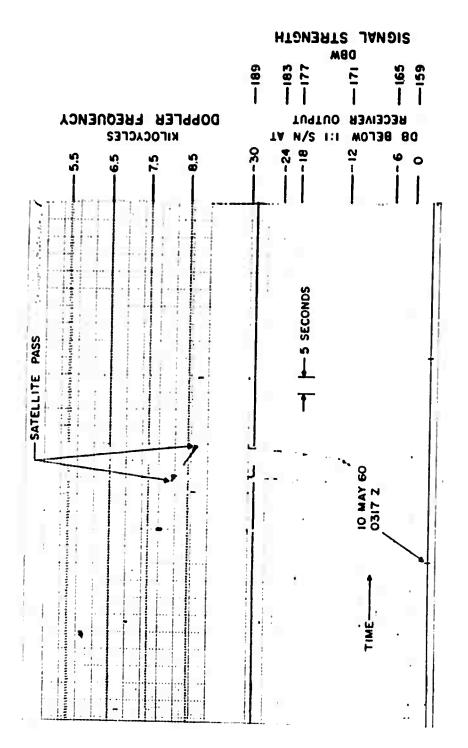
ARPA-BRL DOPLOC DOPPLER RECORD OF 59 LAMBDA REV. 1285 FORREST CITY, ARKANSAS MEASURED 0513:53 Z, PREDICTED 0511 Z ALTITUDE 356 MILES, 342 MILES EAST FT. SILL CENTER ANTENNA, NORTH-SOUTH PASS



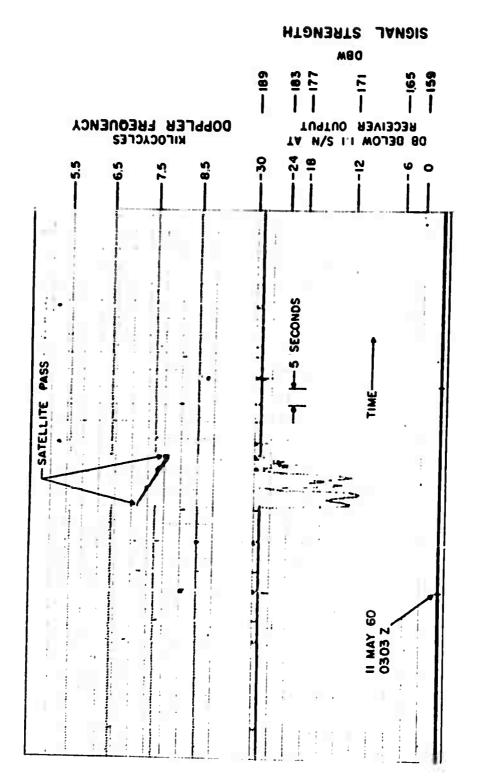
ARPA-BRL DOPLOC DOPPLER RECORD OF
59 LAMBDA REV 1516, FORREST CITY, ARKANSAS
MEASURED 0256:102, PREDICTED 02532
ALTITUDE 137 MILES, 256 MILES EAST FT. SILL
CENTER ANTENNA, NORTH - SOUTH PASS



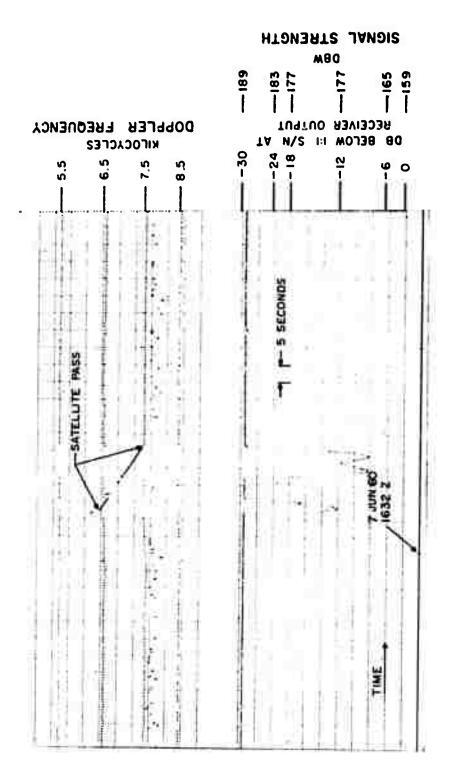
ARPA-BRL DOPLOC DOPPLER RECORD OF
60 GAMMA I REV. 318, FORREST CITY, ARKANSAS
MEASURED 1257:23 Z, PREDICTED 1256 Z
ALTITUDE 440 MILES, 187 MILES EAST FT. SILL
CENTER ANTENNA, NORTH-SOUTH PASS



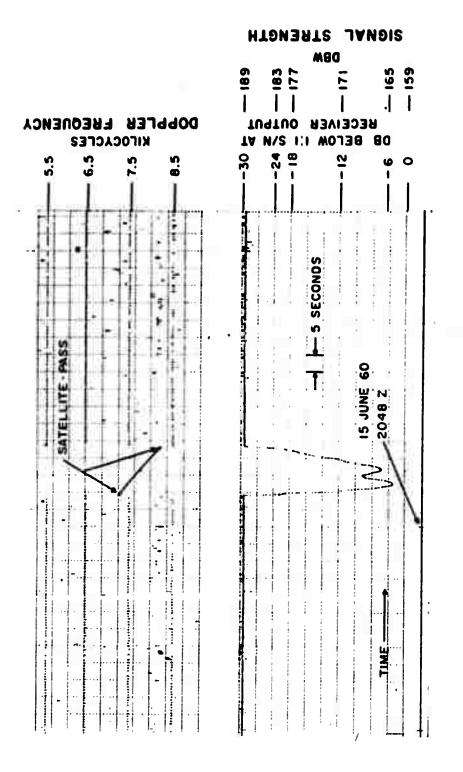
ARPA - BRL DOPLOC DOPPLER RECORD OF
60 GAMMA I REV. 403, FORREST CITY, ARKANSAS
MEASURED 0317:28 Z, PREDICTED 0318 Z
ALTITUDE 271 MILES, 400 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS



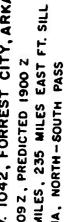
ARPA - BRL DOPLOC DOPPLER RECORD OF 60 GAMMA I REV. 418, FORREST CITY, ARKANSAS MEASURED 0303:26 Z, PREDICTED 0304 Z ALTITUDE 287 MILES, 212 EAST FT. SILL CENTER ANTENNA, SOUTH-NORTH PASS



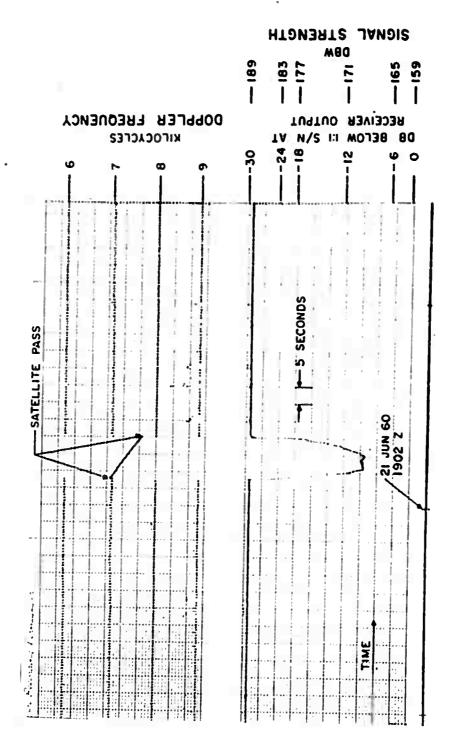
ARPA-BRL DOPLOC DOPPLER RECORD OF
60 GAMMA I REV. 836, FORREST CITY, ARKANSAS
MEASURED 1632:12 Z, PREDICTED 1633 Z
ALTITUDE 312 MILES, 232 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS



ARPA - BRL DOPLOC DOPPLER RECORD OF
60 GAMMA I REV. 960, FORREST CITY, ARKANSAS
MEASURED 2048:10 Z, PREDICTED 2049 Z
ALTITUDE 205 MILES, 154 MILES EAST FT. SILL
CËNTER ANTENNA, NORTH - SOUTH PASS

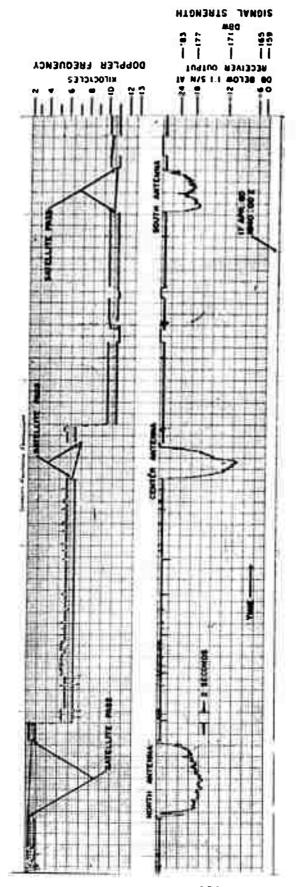


F18. 72

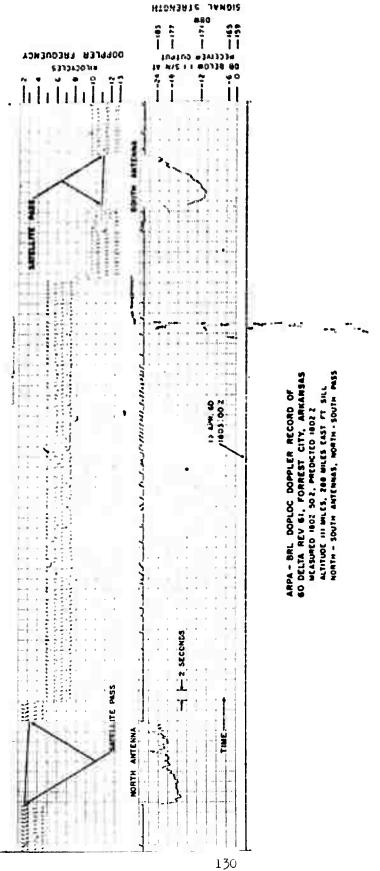


60 GAMMA 2 REV. 1042, FORREST CITY, ARKANSAS ALTITUDE 250 MILES, 235 MILES EAST FT. SILL ARPA-BRL DOPLOC DOPPLER RECORD OF MEASURED 1902:09 Z, PREDICTED 1900 Z CENTER ANTENNA, NORTH-SOUTH PASS

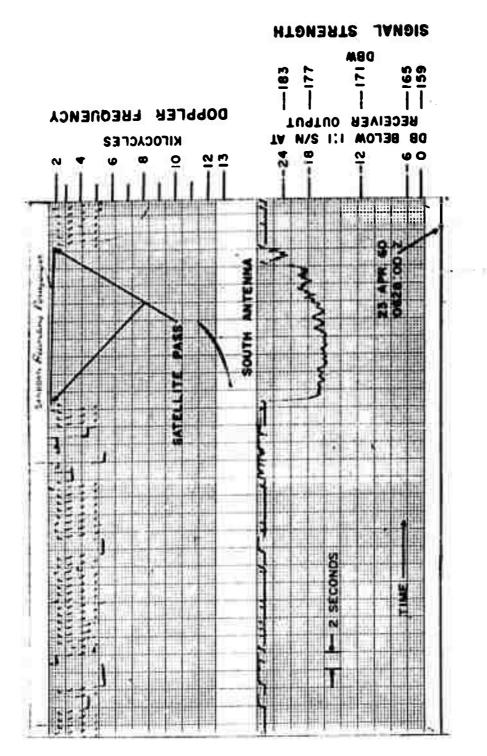




ARPA - BRL DOPLOC DOPPLER RECORD OF 60 DELTA REV. 30, FORREST CITY, ARKANSAS MEASURED 1839:192, PREDICTED 1839:192 ALTITUDE 111 MILES, 72 MILES EAST FT SILL MORTH - CENTER - SOUTH ANTERHAS, MORTH - SOUTH PASS



F18. 74



ARPA - BRL DOPLOC DOPPLER RECORD OF
60 DELTA REV. 117, FORREST CITY, ARKANSAS
MEASURED 3627:41 Z, PREDICTED 0629 Z
ALTITUDE 204 MILES, 255 MILES EAST FT. SILL
SOUTH ANTENNA, SOUTH - NORTH PASS



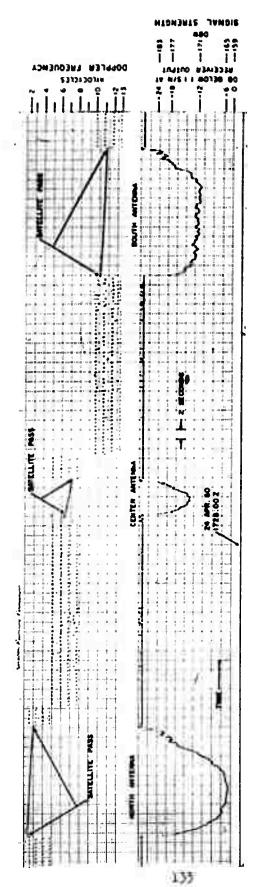


OOPPLEM PREQUENCY T.

HIGHTHE JANGIE

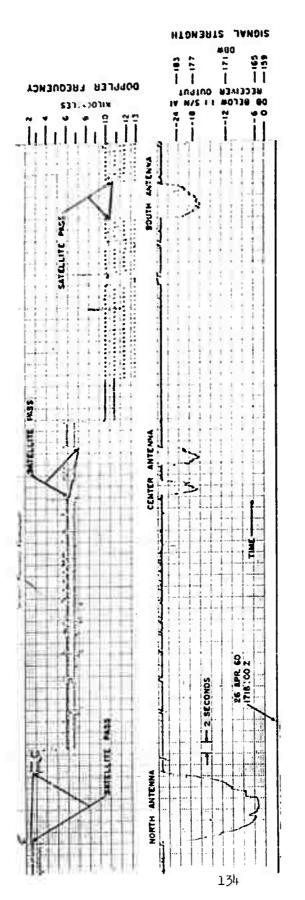
F16. 76



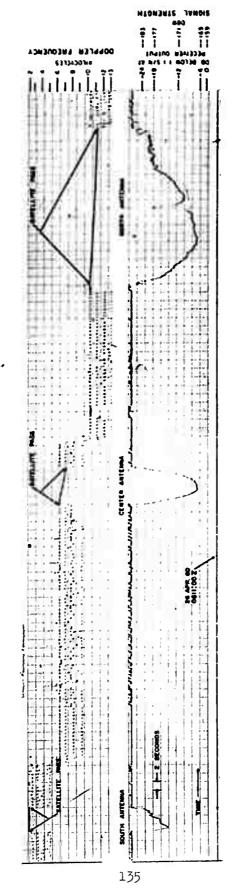


ARPA - BRL DOPLOC DOPPLER RECORD OF 60 DELTA REV 140, FORREST CITY, AMKANSAS WEASURED 1728:04 2, PREDICTED 1732 2 ALTITUDE 116 MICS, 320 WILES EAST FT. SILL MONTH - CENTER - SOUTH ANTENNAS, MONTH - SOUTH PASS

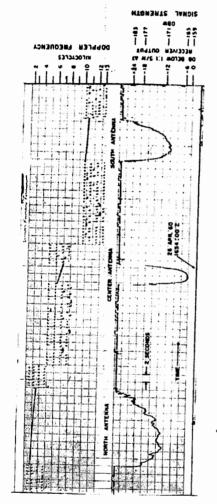




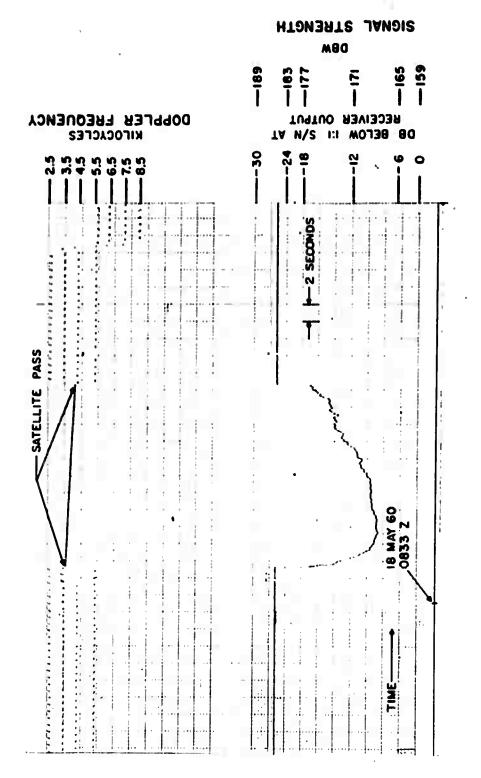
ARPA - BRL DOPLOC DOPPLER RECORD OF 60 DELTA REV. 156, FORREST CITY, ARKANSAS MEASURED 1718: 40 2, PREDICTED 1725 2 ALTITUDE 112 MILES, 143 MILES EAST FT. SILL NORTH - CENTER - SOUTH ANTEWNS, NONTH - SOUTH PASS



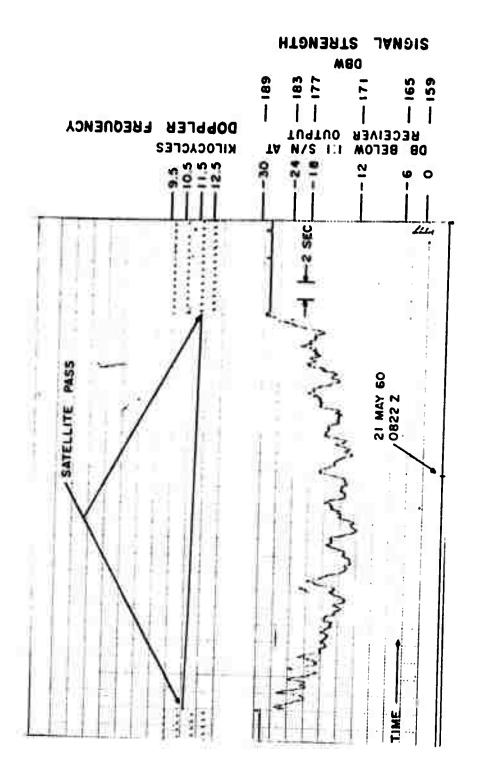
AAPA - BRL DOPLOC DOPPLEM RECORD OF 60 DELTA REV. MB., FORREST CITY, AMEASAS MEASANGO DOH! 12, PRODECTED DOHS ALTITOCH 142 MALE, FORMERS (ASY PT SILL SOUTH - CENTER - WORTH ANTERNAS, SOUTH - WORTH PASS



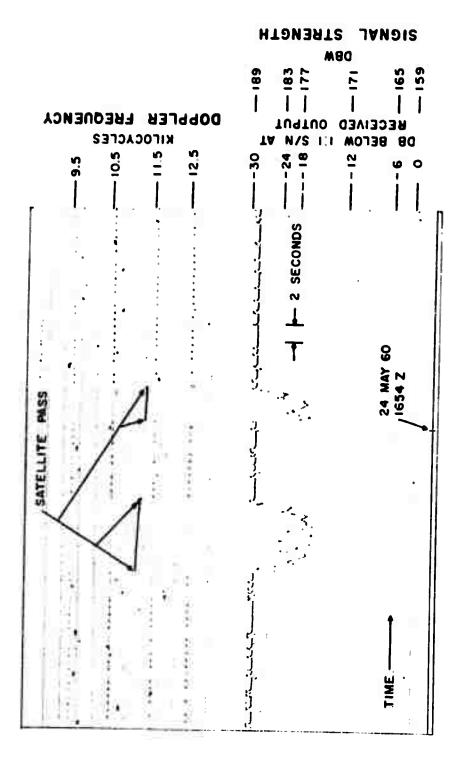
ARPA - BRL DOPLOC DOPPLER RECORD OF 60 DELTA REV. 172, FOREST CITY, ARKANSAS MENSANEO 1933: 34 2, PREDICTED 1702. ALTITUDE 77 HILES, 370 MILES EAST FT, SILL NORTH - CENTER - SOUTH ANTERNAS, NORTH - SOUTH PASS



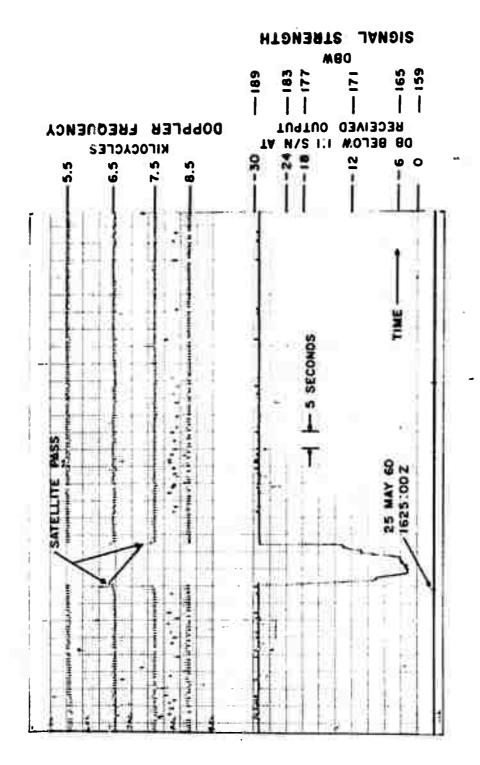
ARPA-BRL DOPLOC DOPPLER RECORD OF
60 EPSILON I REV. 53, FORREST CITY, ARKANSAS
MEASURED 0833:05 Z, PREDICTED 0833 Z
ALTITUDE 194 MILES, 252 MILES EAST FT SILL
SOUTH ANTENNA, SOUTH-NORTH PASS



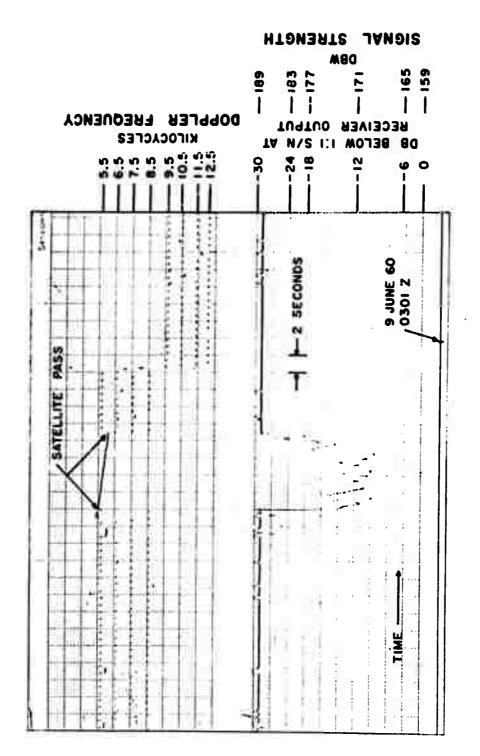
ARPA - BRL DOPLOC DOPPLER RECORD OF
60 EPSILON I REV. 99, FORREST CITY, ARKANSAS
MEASURED 0821:32 Z, PREDICTED 0820 Z
ALTITUDE 200 MILES, 305 MILES EAST FT. SILL
NORTH ANTENNA, SOUTH - NORTH PASS



ARPA - BRL DOPLOC DOPPLER RECORD OF
60 EPSILON I REV. 150, FORREST CITY, ARKANSAS
MEASURED 1653:44 Z, PREDICTED 1654 Z
ALTITUDE 287 MILES, 291 MILES EAST FT. SILL
SOUTH ANTENNA, NORTH - SOUTH PASS



ARPA-BRL DOPLOC DOPPLER RECORD OF 60 EPSILON I REV. 165, FORREST CITY, ARKANSAS MEASURED 1624:592, PREDICTED 16282 ALTITUDE 288 MILES, 96 MILES EAST FT. SILL CENTER ANTENNA, NORTH - SOUTH PASS



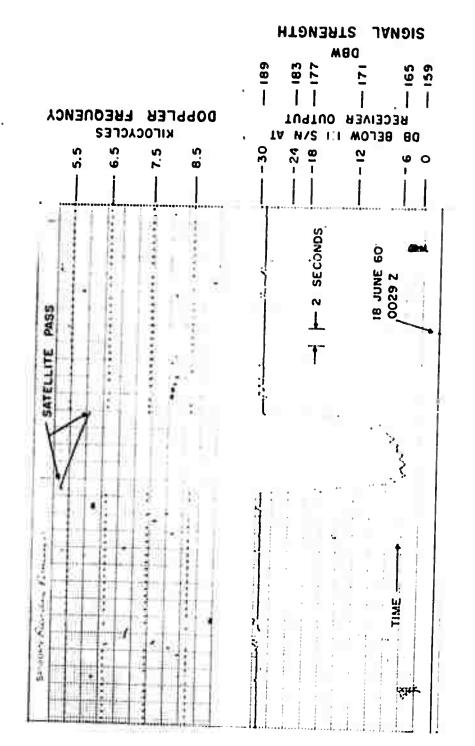
ARPA - BRL DOPPLER RECORD OF

60 EPSILON I REV. 386, FORREST CITY, ARKANSAS

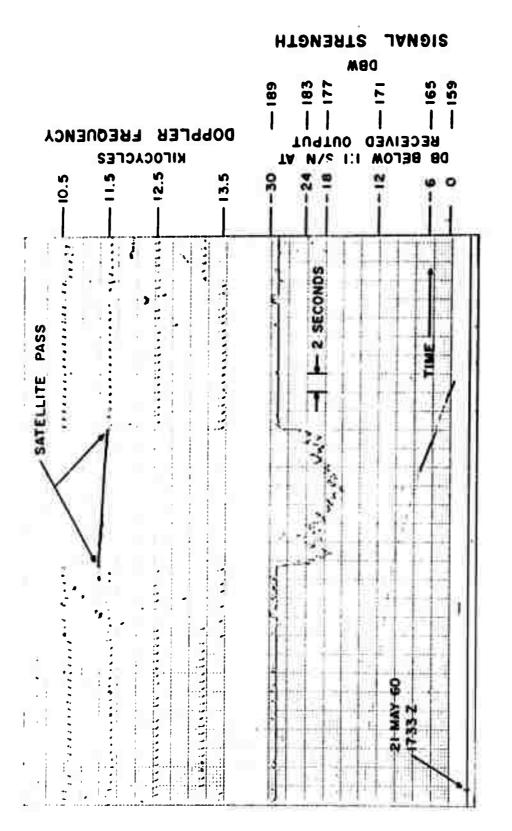
MEASURED 0300:412, PREDICTED 03012

ALTITUDE 230 MILES, 15 MILES EAST FT. SILL

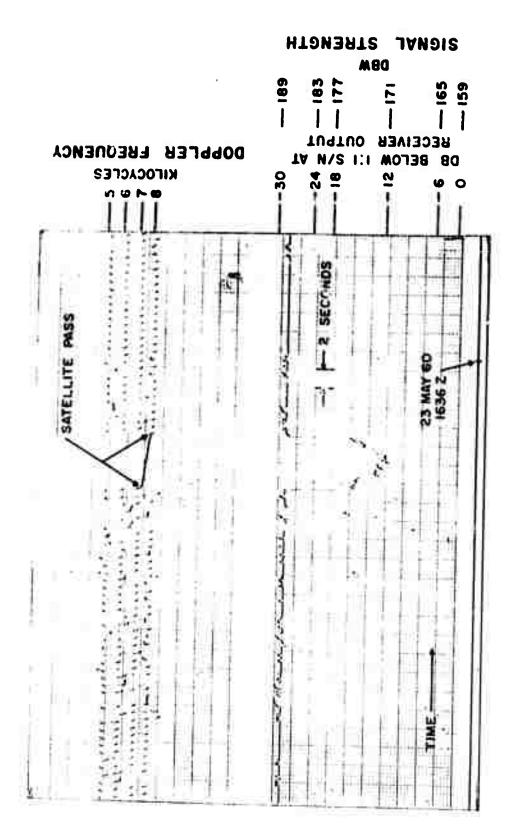
CENTER ANTENNA, SOUTH - NORTH PASS



ARPA - BRL DOPLOC DOPPLER RECORD OF
60 EPSILON I REV. 522, FORREST CITY, ARKANSAS
MEASURED 0028.412, PREDICTED 0028 2
ALTITUDE 225 MILES, 20 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH - NORTH PASS



ARPA - BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 2 REV. 106, FORREST CITY, ARKANSAS
MEASURED 1733:242, PREDICTED 1735 2
ALTITUDE 208 MILES, 203 MILES WEST FT. SILL
SOUTH ANTENNA, NORTH - SOUTH PASS



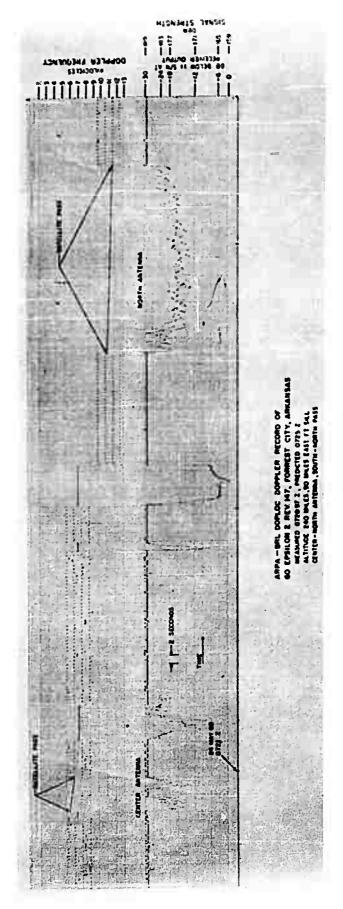
ARPA - BRIL DOPLOC DOPPLER RECORD OF

60 EPSILON 2 REV. 137, FURREST CITY, ARKANSAS

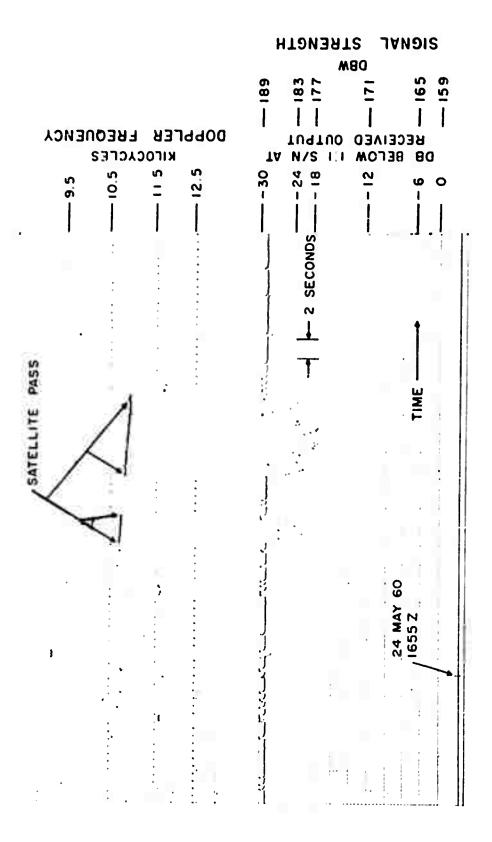
MEASURED 1635:462, PREDICTED 1639 Z

ALTITUDE 209 MILES, 274 MILES EAST FT. SILL

CENTER ANTENNA, NORTH - SOUTH PASS



F18. 89



ARPA -- BRL DOPLOC DOPPLER RECORD OF

60 EPSILON 2 REV. 153, FORREST CITY, ARKANSAS

MEASURED 1655:14 Z, PREDICTED 1657 Z

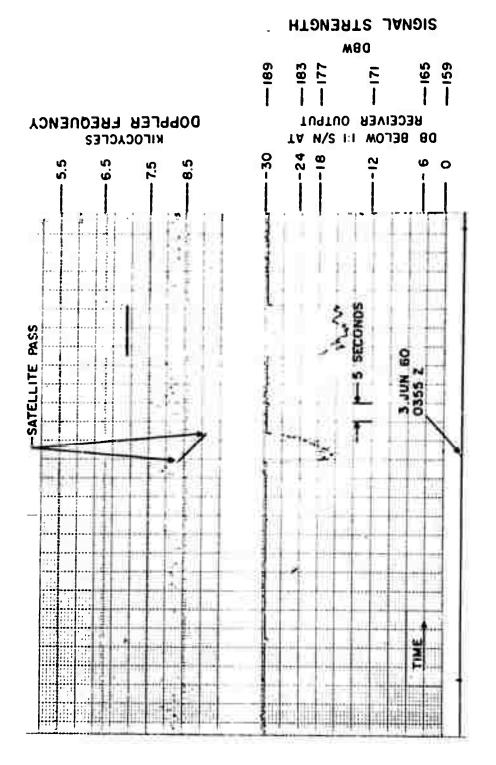
ALTITUDE 209 MILES, 112 MILES EAST FT. SILL

SOUTH ANTENNA, NORTH -- SOUTH PASS

F18. 90

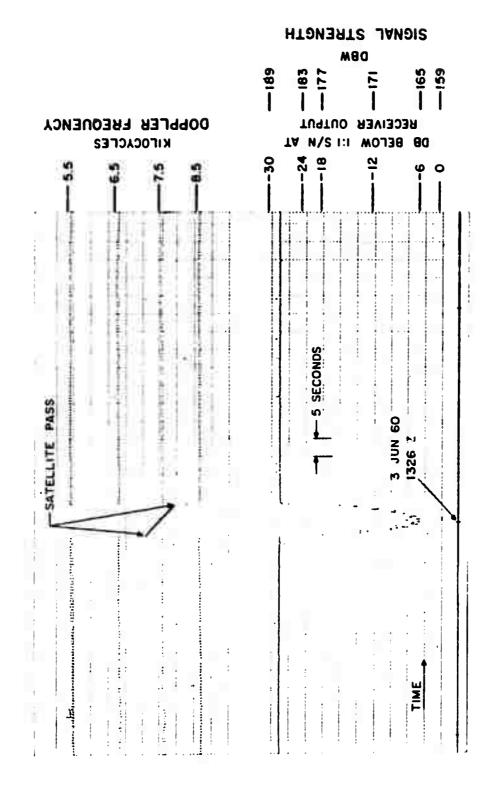
ARPA - BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 2 REV. 194, FORREST CITY, ARKANSAS
MEASURED 0642 Z, PREDICTED 0638 Z
ALTITUDE 193 MILES, 159 MILES EAST FT. SILL
NORTH ANTENNA, SOUTH - NORTH PASS

F18. 91



ARPA-BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 2 REV. 303, FORREST CITY, ARKANSAS
MEASURED 0354:58 2, PREDICTED 0356 2
ALTITUDE 213 MILES, 420 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH-NORTH PASS

F16. 92

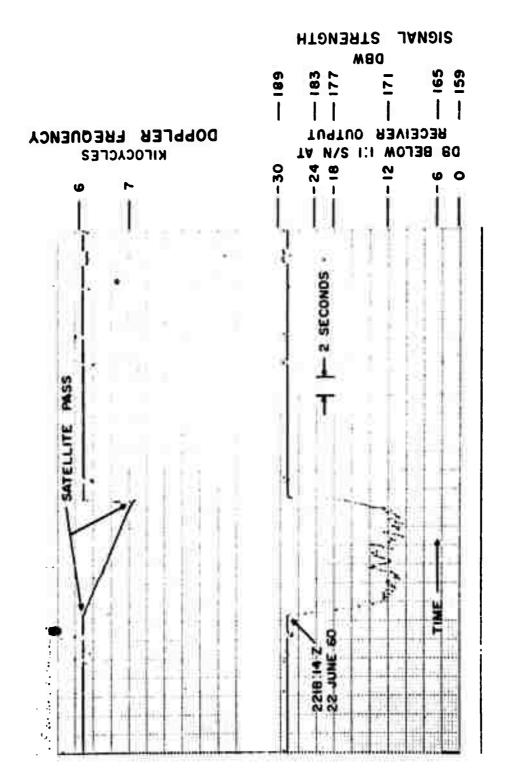


ARPA — BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 2 REV. 309, FORREST CITY, ARKANSAS
MEASURED 1325:57 Z, PREDICTED 1327 Z
ALTITUDE 205 MILES, 305 MILES EAST FT. SILL
CENTER ANTENNA, NORTH — SOUTH PASS

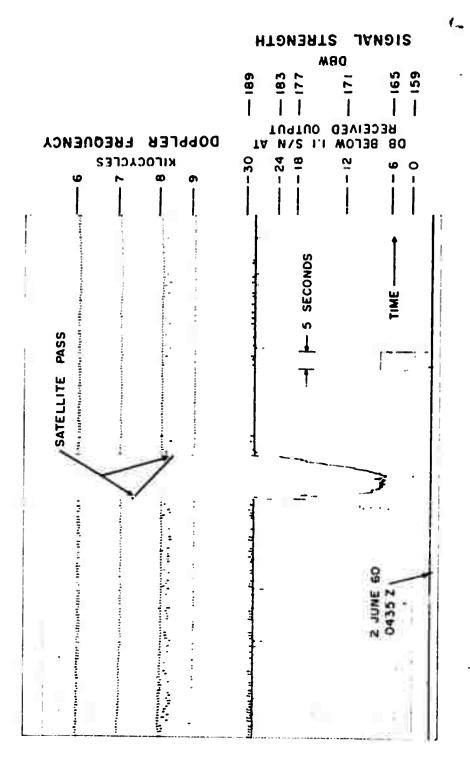
ARPA - BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 2 REV. 356, FORREST CITY, ARKANSAS
MEASURED 1240:39 Z, PREDICTED 1238 Z
ALTITUDE 20! MILES, 30 MILES EAST FT. SILL
NORTH ANTENNA, NORTH - SOUTH PASS

F16. 94

150



ARPA - BRL DOPLOC. DOPPLER RECORD OF
60 EPSILON 2 REV. 617, FORREST CITY, ARKANSAS
MEASURED 2218:14 Z, PREDICTED 2218 Z
ALTITUDE 203 MILES, 77 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH - NORTH PASS



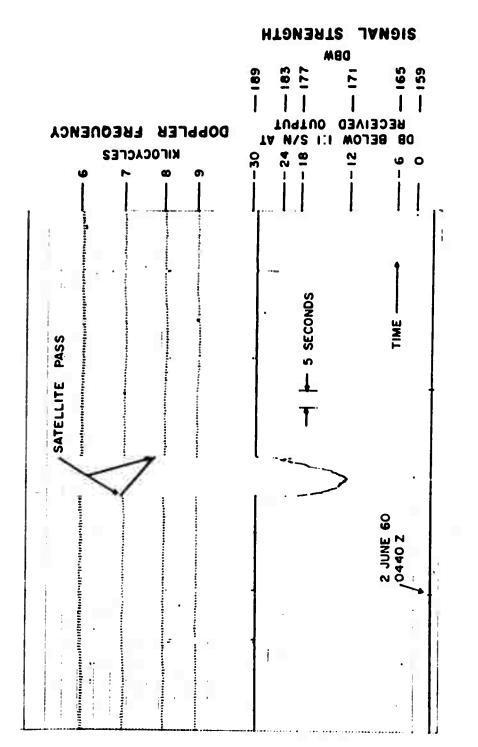
ARPA - BRL DOPLOC DOPPLER RECORD OF

60 EPSILON 3 REV. 280, FORREST CITY, ARKANSAS

MEASURED 0435:21 Z, PREDICTED 0435 Z

ALTITUDE 232 MILES, 358 MILES EAST FT. SILL

CENTER ANTENNA, SOUTH - NORTH PASS

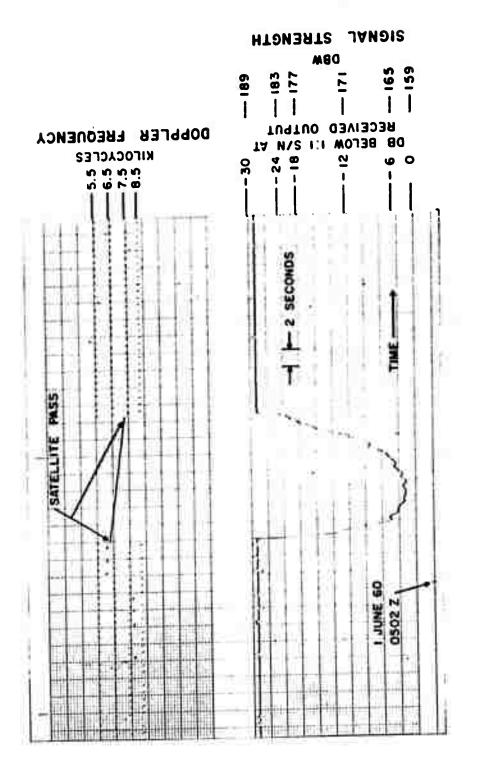


ARPA - BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 4 REV. 280, FORREST CITY, ARKANSAS
MEASURED 0441:29 2, PREDICTED 0441 2
ALTITUDE 240 MILES, 270 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH - NORTH PASS

SIGNAL STRENGTH

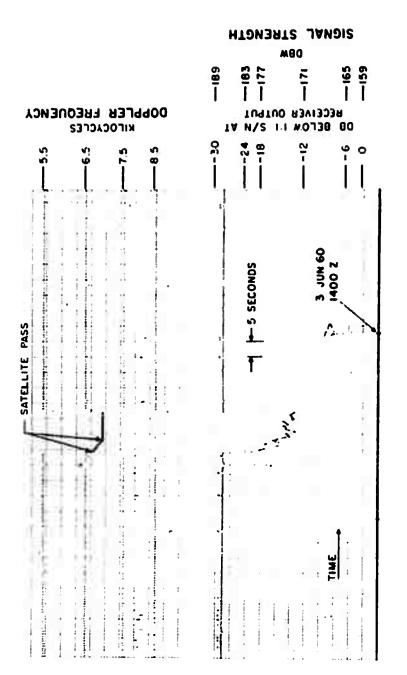
ARPA-BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 5 REV. 569, FORREST CITY, ARKANSAS
MEASURED 2325:46 Z, PREDICTED 2327 Z
ALTITUDE 205 MILES, 231 MILES EAST FT. SILL
CENTER ANTENNA, SOUTH - NORTH PASS

F16. 98

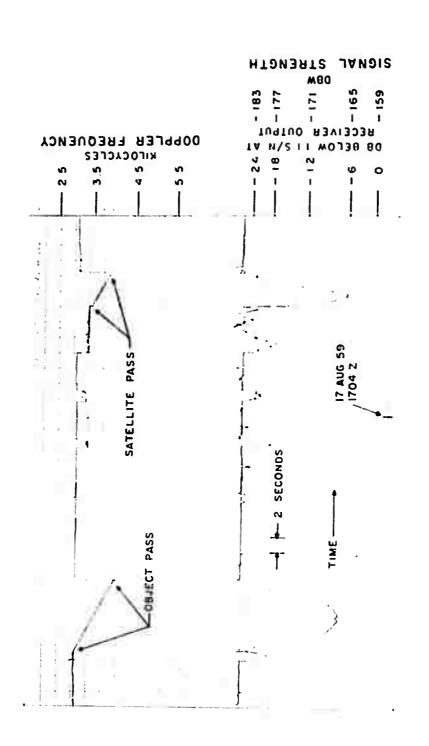


ARPA - BRL DOPLOC DOPPLER RECORD OF 60 EPSILON 6 REV. 265, FORREST CITY, ARKANSAS MEASURED 0502:06 Z, PREDICTED 0502 Z ALTITUDE 218 MILES, 210 MILES EAST FT. SILL CENTER ANTENNA, SOUTH - NORTH PASS





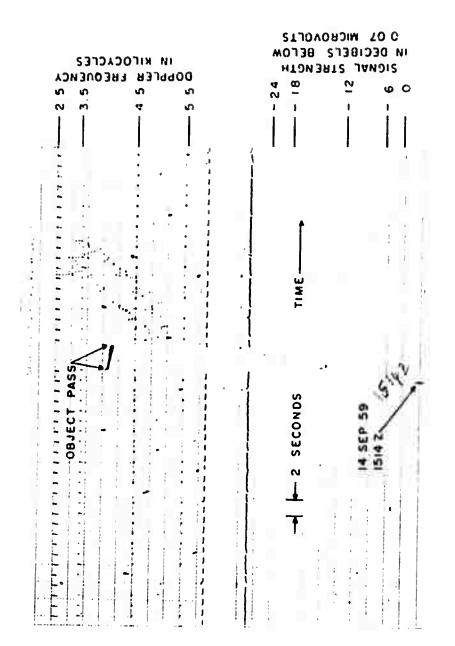
ARPA -- BRL DOPLOC DOPPLER RECORD OF
60 EPSILON 6 REV. 301, FORREST CITY, ARKANSAS
WEASURED 1359:22 Z. PREDICTED 1400 Z
ALTITUDE 233 MILES, 190 MILES EAST FT SILL
CENTER ANTENNA, NORTH -- SOUTH PASS



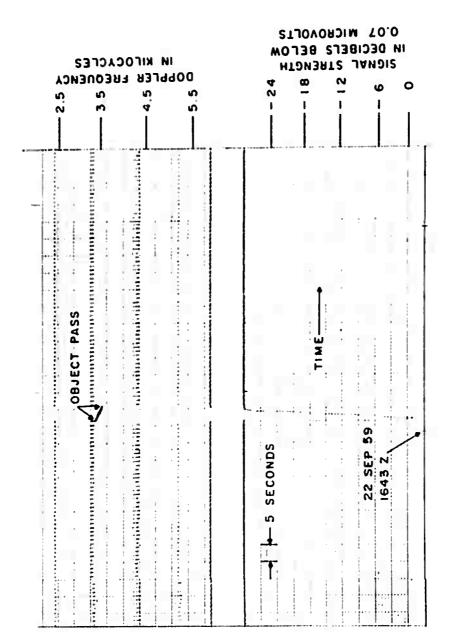
ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT AND 59 EPSILON FORREST CITY, ARKANSAS

UNIDENTIFIED OBJECT MEASURED 1703:30 Z, CENTER ANTENNA ALTITUDE 125 MILES, OVERHEAD FT. SILL

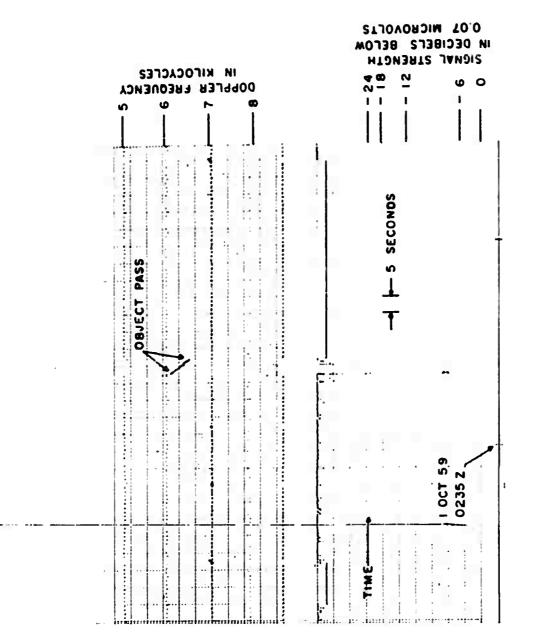
59 EPSILON REV 60 MEASURED 1704-14 Z, PREDICTED 1701 Z ALTITUDE 140 MILES, 76 MILES WEST FT. SILL CENTER ANTENNA, N-S PASS



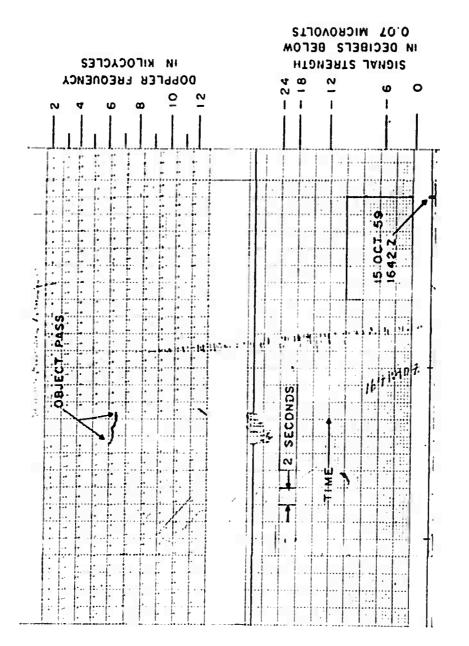
ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 1514:01 2, CENTER ANTENNA ALTITUDE 432 MILES, 318 MILES EAST FT. SILL



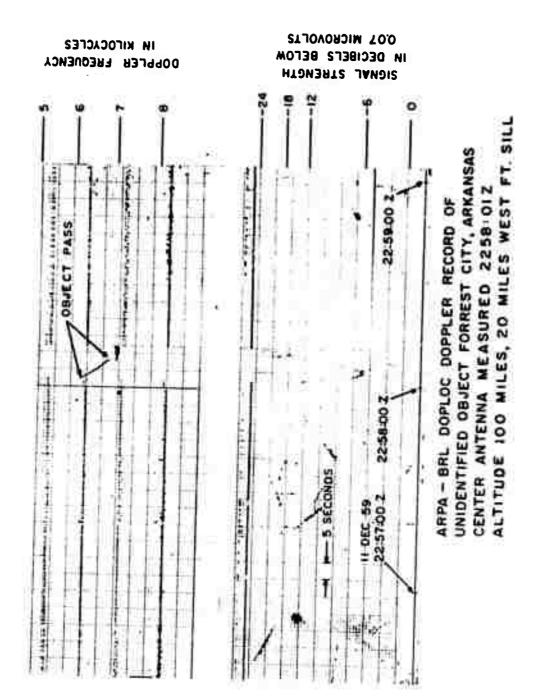
ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 1643:04 Z, CENTER ANTENNA ALTITUDE 398 MILES, 140 MILES EAST FT. SILL

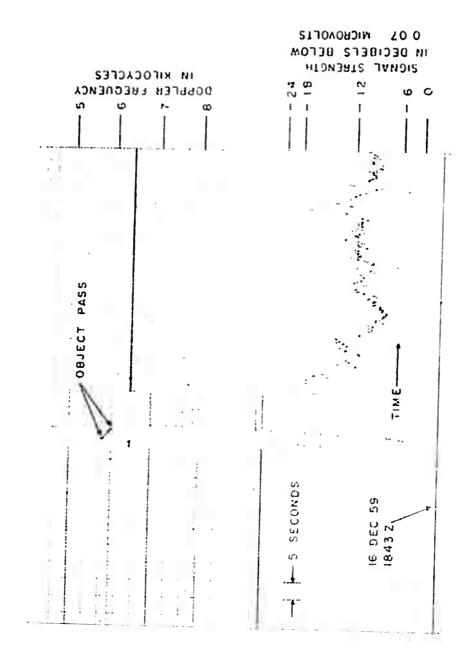


ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 0235:222, CENTER ANTENNA ALTITUDE 190 MILES, 117 MILES EAST FT. SILL

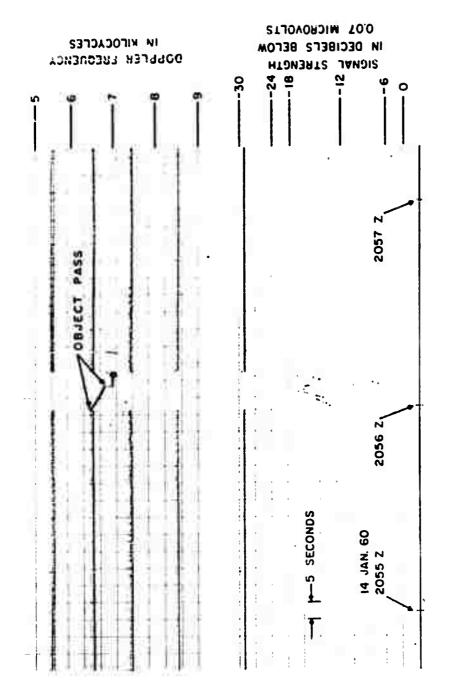


ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 1641:312, CENTER ANTENNA ALTITUDE 178 MILES, 115 MILES EAST FT. SILL

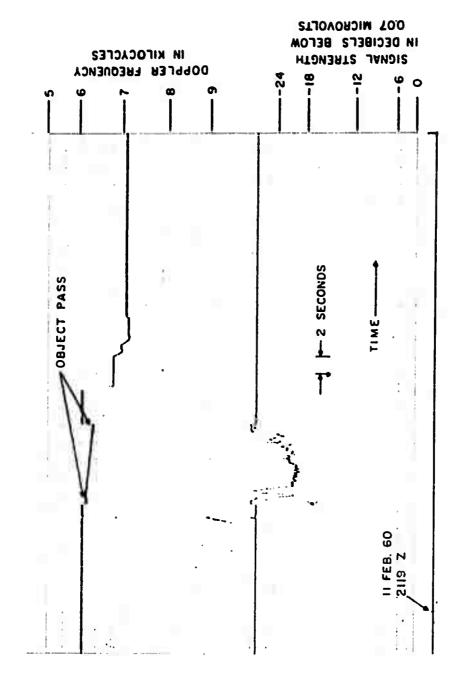




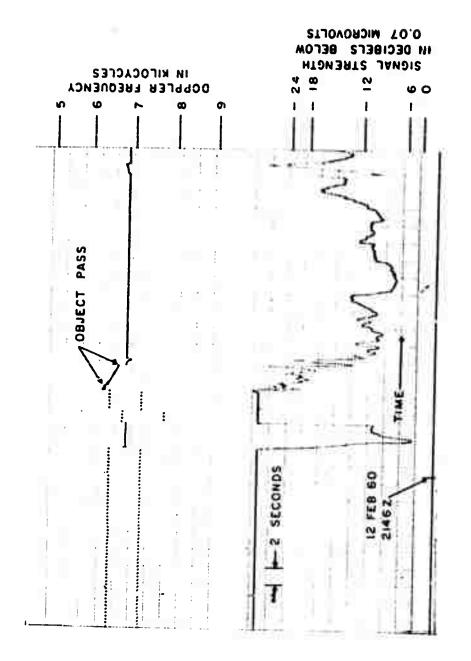
ARPA-9RL DOPLOC DOPPLER RECORD CF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 1843-18 2, CENTER ANTENNA ALTITUDE 198 MILES, 165 MILES EAST FT SILL



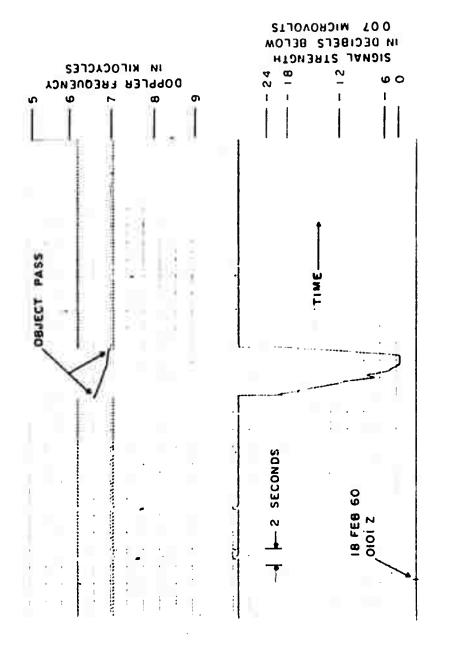
ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT, FORREST CITY, ARKANSAS CENTER ANTENNA MEASURED 2056:00 Z ALTITUDE 485 MILES, 140 MILES EAST FT. SILL



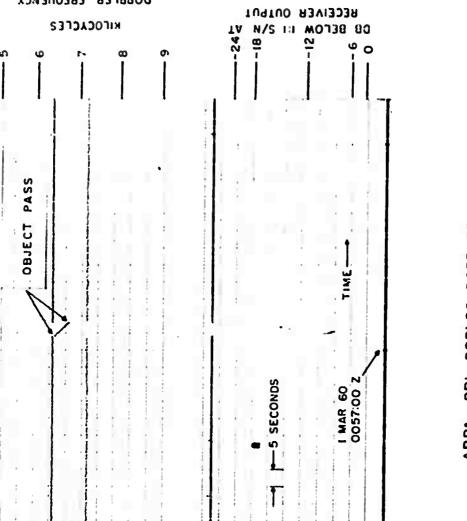
ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 2119:13 Z, CENTER ANTENNA ALTITUDE 900 MILES, OVERHEAD FORT SILL



ARPA-BRL DOPLOC DOPPLER RECORD OF Unidentified object forrest city, arkansas Measured 2146:10 2, center antenna Altitude 190 miles, 135 miles east ft. Sill



ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED DIDI 22 Z, CENTER ANTENNA ALTITUDE 265 MILES, 40 MILES EAST FT SILL



OBW SIGNAL STRENGTH

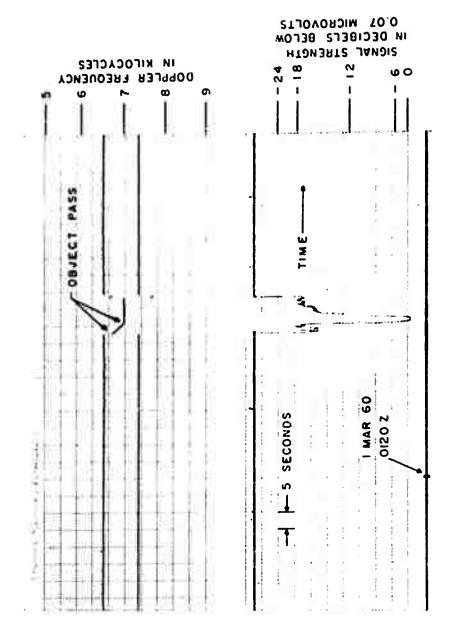
-165 -159

-177

DOPPLER FREQUENCY

ARPA — BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 0057:04 Z, CENTER ANTENNA ALTITUDE 190 MILES, 130 MILES EAST FT. SILL

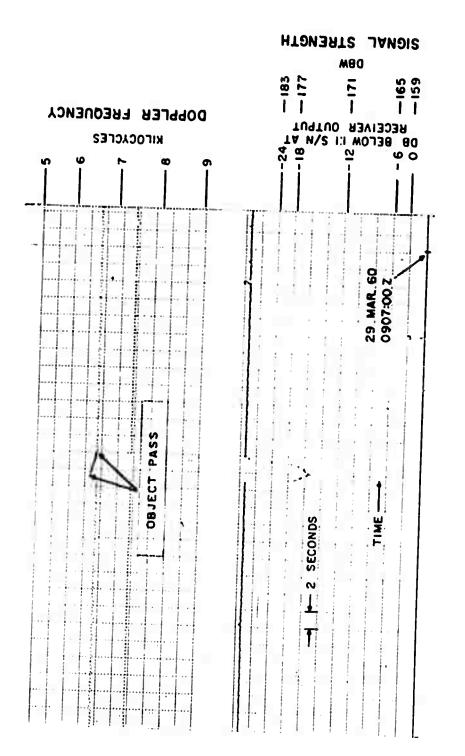
F16. 112



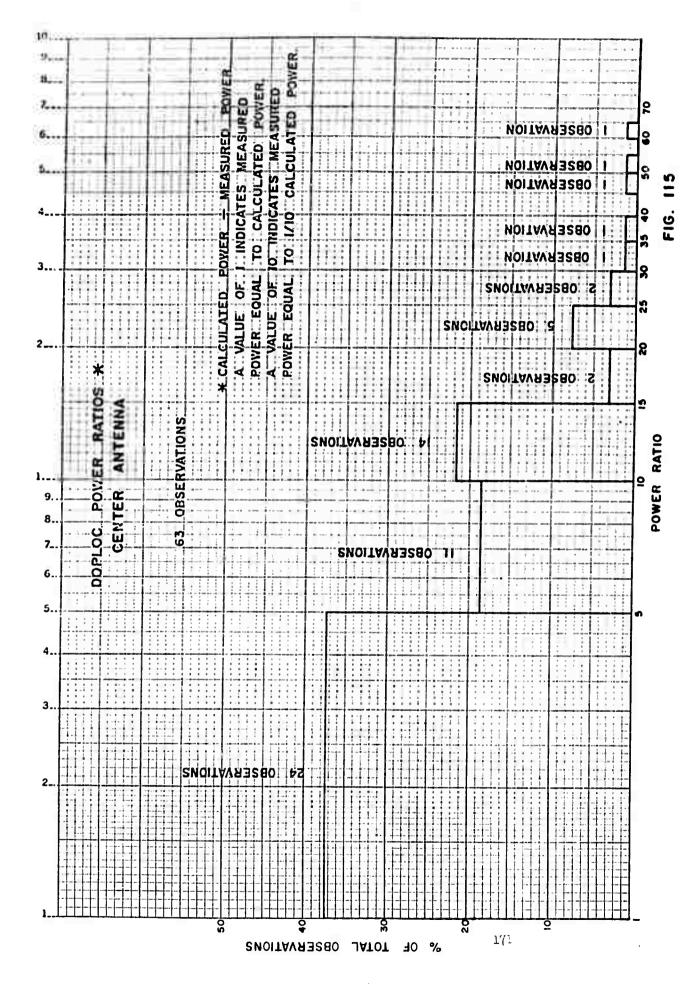
ARPA-BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT FORREST CITY, ARKANSAS MEASURED 0120:43 Z, CENTER ANTENNA ALTITUDE 195 MILES, 117 MILES EAST FT. SILL

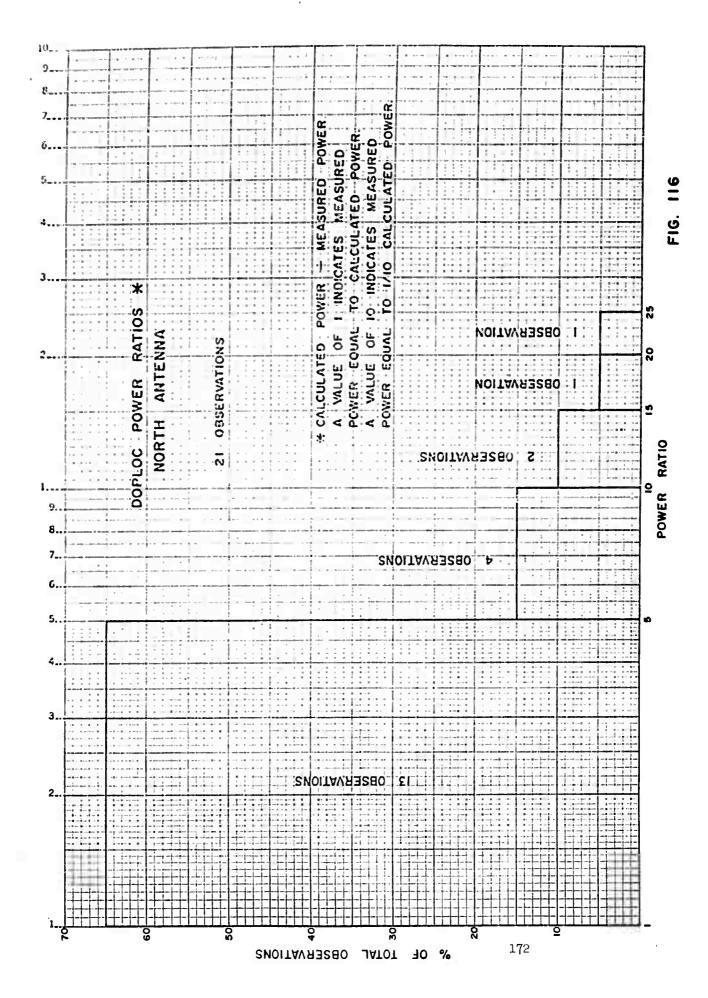
1

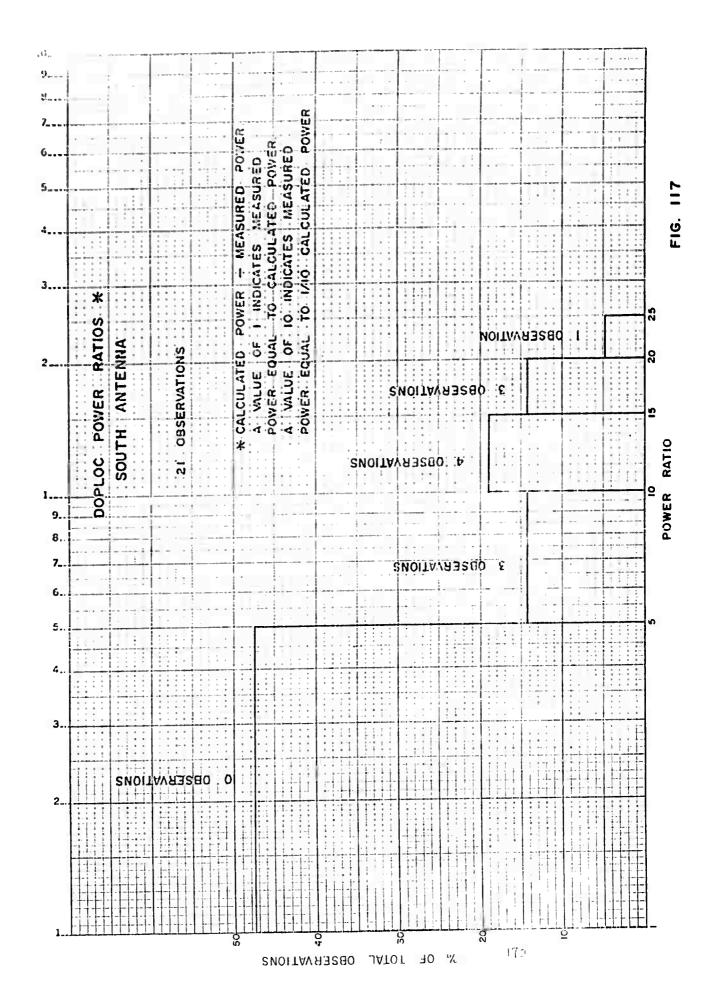


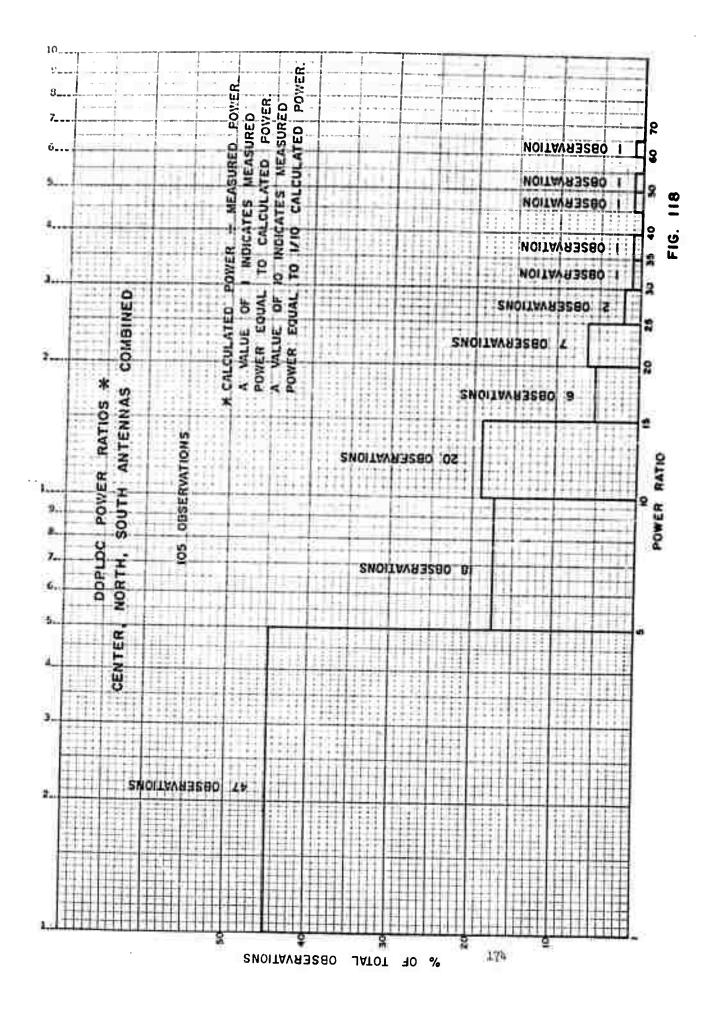


ARPA - BRL DOPLOC DOPPLER RECORD OF UNIDENTIFIED OBJECT, FCAREST CITY, ARKANSAS MEASURED 0906:33 Z, CENTER ANTENNA ALTITUDE 300 MILES, 90 MILES WEST OF FT. SILL



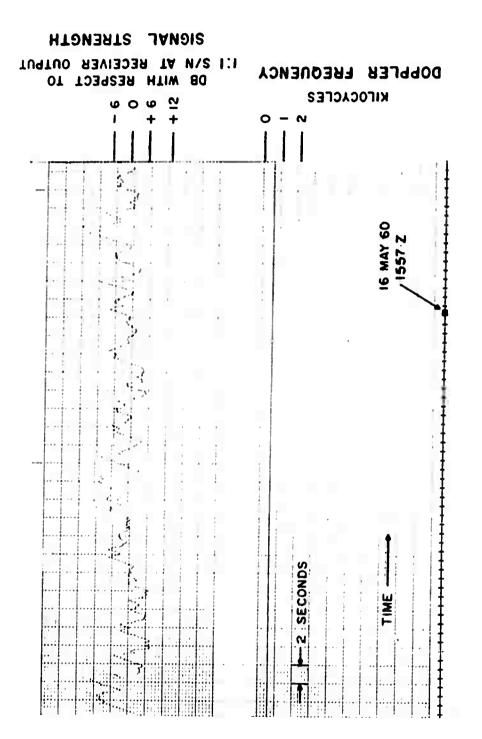




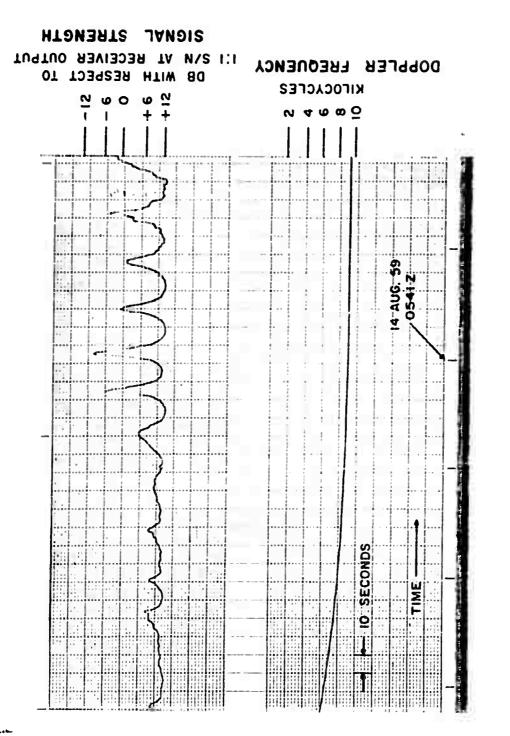


DOPPLER RECORD OF ACTIVE N-S TRACK OF 58 DELTA (SPUTNIK III) REV. 9958 AT ABERDEEN PROVING GROUND, MARYLAND

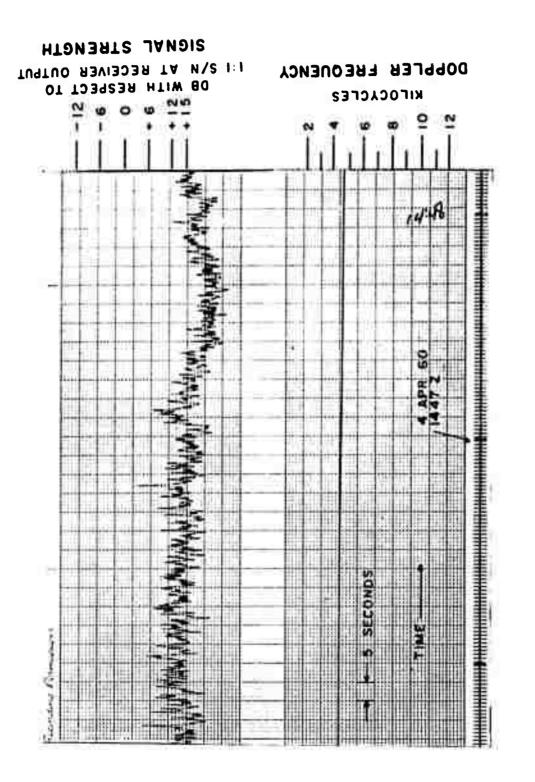
F18. 119



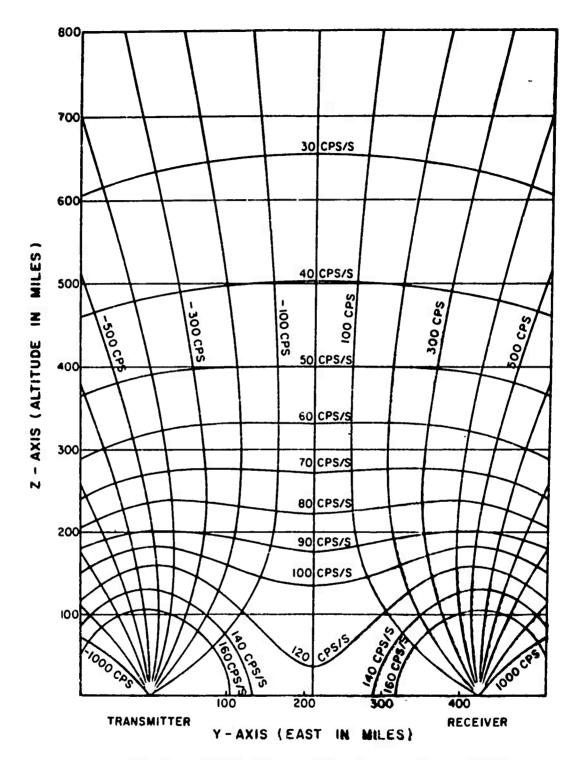
DOPPLER RECORD OF ACTIVE N-S TRACK OF 60 EPSILON I (SPUTNIK IXZ) REV. 26 AT ABERDEEN PROVING GROUND, MARYLAND



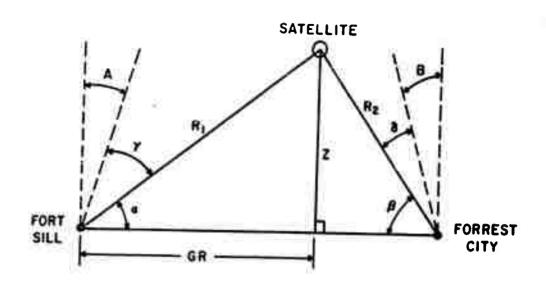
DOPPLER RECORD OF ACTIVE S-N TRACK OF 59 EPSILON (DISCOVERER XZ) REV. 7 AT ABERDEEN PROVING GROUND, MARYLAND



DOPPLER RECORD OF ACTIVE N-S TRACK OF 58 DELTA (SPUTNIK III) REV. 10007 AT ABERDEEN PROVING GROUND, MARYLAND



OPLOC FREQUENCY AND RATE OF CHANGE
OF FREQUENCY AS A FUNCTION OF
POSITION IN THE YZ - PLANE
(FOR 80° INCLINATION)



A = 24 = AZIMUTH ANGLE OF TRANSMITTING ANTENNA. .

B = 25 * = AZIMUTH ANGLE OF RECEIVING ANTENNA.

RI . SLANT RANGE FROM TRANSMITTER TO SATELLITE.

R2 = SLANT RANGE FROM RECEIVER TO SATELLITE.

·Y = ANGLE THAT RI MAKES WITH CENTER OF TRANSMITTING ANTENNA BEAM.

δ = ANGLE THAT R2 MAKES WITH CENTER OF RECEIVING ANTENNA BEAM.

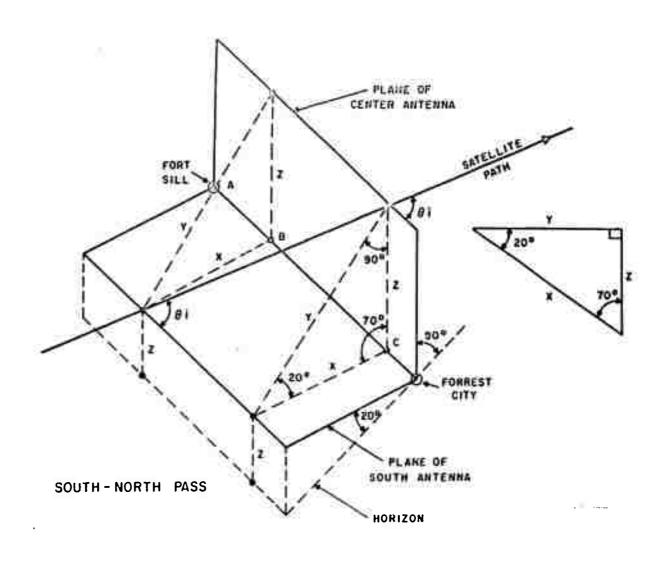
a = ANGLE THAT R, MAKES WITH BASE LINE.

B = ANGLE THAT R2 MAKES WITH BASE LINE.

Z = ALTITUDE OF SATELLITE IN CENTER ANTENNA.

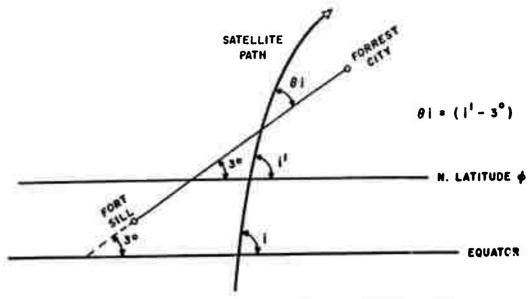
GR = GROUND RANGE FROM FORT SILL.

FIG. 124 - CENTER ANTENNA GEOMETRY

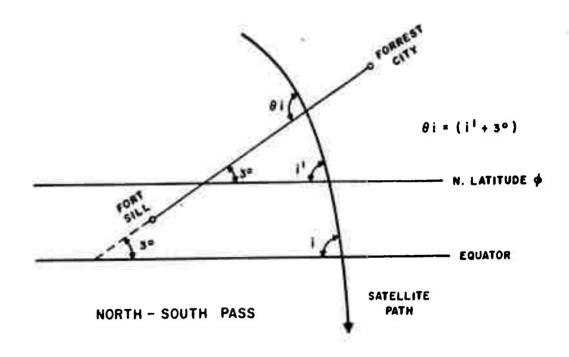


- Z = SATELLITE HEIGHT IN CENTER AND NORTH OR SOUTH ANTENNAS.
- X = PERPENDICULAR DISTANCE BETWEEN BASE LINE AND SATELLITE INTERSECTION POINT IN NORTH OR SOUTH ANTENNAS.
- Y = PERPENDICULAR DISTANCE BETWEEN 20° PLANE AND 90° PLANE.
- AC = r = GROUND RANGE IN CENTER ANTENNA.
- BC = d = DISTANCE MOVED ALONG BASE LINE.
- AB = GROUND RANGE IN SOUTH .ANTENNA.
- θ i = ANGLE BETWEEN SUB SATELLITE TRACE AND BASE LINE.

FIG. 125 - ALTITUDE AND GROUND RANGE GEOMETRY IN SOUTH ANTENNA

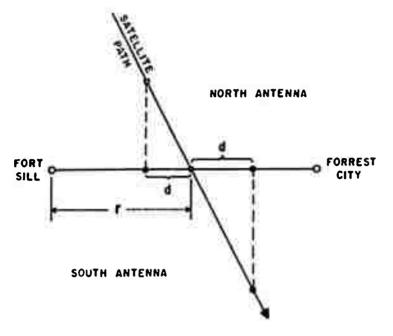


SOUTH - NORTH PASS



- I = ORBITAL INCLINATION AT EQUATOR.
- i^{\dagger} = ORBITAL INCLINATION AT NORTH LATITUDE ϕ .
- # = ANGLE BETWEEN SUB-SATELLITE TRACE AND BASE LINE.

FIG. 126 - SATELLITE INCLINATION WITH RESPECT TO BASE LINE



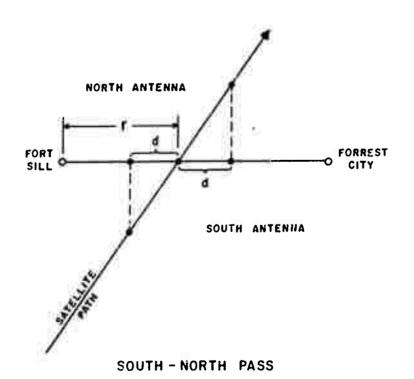
NORTH - SOUTH PASS

GROUND RANGE (GR)

IN NORTH ANTENNA = r - d

IN SOUTH ANTENNA = f+d

WHERE r = GR IN CENTER
ANTENNA AND IS EAST OF
FORT SILL, AND D = DISTANCE
MOVED ALONG BASE LINE.



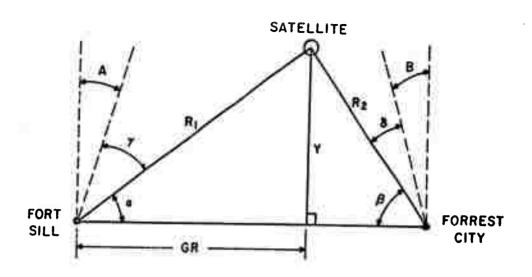
GROUND RANGE (GR)

IN NORTH ANTENNA = r + d

IN SOUTH ANTENNA = r - d

WHERE F = GR IN CENTER
ANTENNA AND IS EAST OF
FORT SILL, AND D = DISTANCE
MOVED ALONG BASE LINE.

FIG. 127 - GROUND RANGE IN NORTH AND SOUTH ANTENNAS



A = 24° = AZIMUTH ANGLE OF TRANSMITTING ANTENNA.

B = 25° = AZIMUTH ANGLE OF RECEIVING ANTENNA.

RI = SLANT RANGE FROM TRANSMITTER TO SATELLITE.

R2 = SLANT RANGE FROM RECEIVER TO SATELLITE.

Y = ANGLE THAT R; MAKES WITH CENTER OF TRANSMITTING ANTENNA BEAM.

δ = ANGLE THAT R2 MAKES WITH CENTER OF RECEIVING ANTENNA BEAM.

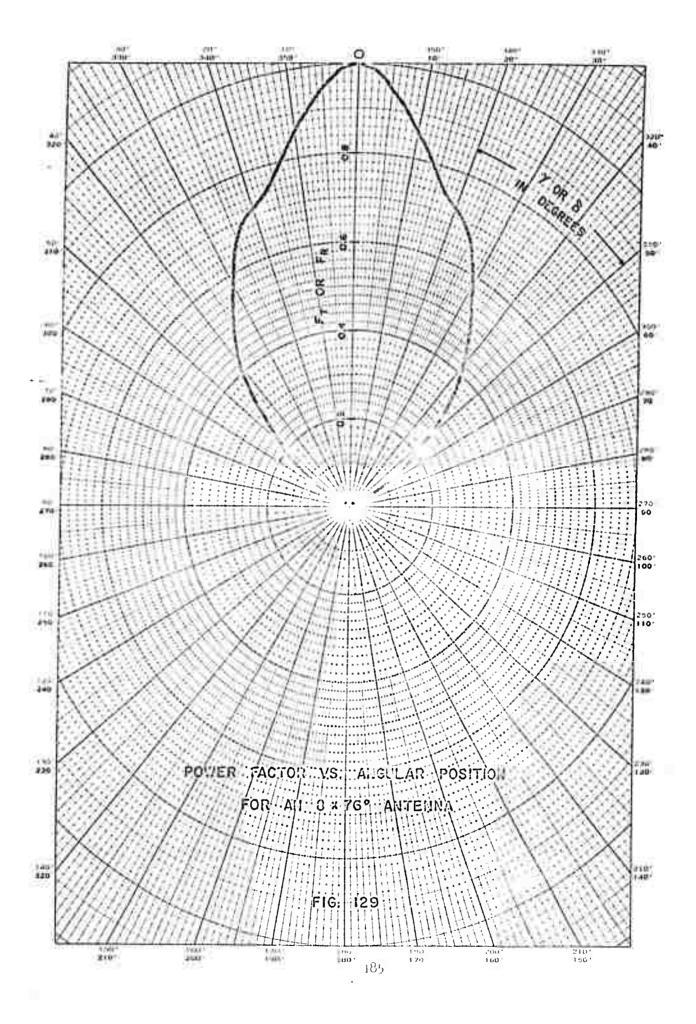
 α = ANGLE THAT R_I MAKES WITH BASE LINE.

 β = ANGLE THAT R₂ MAKES WITH BASE LINE.

Y = PERPENDICULAR DISTANCE BETWEEN SATELLITE INTERSECTION POINT IN 20° PLANE AND 90° PLANE.

GR = GROUND RANGE FROM FORT SILL.

FIG. 128 - NORTH AND SOUTH ANTENNA GEOMETRY



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DOPLOC OBSERVATIONS OF REFLECTION CROSS
SECTIONS OF SATELLITES Harold T. Lootens

BRL Memorandum Report No. 1330 March 1961

DA Proj No. 503-06-011, CMSC No. 5210.11.14.3 UNCLASSIFIED

This report presents reflection eross sections observed for eight satellites during the period 1 January 1959 to 1 July 1960, using the DDFLOC "dark satellite" detection system developed by the Ballistic Research Laboratories. Several related areas are discussed; i.e., satellite "signature", spin and tumble, scintillation and ionized trails. A brief description of the DDFLOC receiving system and antenna configuration is included. The method used for calculation of cross sections is given in Appendix I.

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UNCLASSIED

HEL Menormadum Report No. 1350 Nameh 1961

IN Proj No. 503-66-111, CASC No. 5210.111.143 UNDIASSIFIED

This report presents reflection cross sections observed for eight satelliten during the period 1 January 1959 to 1 July 1950, using the DOFLOC "dark satellite" derection system dereloped by the Ballistic Research Labora. tories. Several related areas are discussed; i.e., satellite "signature", spin and tumble, seintillation and fonfred trails. A brief description of the DOFLOC receiving system and sateinan configuration is included. The method used for enleulation of eross sections is given in Appendix I.

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BRL Memorandum Report No. 1530 March 1961

DA Proj No. 503-06-011, OMSC No. 5210.11.143 UNCIASSIFIED

This report presents reflection cross sections observed for eight satellites during the period 1 January 1959 to 1 July 1960, using the DNPLOC "dark satellite" detection system developed by the Ballistic Research Labornteries. Several related areas are discussed; i.e., sutcilite "signature", spin and tumble, scintillation and ionized trails. A bricf description of the DNPLOC receiving system and antenna configuration is included. The method used for calculation of cross sections is given in Appendix I.

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DA Proj No. 503-06-011, OMSC No. 5210.111.143 UNCIASSIFIED

This report presents reflection cross sections observed for eight satellites during the period 1 January 1959 to 1 July 1960, using the DOFLOC "dark entellite" detection system developed by the Ballistic Research Laboratories. Several related areas are discussed; i.e., satellite "signature", spin and temble, acintillation and fonface trails. A brief description of the DOFLOC receiving system and anticana configuration is included. The method used for calculation of cross sections is given in Appendix I.

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satellites during the period 1 January 1959 to 1 July 1960, using the Davior "dark satellite" detection system developed by the Bullistic Research Labora-tories. Several related area are discussed; i.e., sutellite "signature", spin and tumble, scintillation and ionized trails. A brief description of the Dapiot receiving system and antenna configuration is included. The method used for calculation of eross sections is given in Appendix I. This report presents reflection cross sections observed for eight

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